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**Effective Length Factors
for
Solid Round Diagonals in Guyed Communication Towers**

By

YEAN SUN

**A Thesis Submitted to
The College of Graduate Studies and Research
Through The Faculty of Engineering
(Civil Engineering Program)
In Partial Fulfillment of the Requirements for
The Degree of Master of Applied Science at the University of Windsor**

Windsor, Ontario, Canada

1999



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ABSTRACT

The objective of this investigation is to determine the effective length factors for solid round diagonals in all-welded communication towers. A total of thirty-three welded towers were tested in the investigation with all the towers in a horizontal position and concentrated load at mid-span. Of the thirty-three specimens, fifteen specimens were six-panel cross-braced towers with a span of 4572 mm (180 in). The leg sizes were 38.1 mm (1.5 in), 50.8 mm (2 in), 69.85 mm (2.75 in) diameter, and diagonal sizes were 12.7 mm (0.5 in), 14.29 mm (0.563 in), 15.88 mm (0.625 in) diameter. Sixteen specimens were eight-panel cross-braced towers with a span of 5690 mm (224 in). The leg sizes were 38.1 mm (1.5 in), 50.8 mm (2 in), 57.15 mm (2.25 in) diameter, and the diagonal sizes were 19.05 mm (0.75 in), 22.23 mm (0.875 in), 15.88 mm (0.625 in) diameter. Two specimens were six-panel single-braced towers with leg sizes of 50.8 mm (2.0 in) and 76.2 mm (3.0 in) and diagonal sizes of 22.23 mm (0.875 in), 28.58 mm (1.125 in) diameter.

Two specimens were fixed with strain gauges on every member, while all other specimens were tested with strain gauges fixed on only one panel.

After the test, tensile specimens were cut from the diagonals of the tower, and tested for the yield stress and tensile strength. Axial forces in the diagonal and leg members were calculated based on the actual yield stress and strain readings, assuming an elastic perfectly-plastic stress-strain curve for steel.

The buckling loads were determined from the maximum axial forces in the tested members, and effective length factors were calculated according to CSA-S37-94 Standard and AISC-LRFD Specifications. The average effective length based on CSA-S37-94 for cross-braced diagonals is 0.35 of the total diagonal length, while

the effective length is 0.37 of the total diagonal length based on AISC-LRFD. The effective length factor for single-braced diagonals is 0.50 and 0.56 based on CSA-S37-94 and AISC-LRFD, respectively.

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NOTATION

A	cross-sectional area
B_m	the thickness of slice "m"
C	force in the cross-braced compression member
C_{cr}	the critical load for compression member
C_e	Euler critical load
C_T	the critical tangent modulus load
C_r	compressive resistance
D	diameter of the test member
dy	the thickness of slice
E	modulus of elasticity
E_T	tangent modulus of elasticity
E_r	reduced modulus of elasticity
E_{eff}	effective modulus of elasticity
F	resultant axial force in the test member
F_{cr}	critical load according to AISC-LRFD Specification
F_m	resultant force in the slice "m"
F_y	yield stress of material
I	moment of inertia
I_t	moment of inertia of the tension diagonal

I_c	moment of inertia of the compression diagonal
K	effective length factor
KL/r	effective slenderness ratio of the member
L	length of compressive member
L_d	total length of diagonal center to center of legs
T	force in the cross-braced tension member
X	the depth of neutral axis
Y_m	the distance measured from the center of slice to centroidal axis of the cross section
r	the radius of gyration
α	stiffness furnished by the tension member to compression member
α_{lim}	the limiting value of the stiffness of the elastic support
λ	the nondimensional slenderness parameter
ϕ	resistance factor for compression
ϵ_m	the strain in the center of the slice "m"
σ_m	the stress for slice "m"
ϵ_1, ϵ_2	strain readings in the opposite sides of the test member

CHAPTER 1

INTRODUCTION

1.1 General

With the rapid development of world economy and technology, in the fields of construction and telecommunication, more and more welded towers are constructed and used. The welded towers are either self-supporting or guyed towers. Self-supporting latticed towers are of variable cross-section, whereas guyed communication towers are of constant triangular or square cross-section.

Welded towers consist of legs, diagonals, and sometimes horizontals, which are connected together by welding. Solid rounds, tubes, as well as angles are used for the components of the welded towers. Diagonals can be arranged in cross-braced or single-braced configuration.

Towers with cross-braced diagonals can be fabricated in one of the following ways:

Type I: one diagonal straight and the other bent around and welded at the intersection.

Type II: both the cross-braced diagonals are in-plane with one of them continuous and the other cut and welded to the continuous member.

Type III: both the cross-braced diagonals are out-plane at mid-span and then welded together at their intersection.

During the design of welded towers, effective length factors for diagonals are a major parameter in the computation of the compressive strength of diagonals. Effective length factors for cross-braced diagonals of type III are the subject of this investigation.

1.2 Necessity for the Investigation

To the best of author's knowledge, though many experiments have been carried out on the behavior of cross-braced angle diagonals, little experimental investigation was carried out on the behavior of cross-braced solid round members. CSA-S37-94 "Antennas, Towers and Antenna-Supporting Structures" (CSA 1994) doesn't give specific effective length factors for solid rounds. It only gives 0.75 for the cross-braced angle diagonals, and notes that for the cross-braced solid round diagonals, the philosophy is the same as that of angles. Based on the preliminary results of investigation carried out at the University of Windsor (Jaboo 1998), the effective length was found to be less than 0.5 of the diagonal length. There is a need to confirm this result by carrying out tests on additional specimens fabricated by different manufacturers.

1.3 Objectives of Present Research

The objectives of this investigation are:

- To study the failure patterns of cross-braced and of single-braced solid round diagonals for welded towers.
- To determine effective length factors of type III cross-braced and of single-braced solid round diagonals for welded towers.

CHAPTER 2

LITERATURE REVIEW

2.1 General

The use of solid round columns dates back many centuries. The materials used varied from stone, wood, concrete, and metals. Steel has been widely used in many structural elements. Solid round steel bars have been used in constructing trusses, frames, and towers. In lattice towers, solid rounds are used for legs, horizontals, and bracing diagonals. Single and cross-braced diagonals are frequently used in such structures.

A large amount of literature is available, theoretical and experimental, about the design of cross-bracings, but most of the work carried out was on angles and flat bars. Very little literature is available on cross-bracings made of solid round bars, although this type of section is quite common for antenna towers.

Structural member buckling capacities are dependent upon the mode of buckling. For axisymmetrical solid round cross sections, the mode of buckling is flexural buckling.

2.2 Flexural Buckling Theory and Experimental Investigation

2.2.1 Euler Formula

Stability of columns has been one of the most widely known aspects of structural engineering. Research with regard to a solution to column problem dates back to many centuries. One of the most important contributions to the problem was made by Euler (1759) who established a formula for the buckling of columns in analyzing the elastic stability of many engineering structures. Euler's formula is as follows:

$$C_e = \frac{\pi^2 EA}{\left(\frac{KL}{r}\right)^2} \quad (2.1)$$

where, E is the modulus of elasticity, A is the area of the cross-section, K is the effective length factor, L is the length of the column, and r is the radius of gyration. Euler's load, C_e , is the critical load at which a slender elastic column can be held in a bent configuration under axial load alone. The formula is valid only when the stress does not go beyond the elastic limit of the material.

2.2.2 Tangent Modulus Theory

Due to lack of knowledge as to relationship between stress, strain, curvature, and bending moment beyond the elastic range, progress beyond Euler's early statements concerning inelastic behavior remained dormant for many years. In 1889, Engesser proposed that the Euler's formula be extended to include the inelastic range by introducing a variable quantity E_T which is called the tangent modulus, instead of the constant modulus E . Thus for a hinged-end column, the buckling formula should be changed to tangent modulus formula:

$$C_T = \frac{\pi^2 E_T A}{\left(\frac{KL}{r}\right)^2} \quad (2.2)$$

where, E_T is the slope of the tangent to stress-strain curve corresponding to the stress at failure, KL/r is the effective slenderness ratio of the member, and C_T is the critical tangent modulus load.

2.2.3 Reduced Modulus Theory

Independent of Engesser, Considere in 1889 conducted a series of 32 column tests and suggested that if buckling occurred above the proportional limit, the elastic modulus, E , should be replaced in Euler's formula by the effective modulus E_{eff} . He stated that the effective modulus should be somewhere between E and E_t . Jasinsky (1889) brought Considere's work to the attention of

Engesser, who revised his theory by introducing two different moduli for the two parts, convex and concave sides of the buckled shape of the cross-section.

In 1910 Karman derived explicit expressions for the reduced modulus for both rectangular and idealized H-section columns. The formula for critical reduced modulus stress in a pinned-end column is the same as equation (2.2) except that E_T (tangent modulus) is replaced by E_r (reduced modulus), where

$$E_r = f(E_T, E) \quad (2.3)$$

2.2.4 Effect of Residual Stress and Initial Out-of-Straightness

The need for information on the strength of solid round compression members became necessary from the increased use of these bars in communication towers. For this reason, an extensive program of experimental and analytical work was initiated in 1954 at Lehigh University, Bethlehem, Pennsylvania. Beedle, Galambos, and Tall (Beedle et al. 1961) studied the effect of heat-treatment and cold bending on the residual stresses and column strength. Effects of initial out-of-straightness were also illustrated. It was stated that for solid round bars, the residual stress distribution is triaxial, i.e. there are longitudinal, tangential and radial residual stresses. Different final heat treatment methods like quenching, air cooling, and stress relieving were considered in the study. It was found that of the three types of heat treatment, quenching leads to high residual stresses while the other two methods reduce the stresses considerably. It was also shown that columns with initial out-of-straightness have lower strength than straight columns. Fujita and Driscoll (1962) presented results of the aforementioned experimental investigation on the strength of axially and eccentrically loaded solid round columns of 69.9 mm (2-3/4 in.) diameter. Comparison with a theory based on "the tangent modulus" concept for axially loaded columns showed that the strength of solid round columns may be predicted adequately. Galambos (1965) presented the research on the strength

of solid rounds carried out between 1954 and 1965 at Lehigh University. Recommendations for the design of axially loaded steel columns were presented based on the research program. It was stated that both residual stresses and initial crookedness have a significant influence on the strength of round columns in the inelastic region. Galambos' paper was discussed by Williamson and Johnston (1965).

2.2.5 Flexural Buckling - Experimental Investigation

Experimental investigations on columns date back to the first half of the eighteenth century. A practical investigation of the phenomenon of lateral buckling was first undertaken by Mussghenbrock (Timoshenko 1953). As a result of this research, it was stated that the buckling load was inversely proportional to the square of the length of the strut. During the closing years of the eighteenth century, a series of investigations with wooden struts were made.

Hodgkinson in 1840 (Timoshenko 1953) tested specimens of cast iron. Cylindrical, solid and hollow specimens with rounded and flat ends were tested. For slender, solid struts, good agreement was found with Euler's formula. At the end of the nineteenth century, more experiments on buckling of columns were conducted by Bauschinger, Tetmajer, and Considere (Timoshenko 1953).

Several researchers studied the column buckling criteria in the first half of this century and found that the results showed column failure at tangent modulus loads. Templin and others (Templin et al. 1938) also discovered that practical tests on columns yielded capacities that were closer to tangent modulus theory than to the seemingly more refined reduced modulus theory.

As discussed earlier, the primary research on the strength of solid rounds was carried out between 1954 and 1965 at Lehigh University. The tests were performed primarily on stress relieved constructional alloy steel bars. Twenty-

seven bars, some with 69.9 mm (2-3/4 in) diameter and others with 190.5 mm (7-1/2 in) diameter were tested to failure as axially loaded columns. In the main phase of the research program, the effect of residual stresses and initial crookedness on the strength of the axially loaded columns was investigated. To the best of author's knowledge, no major research work in this area was undertaken after this time.

2.3 Buckling Theory and Experimental Investigation of Bracing

Bracings are widely used in many structures. Bracings in wooden bridge trusses and roof trusses were used by the Romans. The first metal trusses in England were built in 1845 (Timoshenko 1953).

The first experiments on lattice girders, treated as cantilevers, were made in 1857 by Lohse (Todhunter 1960). Single, double, triple, and fourfold types of bracings were used. The bars were riveted to each other. The loads at which the bracing buckled were noted. It is noteworthy that in several cases the bracing bars failed elastically into an approximate S-form, a result which the researchers at the time did not take into account in their theoretical analysis. From these experiments, a great increase of strength, due to multiple bracing and to the riveting together of the bracing bars, was noticed.

Wohler in 1855 (Todhunter 1960) deduced that the stresses in the bracing bars of a girder from purely statistical consideration. De Clercq in 1857 and Winkler in 1859 theoretically analyzed lattice girders and the stresses in the bracing bars (Todhunter 1960).

For cross-braced or tension-compression diagonals, Jasinsky was the first to investigate the stability of the compressed diagonals and to evaluate the strengthening afforded by the diagonals in tension in lattice trusses (Timoshenko 1953).

Dewolf and Pelliccione (1979) reported that the design practice for cross-braced members adopted at that time, which entirely neglected the contribution of the compression member and relied solely on the tension member, was conservative and would result in overdesign. Eight sets of cross-bracings, using flat bars with rectangular cross section, were tested to failure. A square frame was used with connections at each corner of the frame designed to allow all members to rotate in the plane of the frame and for the two diagonals to rotate out of the plane. Strain gauges were placed in pairs on opposite sides of the members on the faces parallel to the plane of the frame. From the tests in that investigation, it was observed that using the design practice, which neglects the contribution of compression members, predicted frame load ranged from 40% to 78% of the failure load. The critical load of the compression member about its in-plane axis is related to the force in the tension member that braces it at the center, and when the two members are made of the same material, the tension member is equivalent to an unyielding support. Thus the compression member buckles into a full sine wave, S-shape, at a load equal to four times that without any center bracing. The stiffness, α , furnished by the tension member acting on compression member in the out-of-plane direction was given as follows:

$$\alpha = \frac{48EI}{L^3} \left(\frac{\mu^3}{3(\mu - \tanh\mu)} \right) \quad (2.4)$$

where L represents the total length of the diagonal, and

$$\mu = \frac{L}{2} \sqrt{\frac{T}{EI}} \quad (2.5)$$

in which T is the force in tension member.

It was also stated that following buckling of the compression member, its load decreased and the load in the tension member then increased with an overall increase in frame load. Predicted frame loads, based on the critical load in the compression member determined by using the tension member as a brace, were within 10% of the test load for compression members with low slenderness ratios and on the conservative side for those with higher ratios.

A theoretical investigation was made by Vickers (1982) into the behavior and design of cross-bracing. His discussion emphasized the use of cruciform (star shaped) double angle struts for bracing members. The extent of lateral support provided to the compression diagonal by the tension diagonal at their point of intersection was analyzed. From his study, it was concluded that the design concept of shared load between tension and compression members with the compression member designed using effective length equal to half the total length of the diagonals is more realistic than "tension only" bracing design.

El-Tayem and Goel (1986) studied experimentally and theoretically full-scale cross-bracing specimens. Five single-angle cross-braced specimens and one double-angle cross-braced specimen were included in the study. Quasi-static cyclic loading was used in the tests and strain gauges were attached to measure the loads. It was noted that the interconnection provided an elastic restraint against both lateral and rotational deformations of the compression diagonal at the point of intersection. It was concluded that for cross-braced systems made from single equal-leg angles, an effective length of 0.85 times the half diagonal length is reasonable.

Picard and Beaulieu (1987) did a theoretical study aimed at the determination of the transverse stiffness offered by the tension diagonal in cross-braced systems and at the evaluation of this stiffness on the out-of-plane buckling resistance of the compression diagonal. When the diagonals were continuous and attached at the intersection, it was concluded that the effective length of the compression

diagonal is 0.5 times the total diagonal length. A simpler form of equation was given by the writers for the stiffness, α , provided by the tension diagonal, assuming the two diagonals to be equal in length and to have the same cross-sectional area:

$$\alpha = \frac{48 EI}{L^3} + 4.36 \frac{T}{L} \quad (2.6)$$

where, L is the total length of the diagonal, and T is the force in tension diagonal. They also suggested that the effective length factor, K , for calculating the buckling load of the compression diagonal to be as follows:

$$K = \sqrt{0.523 - \frac{0.428}{C/T}} \geq 0.5 \quad (2.7)$$

where C and T are the forces, just before buckling, in the compression and the tension diagonals respectively.

Picard and Beaulieu (1988) performed two series of tests to demonstrate the validity of their theoretical study. Seven transverse stiffness tests and fifteen buckling tests were performed on flat bars and the results verified the validity of the equations presented by them.

The theoretical analysis carried out in 1987 by the same authors was generalized in 1989 (Picard and Beaulieu 1989), which gives the following the effective length factor for cross bracings:

$$K = \sqrt{\frac{1 - 0.818 (C/T)}{1 + 0.911 (I_t / I_c)}} \quad (2.8)$$

where, C/T is the ratio of the force in compression member just before buckling to the force in tension diagonal, and I_t/I_c is the ratio of moment of inertia of the

tension diagonal to the moment of inertia of the compression diagonal. This theoretical study was also verified by fifteen buckling tests carried out in 1988 (Picard and Beaulieu 1988).

Kemp and Behncke (1998) described a series of tests on cross-braced systems with slenderness ratios in the range of 102 to 160. Other variables included the inclination of the main legs and bracing, the number of bolts in each end connection, and the size of the main leg relative to the bracing. The measured behaviour was compared with the results of a flexibility-based analysis and the formulas from the American and European Transmission Tower Design manuals. The results confirmed the complexity of the behavior of cross-bracing in latticed towers. Strain measurements showed that yielding of the extreme fiber of the strut in the central region of the largest sub-span is the primary cause of failure. The effect of the end eccentricity was partially alleviated by the restraint provided by the main legs to the ends of the compression diagonal. Consequently the ultimate strength in the tests was increased up to 17% by changing the number of bolts at the end connection from one to two. A smaller but nevertheless significant, 10%, benefit was obtained by increasing the size ratio of the main leg relative to the bracing.

Jaboo (1998) conducted an experimental investigation on 8 single-braced and 18 cross-braced towers to determine effective length factors for solid round members. He found effective length factor, $K = 0.525$ for single-braced solid round diagonal, $K=0.429$ and 0.365 for type I and type II cross-braced solid round members, respectively.

2.4 Cross-Bracing as a Continuous Beam on Elastic Support

Connections in trusses and frames are treated in many different ways, but the traditional assumption of considering these connections to be perfectly hinged joints in an idealized frame made the design of such frames or trusses simpler.

In cross-braced diagonals, the tension diagonal acts as an elastic spring at the point of intersection with the compression diagonal. Thus the compression diagonal could be assumed as a continuous bar simply supported at the ends and having an intermediate elastic support.

Timoshenko and Gere (1961) discussed such a problem. A case where the intermediate elastic support is at the middle and the axial force does not change within the two halves of the member was explained. It was shown that if the stiffness of the elastic support, α , approaches infinity, the deflected shape of a bar on three hinged supports will be a full-sine wave, and the critical load, C_{cr} , for the compression member will be:

$$C_{cr} = \frac{\pi^2 EA}{\left(\frac{0.5 L}{r}\right)^2} \quad (2.9)$$

And when α approaches zero assuming the intermediate support to be absolutely flexible, the shape of the deflection curve of the buckled bar will be as half sine wave, and the critical load, C_{cr} , will be:

$$C_{cr} = C_e = \frac{\pi^2 EA}{\left(\frac{L}{r}\right)^2} \quad (2.10)$$

The limiting value of the stiffness of the elastic support, at which the full sine wave shape of the buckled bar occurs, is given as below:

$$\alpha_{lim} = \frac{16 \pi^2 EI}{L^3} \quad (2.11)$$

For values of α smaller than α_{lim} , the flexibility of the intermediate support should be considered, and it is shown that the relation between the elastic critical load C_{cr} and the stiffness of the elastic support α is:

$$C_{cr} = C_e + \frac{3\alpha L}{16} \leq 4C_e \quad (2.12)$$

It has to be mentioned that the assumption of hinged joints does not represent correctly all the types of connections in trusses or frames as the rotational restraint exists in cases where the joints are welded or even bolted. But because of the complexity associated, especially with the degree of rotational constraint provided by different types of connections, it is always preferable to assume a hinged end condition which is a conservative assumption.

2.5 Design Specifications for Cross-Bracings Based on CSA-S37-94 and AISC-LRFD

2.5.1 CSA-S37-94 Standard

CSA-S37-94 is the Canadian Standard for communication towers. The maximum slenderness ratio for bracing members is limited to 200 as stated in Clause 6.1.5.2 of CSA-S37.

For the computation of compressive strength of solid rounds as single bracing in towers, Clause 6.2.2.1 of the Standard states:

“The unbraced length, L_d of compression members other than leg members, shall be the distance along the axis of the member to which it is attached. For simple web bracing systems, with web member connected directly or by gussets to the leg member, the slenderness ratio is L_d/r , where, L_d is the length of the diagonal, and r is the radius of gyration.”

As for cross bracing, Clause 6.2.2.2 of the Standard states:

“For tension-compression web systems the diagonals shall be connected where they intersect and the lengths of the compression member (L_1 and L_2 in Figure 2.5.1) shall be the distances from the intersection of the two diagonal members to the center of the leg members at the ends of the compression member.”

The effective slenderness ratio of the compression member is KL/r , where $KL = L_1 + 0.5L_2$ (L_1 larger than L_2), and r = radius of gyration.

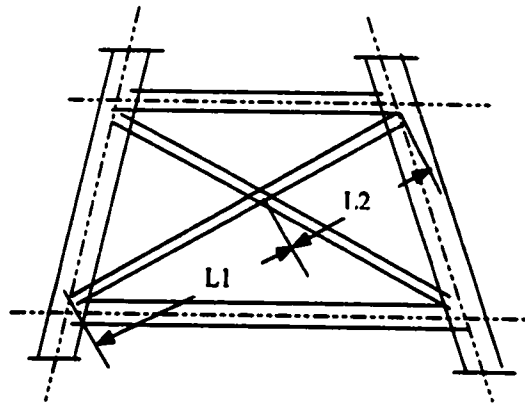


Figure 2.5.1 Tension-Compression Web System

After determining the slenderness ratio of the member, the nondimensional slenderness parameter, λ , is calculated as:

$$\lambda = \frac{KL_d}{r} \sqrt{\frac{F_y}{E \pi^2}} \quad (2.13)$$

This nondimensional slenderness parameter, λ , is used in computing factored compressive resistance C_r of the member:

$$C_r = \phi A F_y \left(1 + \lambda^{2n}\right)^{-\frac{1}{n}} \quad (2.14)$$

where

ϕ resistance factor for compression

A cross-sectional area of the member

F_y yield stress

λ nondimensional slenderness parameter

$n = 1.34$

2.5.2 AISC-LRFD Specifications

Formula given by above specifications is as follows:

$$\lambda = \frac{KL}{r \pi} \sqrt{\frac{F_y}{E}} \quad (2.15)$$

For

$$\lambda \leq 1.5,$$

$$F_{cr} = (0.685 \lambda_c^2) F_y \quad (2.16)$$

$$\lambda > 1.5,$$

$$F_{cr} = \left\{ \frac{0.877}{\lambda_c^2} \right\} F_y \quad (2.17)$$

where

F_y = specified yield stress

E = modulus of elasticity

K = effective length factor

L = unbraced length of member

r = governing radius of gyration about plane of buckling

CHAPTER 3

EXPERIMENTAL INVESTIGATION

3.1 General

For this investigation, tests were conducted on thirty-three specimens that were fabricated by three different American tower manufactures. Fifteen of the specimens were six-panel cross-braced towers with seven of the specimens having horizontals and the others no horizontals. Sixteen of the specimens were eight-panel cross-braced towers with half of the specimens having horizontals and the other half no horizontals. Finally, the last two specimens were six-panel single-braced towers with flat horizontals. Three sizes of legs (38.1 mm, 50.8 mm, 69.9 mm diameter), three sizes of diagonals (12.7 mm, 14.3 mm, 15.9 mm diameter) and one size of horizontal (19.1 mm diameter) were used for the six-panel towers. Similarly, the eight-panel towers were fabricated with three sizes of legs (38.1 mm, 50.8 mm, 57.2 mm diameter) and three sizes of diagonals (15.9 mm, 19.1 mm, 22.2 mm diameter). Two sizes of diagonals (28.6 mm, 22.2 mm diameter) and two sizes of legs (50.8 mm and 76.2 mm diameter) were used for the two single-braced specimens.

3.2 Details of Specimens

3.2.1 Cross-Braced Specimens

Fifteen six-panel specimens and sixteen eight-panel specimens are included in the investigation. They all have a 914.4 mm triangular section with all legs, horizontals (where present), and diagonal members connected together by welding. The six-panel specimens have a total length of 4572 mm (one panel of 762 mm) with the diagonals continuously-bent and welded at the leg junction. The eight-panel specimens have a total length of 5690 mm (one panel length of 711 mm) with the diagonals cut and welded at the leg junction. Each pair of cross-braced diagonals is welded together at their junction. The details of the specimens are shown in Tables 3.2.1a and b and Figures 3.2.1a and b.

Table 3.2.1a: DETAILS OF TEST SPECIMENS

6-PANEL CROSS-BRACED SPECIMENS

Serial No.	Test No.	Specimen I.D.	Span Length (mm)	Panel Length (mm)	Face Width (mm)	Leg Size (mm)	Diagonal Size (mm)	Horizontal Size (mm)	Fabricator
1	6	S1A	4572	762	914.4	38.1	12.7	No	ERI, Inc., Chandler, IN
3	4	S2A	4572	762	914.4	38.1	12.7	19.1	ERI, Inc., Chandler, IN
4	8	S2B	4572	762	914.4	38.1	12.7	19.1	ERI, Inc., Chandler, IN
5	23	S2C	4572	762	914.4	38.1	12.7	19.1	ERI, Inc., Chandler, IN
6	7	S3A	4572	762	914.4	50.8	14.3	No	ERI, Inc., Chandler, IN
7	25	S3B	4572	762	914.4	50.8	14.3	No	ERI, Inc., Chandler, IN
8	1	S4A	4572	762	914.4	50.8	14.3	19.1	ERI, Inc., Chandler, IN
9	3	S4B	4572	762	914.4	50.8	14.3	19.1	ERI, Inc., Chandler, IN
10	5	S4C	4572	762	914.4	50.8	14.3	19.1	ERI, Inc., Chandler, IN
11	2	S5A	4572	762	914.4	69.9	15.9	No	ERI, Inc., Chandler, IN
12	10	S5B	4572	762	914.4	69.9	15.9	No	ERI, Inc., Chandler, IN
13	12	S5C	4572	762	914.4	69.9	15.9	No	ERI, Inc., Chandler, IN
14	11	S6A	4572	762	914.4	69.9	15.9	19.1	ERI, Inc., Chandler, IN
15	13	S6B	4572	762	914.4	69.9	15.9	19.1	ERI, Inc., Chandler, IN
16	24	S6C	4572	762	914.4	69.9	15.9	19.1	ERI, Inc., Chandler, IN

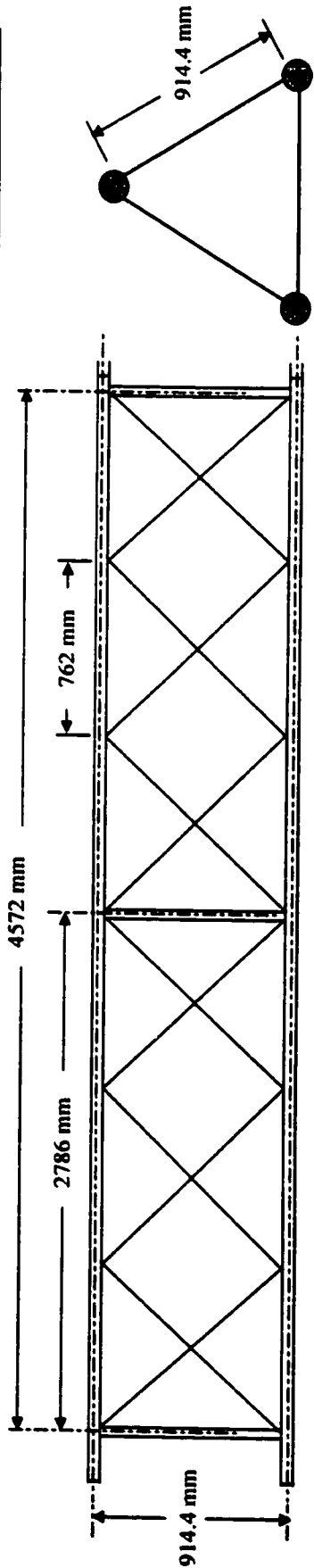


Figure 3.2.1a: DIMENSIONS OF 6-PANEL SPECIMEN

CROSS-SECTION

**Table 3.2.1b: DETAILS OF TEST SPECIMENS
(8-PANEL CROSS-BRACED SPECIMENS)**

Serial No.	Test No.	Specimen I.D.	Span Length (mm)	Panel Length (mm)	Face Width (mm)	Leg Size (mm)	Diagonal Size (mm)	Horizontal	Fabricator
1	22	P1A	5690	711.2	914.4	38.1	15.9	Yes	PIROD Inc., Plymouth, IN
2	31	P1B	5690	711.2	914.4	38.1	15.9	Yes	PIROD Inc., Plymouth, IN
3	21	P2A	5690	711.2	914.4	38.1	15.9	No	PIROD Inc., Plymouth, IN
4	27	P2B	5690	711.2	914.4	38.1	15.9	No	PIROD Inc., Plymouth, IN
5	30	P3A	5690	711.2	914.4	50.8	19.1	Yes	PIROD Inc., Plymouth, IN
6	32	P3B	5690	711.2	914.4	50.8	19.1	Yes	PIROD Inc., Plymouth, IN
7	29	P4A	5690	711.2	914.4	50.8	19.1	No	PIROD Inc., Plymouth, IN
8	20	P4B	5690	711.2	914.4	50.8	19.1	No	PIROD Inc., Plymouth, IN
9	26	P5A	5690	711.2	914.4	57.2	22.2	Yes	PIROD Inc., Plymouth, IN
10	34	P5B	5690	711.2	914.4	57.2	22.2	Yes	PIROD Inc., Plymouth, IN
11	28	P6A	5690	711.2	914.4	57.2	22.2	No	PIROD Inc., Plymouth, IN
12	15	P6B	5690	711.2	914.4	57.2	22.2	No	PIROD Inc., Plymouth, IN
13	18	P7A	5690	711.2	914.4	38.1	22.2	Yes	PIROD Inc., Plymouth, IN
14	17	P7B	5690	711.2	914.4	38.1	22.2	Yes	PIROD Inc., Plymouth, IN
15	33	P8A	5690	711.2	914.4	38.1	22.2	No	PIROD Inc., Plymouth, IN
16	19	P8B	5690	711.2	914.4	38.1	22.2	No	PIROD Inc., Plymouth, IN

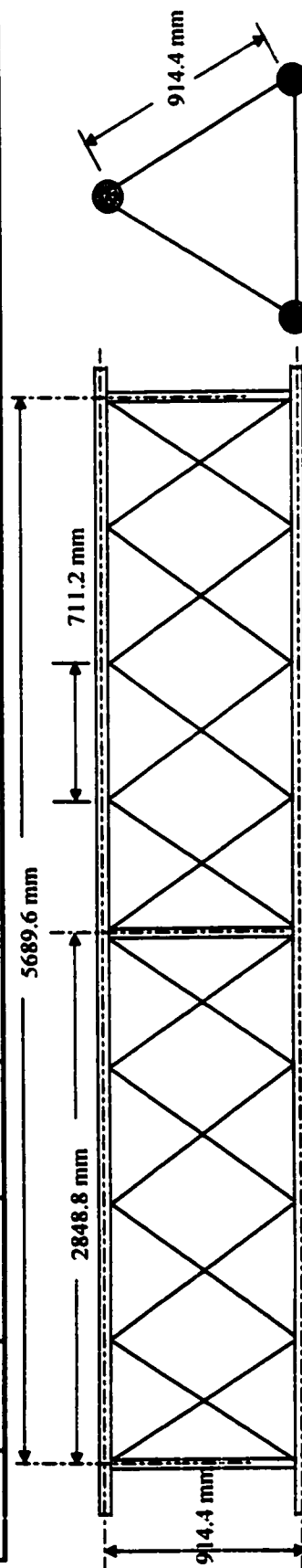


Figure 3.2.1b DIMENSIONS OF 8-PANEL SPECIMEN

CROSS-SECTION

3.2.2 Single-Braced Specimens

Two six-panel single-braced specimens were tested in the investigation. They also have a triangular cross-section with all legs, horizontals and diagonal members welded together. The specimens have a total length of 6096 mm (one panel length of 1016 mm). The details of the specimens are given in Table 3.2.2 and Figure 3.2.2.

3.3 Test Set-up

3.3.1 Specimen Set-up

The tests of specimens were carried out in the Structural Laboratory, University of Windsor, with the specimens tested as horizontal simply supported beams, loaded with a central concentrated load which was provided by a hydraulic jack attached to the steel test frame. A laboratory crane was used to set up the test specimens in proper position. i.e. the specimens were seated on the two end supports with the middle of the specimen just under the load jack.

3.3.2 Reinforcement of Compression Diagonals

All compression diagonals in panels, other than the one under investigation, were reinforced with four U-bolted angles to make sure that the diagonals of interest would buckle. Details of specimen reinforcement are given in Figures 3.3.2a and b.

3.3.3 Strain Gauges

Electric resistance strain gauges, fabricated by KYOWA, JAPAN, type KFG-5-120-c1-11, with a gauge length of 5 mm and a gauge factor of 2.12 were used to measure the strain in the members of the test panel of the specimen.

Table 3.2.2: DETAILS OF TEST SPECIMENS
(6-PANEL SINGLE-BRACED SPECIMENS)

Serial No.	Test No.	Specimen I.D.	Span Length (mm)	Panel Length (mm)	Face Width (mm)	Leg Size (mm)	Diagonal Size (mm)	Horizontal	Fabricator
1	14	31SRB	6096	1016	787	50.8	22.23	Yes	Rohn Inc., Peoria, Illinois, USA
2	16	33SRC	6096	1016	838	76.2	28.58	Yes	Rohn Inc., Peoria, Illinois, USA

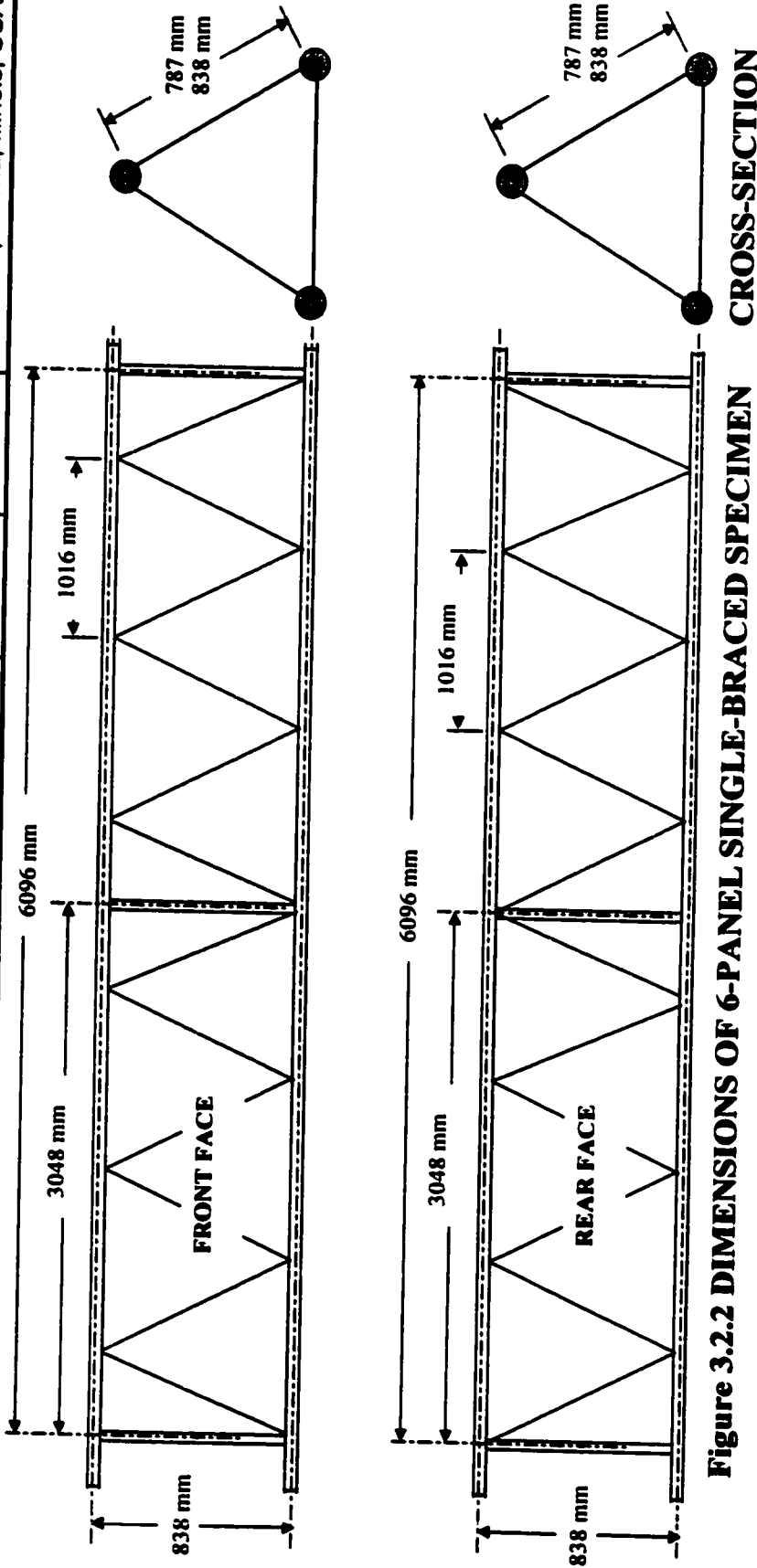


Figure 3.2.2 DIMENSIONS OF 6-PANEL SINGLE-BRACED SPECIMEN

CROSS-SECTION

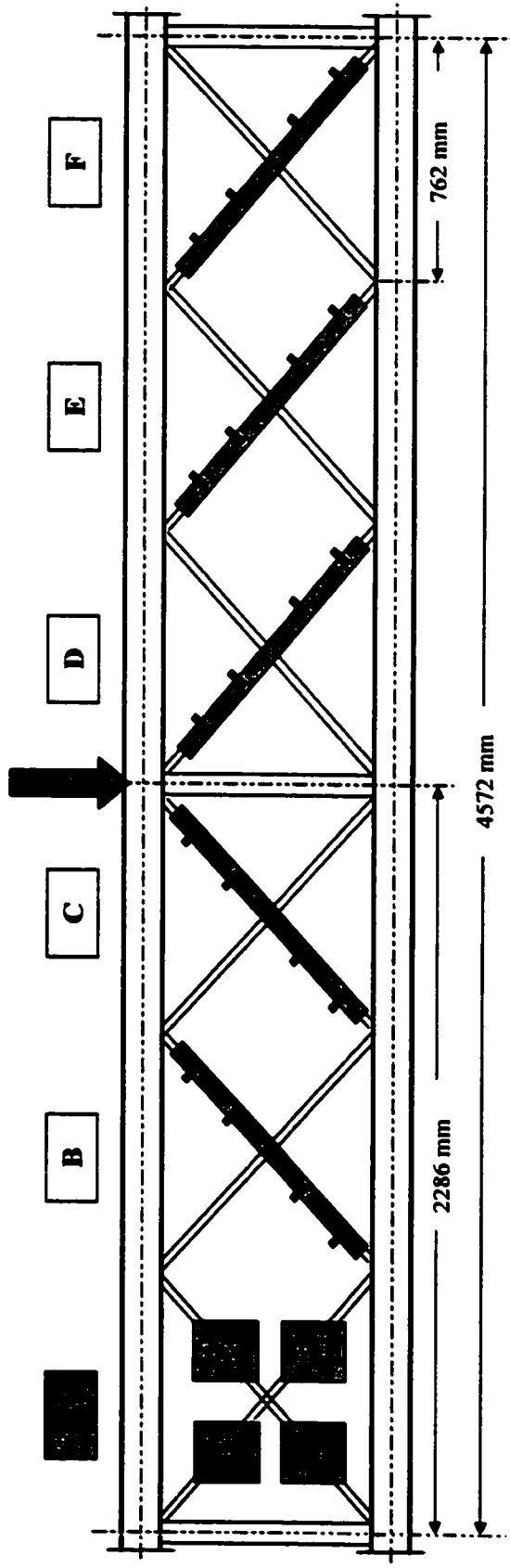


Figure 3.3.2a Test Panel A (strain gauges attached). Compression Diagonals in panels B, C, D, E and F were Reinforced with Angles as Shown

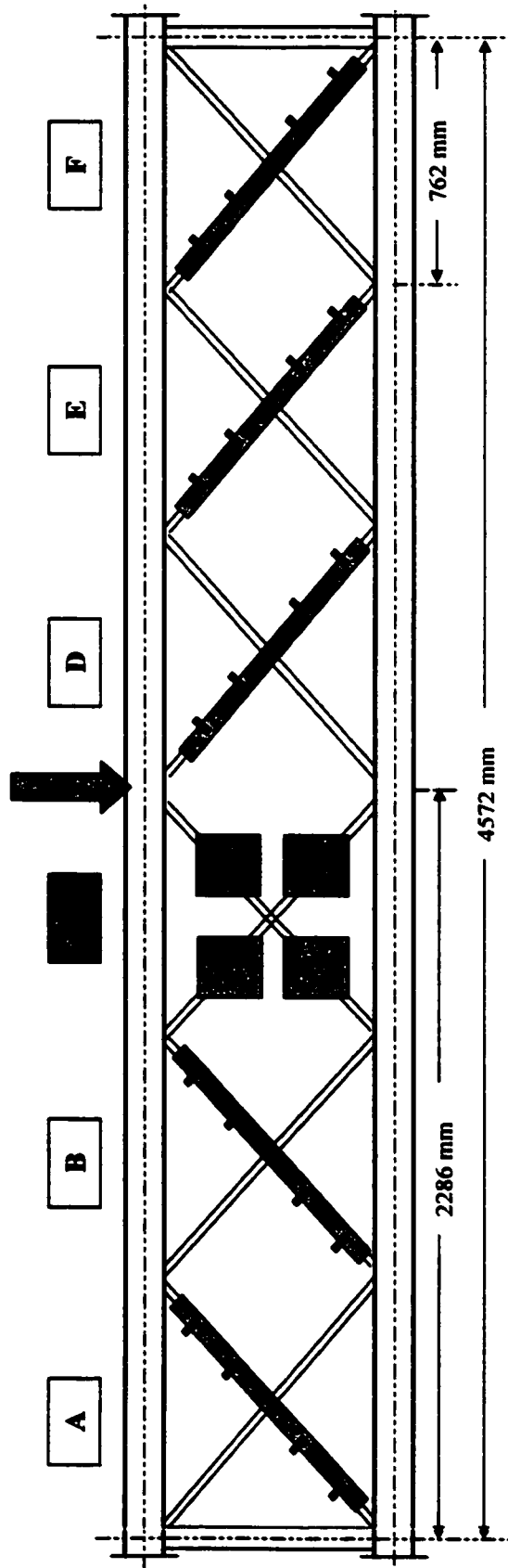


Figure 3.3.2b: Test Panel C (strain gauges attached). Compression Diagonals in Panels A, B, D, E and F were Reinforced with Angles as Shown

The strain gauges were fixed to individual members of the test specimen at the proper position after the surface was properly polished, cleaned with chemicals, and dried with Q-tips.

Based on previous experience of buckling of diagonal members, all the strain gauges were located at one-third the diagonal length, measured from the leg. For the first specimen, each panel was tested with 18 strain gauges to determine the force distribution in the diagonals and legs of the specimen whereas 14 strain gauges are used for each panel of the second specimen. For all other specimens, only the one pre-selected failure panel was tested with 16, 24 or 32 strain gauges fixed on both the compressive and tensile diagonals (strain gauges were put on legs for a few cases). Details of strain gauge location and numbering for each specimen are shown in Appendix A (Strain gauge location on specimens).

3.3.4 Data-Acquisition System

Readings of load cell and strain gauges were recorded automatically in the hardware of the computer as a text file with Datascan data acquisition system. The load cell and strain gauges were connected to the Datascan channels twelve hours after the strain gauges were fixed to the specimens. Microsoft Word and Excel programs were used to transfer the text file into Excel table for the convenience of data output and axial force calculations.

3.3.5 Load Application

The load was applied with a mechanical pump through a hydraulic jack attached to the beam of the steel test frame. Previously calibrated 445 kN (100 kip) and 890 kN (200 kip) load cells were used to measure the applied load. Details of load application are shown in Figures 3.3.5a and b.

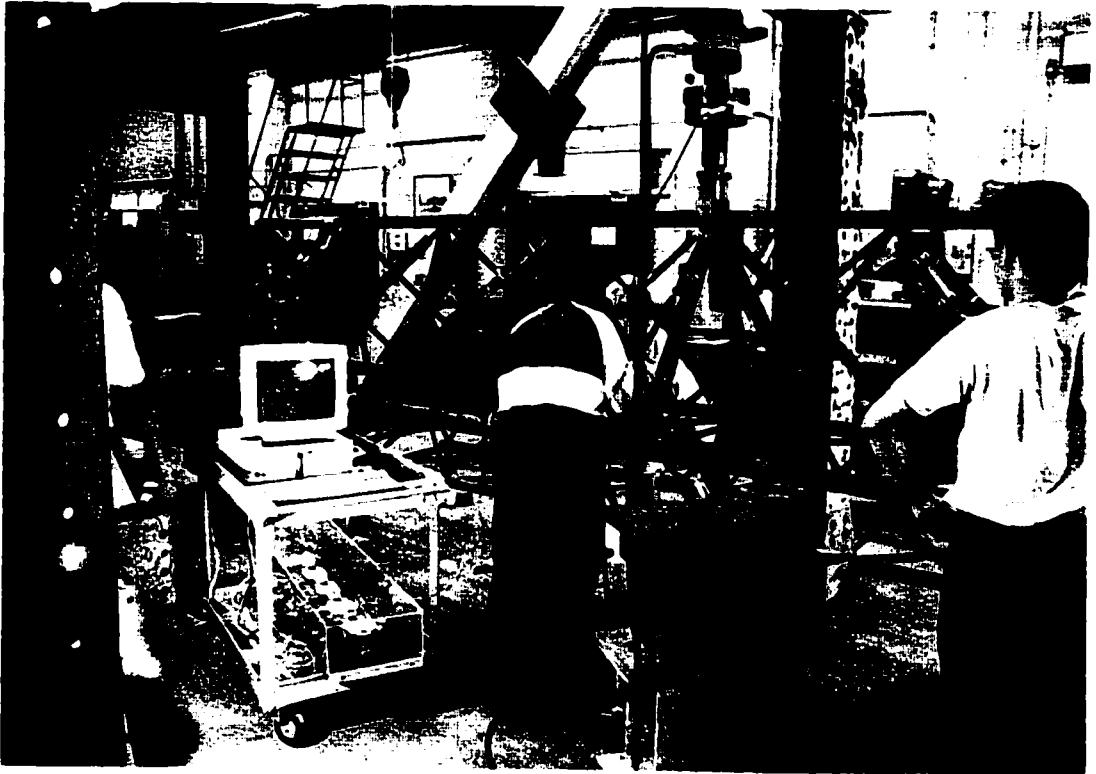


Figure 3.3.5a: Photograph of Test Set-up

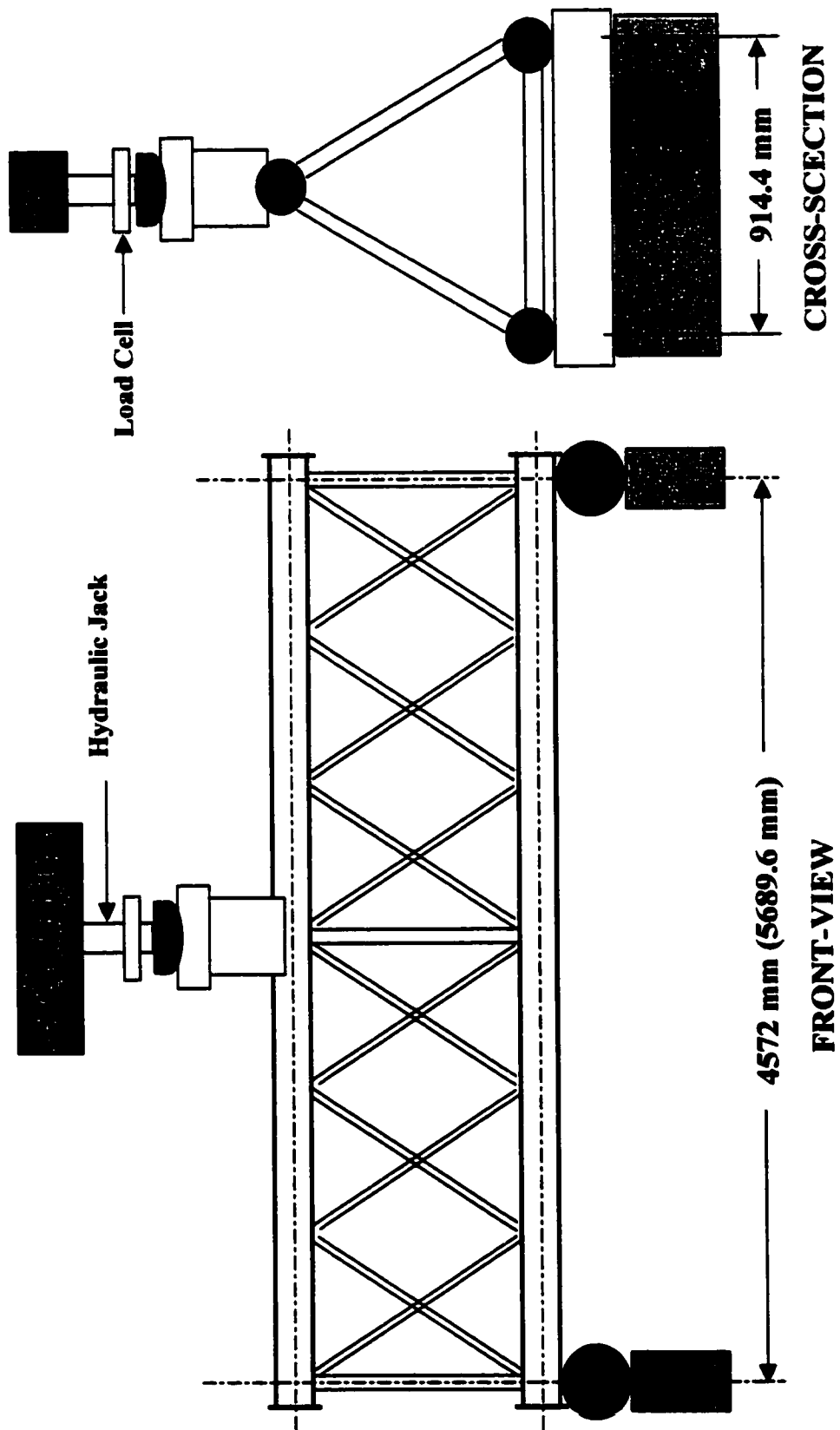


Figure 3.3.5b: Test Set-Up (not to scale)

Before starting the test, a Tinius Olsen Universal Testing Machine and a strain indicator were used to calibrate the load cells. Calibration data and curves are shown in Tables 3.3.5a and b and Figures 3.3.5c and d.

In order to get an accurate value of the axial forces based on the readings of strain gauges, care was taken in positioning the strain gauges to ensure that they are diametrically opposite.

3.4 Experimental Results

3.4.1 Structural Response

With the application of a central concentrated load, the axial forces and bending moments in the tested diagonals increase until the tested diagonals buckled. During the tests, the buckled diagonals and their buckling sequences were noted and are presented in Table 4.1.1. The load increments towards the end were kept small by monitoring the maximum strain readings of the test diagonals. The strain readings and axial forces are shown in Appendix B (Load-Strain Data and Axial Forces).

3.4.2 Material Properties

A total of 132 diagonal tensile specimens were cut from the compression diagonals (four tensile specimens for each tower), and tested in the Tinius Olsen Universal Testing Machine. The yield load and maximum failure load were obtained and the yield stress and tensile strength for each specimen were determined. The values of yield stress and tensile strength for each tower are given in Tables 3.4.2a, b and c.

Table 3.3.5.a: Calibration Data for 445 kN (100 kip) Load Cell

LOAD(kN)	MICRO-STRAIN READINGS			AVERAGE
	READING1	READING2	READING3	
0	0	0	0	0
40	351	367	358	359
80	702	720	710	711
120	1059	1073	1062	1065
160	1405	1421	1412	1413
200	1762	1771	1762	1765
240	2098	2118	2109	2108
280	2446	2464	2454	2455
320	2797	2812	2801	2803
360	3147	3162	3148	3152
400	3500	3506	3497	3501.00

**Figure 3.3.5.c: Calibration Curve for 100 kip Load Cell
(gauge factor 2.062)**

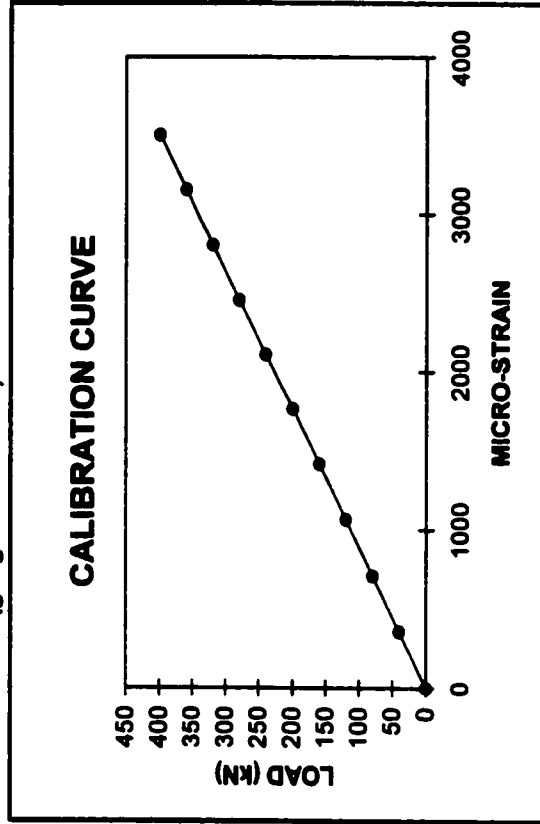


Table 3.3.5.b: Calibration Data for 890 kN (200 kip) Load Cell

LOAD(kN)	MICRO-STRAIN READINGS
0	0
10	30
20	65
30	108
40	151
50	194
60	238
70	281
80	325
90	368
100	411
150	628
200	847
250	1067
300	1285
350	1505
400	1724
450	1944
500	2164

**Figure 3.3.5.d: Calibration Curve for 200 kip Load Cell
(gauge factor 2.062)**

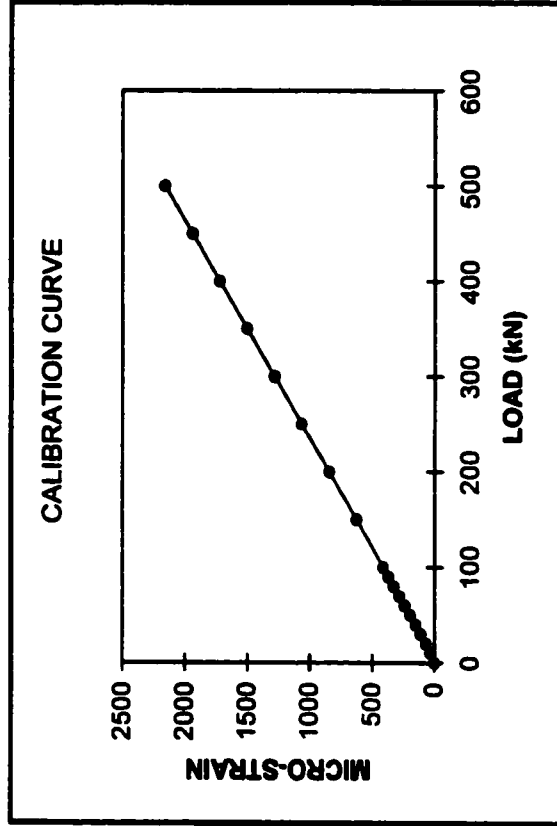


Table 3.4.2a: YIELD STRESSES AND TENSILE STRENGTHS OF DIAGONALS

(6-PANEL CROSS-BRACED SPECIMENS)

TEST SERIAL NO.	SPECIMEN I.D.	DIAMETER OF DIAGONAL (mm)	AVERAGE MEASURED YIELD STRESS (MPa)	AVERAGE MEASURED TENSILE STRENGTH (MPa)
6	S1A	12.7	367	488
9	S1B	12.7	396	502
4	S2A	12.7	378	498
6	S2B	12.7	369	498
23	S2C	12.7	363	478
7	S3A	14.3	385	532
25	S3B	14.3	398	532
1	S4A	14.3	378	535
3	S4B	14.3	378	535
6	S4C	14.3	367	527
2	S5A	15.9	385	504
10	S5B	15.9	385	504
12	S5C	15.9	378	535
11	S6A	15.9	385	502
13	S6B	15.9	382	502
24	S6C	15.9	385	502

Table 3.4.2b: YIELD STRESSES AND TENSILE STRENGTHS OF DIAGONALS

(8-PANEL CROSS-BRACED SPECIMENS)

TEST SERIAL NO.	SPECIMEN I.D.	DIAMETER OF DIAGONAL (mm)	AVERAGE MEASURED YIELD STRESS (MPa)	AVERAGE MEASURED TENSILE STRENGTH (MPa)
22	P1A	15.9	363	488
31	P1B	16.9	375	606
21	P2A	15.9	373	603
27	P2B	16.9	373	603
30	P3A	19.1	407	681
32	P3B	19.1	409	696
29	P4A	19.1	410	697
20	P4B	19.1	409	692
26	P5A	22.2	437	673
34	P5B	22.2	437	673
28	P6A	22.2	442	676
16	P6B	22.2	442	676
18	P7A	22.2	381	486
17	P7B	22.2	343	494
33	P8A	22.2	346	486
19	P8B	22.2	349	493

Table 3.4.2c: YIELD STRESSES AND TENSILE STRENGTHS OF DIAGONALS

(6-PANEL SINGLE-BRACED SPECIMENS)

TEST SERIAL NO.	SPECIMEN I.D.	DIAMETER OF DIAGONAL (mm)	AVERAGE MEASURED YIELD STRESS (MPa)	AVERAGE MEASURED TENSILE STRENGTH (MPa)
14	318RB	22.2	328	490
16	338RC	28.6	364	510

CHAPTER 4

ANALYSIS OF RESULTS

4.1 Failure Pattern of Tower Members

4.1.1 Cross-Braced Specimens

Thirty-one cross-braced specimens were tested in this investigation; fifteen of the specimens were with horizontals while the others were without horizontals. For the specimens with horizontals, the left-most panel was tested with all other panels reinforced and it was observed that the tested compression diagonals had buckled out-of-plane with an almost S-shape or an unsymmetrical S-shape. For the specimens without horizontals, left middle panel was strain-gauged while all other panels were reinforced. The tested diagonals had buckled with one half-sine wave failure shape due to very small tensile force in the tensile diagonals. (The failure pattern of all the compression diagonals are given in Table 4.1.1).

4.1.2 Single-Braced Specimens

Two single-braced towers 31SRB and 33SRC were tested during the investigation. For tower 31SRB, the top leg of right middle panel buckled first due to the small leg-diagonal size ratio and no diagonals buckled. For tower 33SRC with a large leg-diagonal size ratio, two of the tested diagonals buckled out-of-plane almost at the same time with a one half sine wave shape.

4.2 Calculation of Axial Forces in Tested Members

The finite strip method was used to determine the axial forces, assuming a linear variation of strain across the cross-section. The cross-section of each tested member was divided into 1000 strips, and the axial force in each strip was determined from the strain in the strip.

Table 4.1.1: FAILURE PATTERN OF ALL BUCKLED COMPRESSION DIAGONALS

Test No.	Specimen I.D.	Location of Buckled Diagonal	Failure Pattern	Test No.	Specimen I.D.	Location of Buckled Diagonal	Failure Pattern
6	S1A	panel C front diagonal	half-sine inside	22	P1A	test panel did not buckle; panel H buckled first	
		panel C rear diagonal	did not buckle				
4	S2A	panel A front diagonal	s-shape top half outside	31	P1B	panel A front diagonal	bottom half outside
		panel A rear diagonal	s-shape bottom half inside			panel A rear diagonal	top half inside
8	S2B	panel A front diagonal	s-shape top half inside	21	P2A	panel A front diagonal	did not buckle
		panel A rear diagonal	s-shape bottom half outside			panel A rear diagonal	top half outside
23	S2C	panel A front diagonal	s-shape top half inside	27	P2B	panel D front diagonal	top half outside
		panel A rear diagonal	s-shape top half inside			panel D rear diagonal	top half outside
7	S3A	panel C front diagonal	top half outside	30	P3A	panel A front diagonal	top half inside
		panel C rear diagonal	bottom half inside			panel A rear diagonal	bottom half outside
25	S3B	panel A front diagonal	s-shape top half outside	32	P3B	panel A front diagonal	top half inside
		panel A rear diagonal	s-shape top half inside			panel A rear diagonal	bottom half outside
1	S4A	panel A front diagonal	s-shape top half inside	29	P4A	panel D front diagonal	not buckle
		panel B front diagonal	s-shape top half inside			panel D rear diagonal	top half inside
3	S4B	panel A front diagonal	s-shape top half inside	20	P4B	panel D front diagonal	top half inside
		panel A rear diagonal	s-shape bottom half inside			panel D rear diagonal	top half outside
5	S4C	panel A front diagonal	s-shape top half inside	26	P5A	test panel did not buckle	
		panel A rear diagonal	s-shape bottom half inside				
2	S5A	panel C front(rear) diagonal	half-sine outside(inside)	34	P5B	panel A front diagonal	did not buckle
		panel D front(rear) diagonal	half-sine outside(inside)			panel A rear diagonal	top half inside
10	S5B	panel C front diagonal	half-sine outside(inside)	28	P6A	panel D front diagonal	half-sine inside
		panel C rear diagonal	did not buckle			panel D rear diagonal	half-sine outside
12	S5C	panel C front diagonal	half-sine inside	15	P6B	panel D front diagonal	did not buckle
		panel C rear diagonal	half-sine outside			panel D rear diagonal	top half inside
11	S6A	panel A front diagonal	s-shape top half inside	18	P7A	panel C diagonalf did not buckle; panel D top leg buckled first	
		panel A rear diagonal	s-shape bottom half inside				
13	S6B	panel A front diagonal	s-shape top half inside	17	P7B	panel A diagonal did not buckle; panel E top leg buckled first	
		panel A rear diagonal	s-shape bottom half inside				
24	S6C	panel A did not buckle; panels C and D buckled first after the horizontal buckled		33	P8A	panel D front diagonal	top half outside
14	31SRB	no diagonal buckled due to the right middle panel top leg buckling		19	P8B	panel D rear diagonal	did not buckle
						panel D front diagonal	top half inside
16	33SRC	panel B rear diagonal	half-sine out-of-plane			panel D rear diagonal	did not buckle
		panel c rear diagonal					

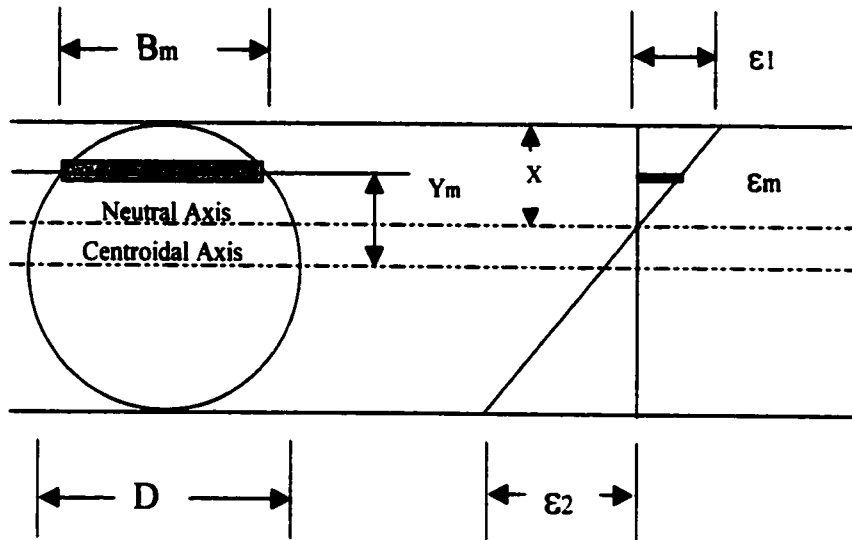


Figure 4.2.1. Diagram of Axial Force Calculation from Strains

The detailed calculation procedure is as follows:

Case 1: Strains at Top and Bottom are of Opposite Sign (Fig. 4.2.1)

The distance Y_m measured from the center of slice to centroidal axis of the cross-section is:

$$Y_m = \frac{D}{2} - (m-1)dy - \frac{dy}{2} \quad (4.1)$$

where $dy = D/1000$, and $m = 1$ to 1000 (from top to bottom)

The width of slice "m" is given by

$$B_m = 2\sqrt{\left(\frac{D}{2}\right)^2 - (Y_m)^2} \quad (4.2)$$

The depth of neutral axis X measured from the strain ϵ_1 is calculated as:

$$X = \frac{|\epsilon_1| D}{|\epsilon_1| + |\epsilon_2|} \quad (4.3)$$

The strain ϵ_m in the center of slice "m" is given by

$$\epsilon_m = \frac{\epsilon_1}{X} \left[Y_m - \left(\frac{D}{2} - X \right) \right] \quad (4.4)$$

The stress for slice "m" is calculated according to Hooke's law:

$$\sigma_m = \epsilon_m E \quad (4.5)$$

The actual force in slice "m" is

$$\text{If } \sigma_m < F_y \quad \text{then } F_m = \sigma_m B_m dy \quad (4.6)$$

$$\text{If } \sigma_m > F_y \quad \text{then } F_m = F_y B_m dy \quad (4.7)$$

The axial force in the total cross-section of the member is the sum of all the slice forces:

$$F = \sum F_m \quad (4.8)$$

Case 2: Strains at the Top and Bottom are of Same Sign (Figure 4.2.2)

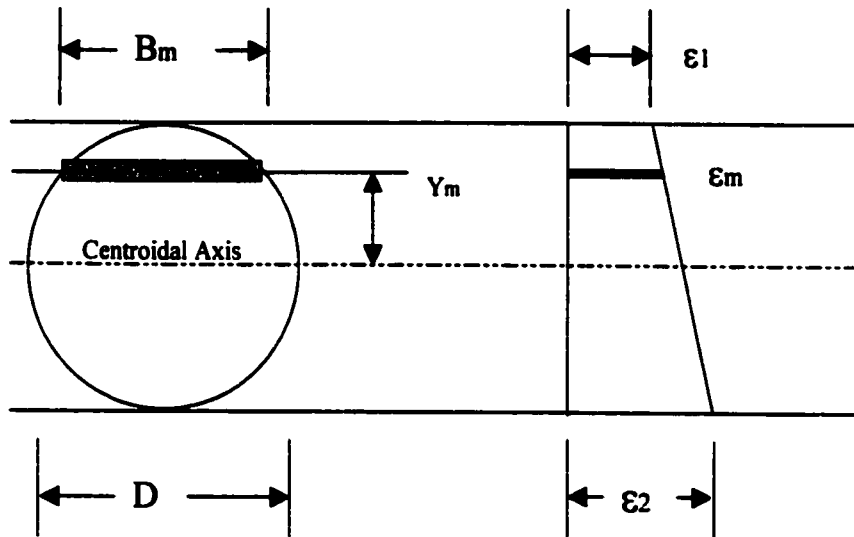


Figure 4.2.2. Diagram of Axial Force Calculation from Strains

The distance Y_m measured from the center of slice to centroidal axis of the cross-section is:

$$Y_m = \frac{D}{2} - (m-1)dy - \frac{dy}{2} \quad (4.9)$$

where $dy = D/1000$, and $m = 1$ to 1000 (from top to bottom)

The width of slice "m" is given by

$$B_m = 2\sqrt{\left(\frac{D}{2}\right)^2 - (Y_m)^2} \quad (4.10)$$

The strain ϵ_m in the center of slice "m" is given by

$$\epsilon_m = \frac{D(\epsilon_1 + \epsilon_2) - 2Y_m(\epsilon_2 - \epsilon_1)}{2D} \quad (4.11)$$

The stress for slice “m” is calculated according to Hooke’s law:

$$\sigma_m = \varepsilon_m E \quad (4.12)$$

The actual force in slice “m” is

$$\text{If } \sigma_m < F_y \quad \text{then } F_m = \sigma_m B_m dy \quad (4.13)$$

$$\text{If } \sigma_m > F_y \quad \text{then } F_m = F_y B_m dy \quad (4.14)$$

The axial force in the total cross-section of the member is the sum of all the slice forces:

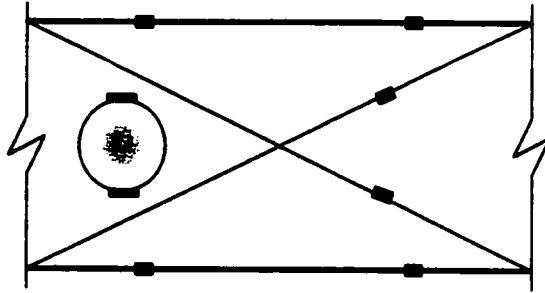
$$F = \sum F_m \quad (4.15)$$

4.3 Specimen Equilibrium Check

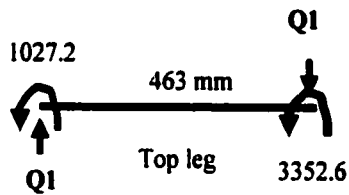
Accuracy of the axial forces based on the strain readings was verified with the tower’s equilibrium check. In this investigation, a vertical equilibrium check was conducted on one tower (S5C) by calculating the axial forces and shears in the tension and compression diagonals and the shear forces in the three legs at one cross-section. The location of strain gauges are given in Appendix A. The equilibrium was checked as shown below:

Specimen S5C equilibrium check (applied load level 106 kN):

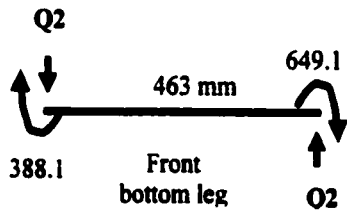
- 1) Diagram for strain gauge location (front-view)



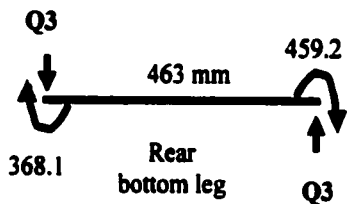
2) Shear forces in three legs (from Table S5C of Appendix B)



$$Q_1 = (3352.6 + 1027.2) / 463 = 9.46 \text{ kN}$$



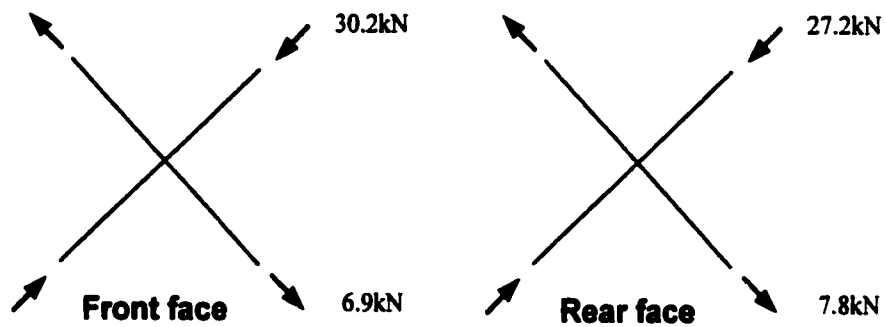
$$Q_2 = (649.1 + 388.1) / 463 = 2.24 \text{ kN}$$



$$Q_3 = (459.2 + 368.1) / 463 = 1.79 \text{ kN}$$

$$Q_{leg} = Q_1 + Q_2 + Q_3 = 9.46 - 2.24 - 1.79 = 5.43 \text{ kN}$$

3) Vertical component of axial forces in diagonals (from Table S5C of Appendix B)



$$Q_{\text{diagonal}} = (30.2 + 6.9 + 27.2 + 7.8) \cdot \sin(41.71^\circ) = 47.97 \text{ kN}$$

4) Vertical component of shear forces in diagonals (ignored)

5) Total shear force in the cross-section

$$Q_{\text{total}} = Q_{\text{leg}} + Q_{\text{diagonal}} = 5.43 + 47.97 = 53.40 \text{ kN}$$

6) Support reaction

$$R = \text{Load Level}/2 = 106.0/2 = 53.0 \text{ kN}$$

7) Error

$$\text{Error} = (Q_{\text{total}} - R)/R = (53.4 - 53.0)/53.0 = 0.75\%$$

4.4 Determination of Experimental Buckling Loads and Effective Length Factors

The buckled diagonals were noted during each tower test. The maximum axial force that occurred in the tested diagonal was taken as the buckling load. Only for specimen S5A, there was a large variation in the buckling loads of the front and rear diagonals (38.9 kN and 21.6 kN). This is probably due to the possibility of the strain gauges not being located diametrically opposite to each other on one of the diagonals. In this investigation, a total of 44 buckling loads were used and the corresponding effective length factors based on CSA-S37-94 Standard and AISC-LRFD Specifications were calculated. The details of the buckling loads and effective length factors are shown in Tables 4.4a to 4.4g.

The leg member of specimen 31SRB failed before any diagonal failed. The maximum axial compressive force recorded in the diagonals was 51.6 kN. This corresponds to an effective length factor of 0.64 according to CSA-S37-94 and 0.71 according to AISC-LRFD. Though it is not possible to determine the actual effective length factors of diagonal members based on this test, it can be stated that they will be less than the above values.

4.5 Effect of Axial Forces in Tension Diagonals on Effective Length Factors

Examining Tables 4.4a to 4.4f, it can be seen that although the ratio of forces in the compression diagonal and tension diagonal varied significantly, the effective length factors varied from a minimum of 0.296 to a maximum of 0.403. There is no definite relationship between the force in the tension diagonal and the resistance of the compression diagonal. The tension diagonals of the middle panels of specimens with no horizontals carried very little tensile forces. Still they were able to provide the necessary support to the compression diagonal. It can therefore be stated that even a small tensile force is able to provide the necessary support to the compression diagonal.

Table 4.4a: EFFECTIVE LENGTH FACTORS USING CAN/CSA-S16.1-94

(6-panel cross-braced specimens)										
Test No.	Specimen I.D.	Diagonal Diameter (mm)	c/c Length of Diagonal Ld (mm)	Measured Yield Stress Fy (MPa)	Buckling Load (kN)	Force in the Tension Diagonal (kN)	Compression/Tension Ratio	λ	Effective Length Factor K	
6	S1A	12.7	1190	357	14.1	2.6	5.42	1.640	0.325	
9	S1B	12.7	1190	398	12.4	3.3	3.76	1.776	0.352	
4	S2A	12.7	1190	378	Did not buckle					
8	S2B	12.7	1190	369	14.8	15.0	0.99	1.649	0.318	
23	S2C	12.7	1190	353	14.3	16.0	0.89	1.665	0.325	
7	S3A	14.3	1190	385	13.7	13.6	1.01	1.706	0.333	
25	S3B	14.3	1190	398	13.4	13.4	1.02	1.708	0.333	
1	S4A	14.3	1190	376	14.5	16.6	0.87	1.598	0.319	
3	S4B	14.3	1190	378	14.8	14.0	1.08	1.578	0.315	
5	S4C	14.3	1190	367	21.3	4.6	4.63	1.537	0.331	
2	S5A	15.9	1190	385	19.4	8.1	2.52	1.633	0.351	
10	S5B	15.9	1190	395	20.4	4.6	4.37	1.617	0.342	
12	S5C	15.9	1190	385	20.1	16.9	1.19	1.633	0.345	
11	S6A	15.9	1190	385	20.1	16.9	1.19	1.578	0.343	
13	S6B	15.9	1190	382	24.9			1.365	0.296	
24	S6C	15.9	1190	385	21.4	19.3	1.11	1.485	0.327	
					22.8	22.0	1.04	1.422	0.313	
					21.6	1.9	11.37	1.741	0.416	
					20.9	8.1	2.58	1.776	0.424	
					33.2	10.4	3.19	1.306	0.312	
					29.0	6.9	4.20	1.439	0.343	
					30.2	9.8	3.08	1.399	0.334	
					27.3	23.9	1.14	1.468	0.358	
					29.2	13.8	2.12	1.431	0.342	
					25.4	17.3	1.47	1.563	0.375	
					Did not buckle					

Table 4.4b: EFFECTIVE LENGTH FACTORS USING CAN/CSA-S16.1-94
(8-panel cross-braced specimens)

Test No.	Specimen I.D.	Diagonal Diameter (mm)	c/c Length of Diagonal Ld (mm)	Measured Yield Stress Fy (MPa)	Buckling Load (kN)	Force in the Tension Diagonal (kN)	Compression/Tension Ratio	λ	Effective Length Factor K
22	P1A	15.9	1158	363	Did not buckle				
31	P1B	15.9	1158	375	26.3	27.6	0.95	1.510	0.375
21	P2A	15.9	1158	373	29.0	25.6	1.13	1.413	0.361
27	P2B	15.9	1158	373	29.7	6.4	4.64	1.363	0.345
30	P3A	19.1	1158	407	23.5	7.2	3.26	1.619	0.403
32	P3B	19.1	1158	409	30.2	6.1	4.95	1.368	0.341
29	P4A	19.1	1158	410	55.0	44.6	1.23	1.225	0.351
20	P4B	19.1	1158	409	52.0	49.0	1.06	1.260	0.366
26	P5A	22.2	1158	437	60.0	53.2	1.13	1.144	0.327
34	P5B	22.2	1158	437	47.3	53.6	0.88	1.375	0.393
28	P6A	22.2	1158	442	48.7	7.0	6.98	1.348	0.385
15	P6B	22.2	1158	442	51.5	11.9	4.33	1.293	0.369
18	P7A	22.2	1158	351	50.2				0.376
17	P7B	22.2	1158	343	Did not buckle				
33	P8A	22.2	1158	346	88.2	89.6	0.98	1.144	0.367
19	P8B	22.2	1158	349	91.3	21.4	4.27	1.110	0.356
					82.5	6.7	9.48	1.209	0.388
					101.5	12.7	7.99	1.006	0.323
					Did not buckle				
					Did not buckle				
					75.4	12.5	6.03	1.058	0.383
					82.0	19.6	4.18	0.993	0.355

Table 4.4c: EFFECTIVE LENGTH FACTORS USING CAN/CSA-S16.1-94

(6-panel single-braced specimens)

Test No.	Specimen I.D.	Diagonal Diameter (mm)	c/c Length of Diagonal L _d (mm)	Measured Yield Stress F _y (MPa)	Buckling Load (kN)	λ	Effective Length Factor K
14	31SRB	22.2	937	328			
16	33SRC	28.6	979	364	148.6	0.93	0.502
					150.5	0.92	0.495

Table 4.4d: EFFECTIVE LENGTH FACTORS USING AISC-LRFD SPECIFICATIONS

(6-panel cross-braced specimens)

Test No.	Specimen I.D.	Diagonal Diameter (mm)	c/c Length of Diagonal L _d (mm)	Measured Yield Stress F _y (MPa)	Buckling Load (kN)	Force in the Teelson Diagonal (kN)	Compression/Tension Ratio	λ	Effective Length Factor K
6	S1A	12.7	1190	357	14.1	2.6	5.42	1.677	0.325
9	S1B	12.7	1190	396	12.4	3.3	3.76	1.788	0.352
					Did not buckle				
4	S2A	12.7	1190	378	14.8	15.0	0.99	1.684	0.318
8	S2B	12.7	1190	369	14.3	16.0	0.89	1.713	0.325
23	S2C	12.7	1190	353	13.7	13.6	1.01	1.731	0.333
7	S3A	14.3	1190	395	13.7	13.4	1.02	1.731	0.333
25	S3B	14.3	1190	396	14.5	16.6	0.87	1.643	0.319
1	S4A	14.3	1190	378	14.8	14.0	1.06	1.627	0.315
3	S4B	14.3	1190	378	21.3	4.6	4.63	1.595	0.331
5	S4C	14.3	1190	367	19.4			1.671	0.351
2	S5A	15.9	1190	385	20.4	8.1	2.52	1.658	0.342
10	S5B	15.9	1190	385	20.1	4.6	4.37	1.671	0.345
12	S5C	15.9	1190	385	20.1	16.9	1.19	1.627	0.343
11	S6A	15.9	1190	385	24.9			1.459	0.296
13	S6B	15.9	1190	382	21.4	19.3	1.11	1.554	0.327
24	S6C	15.9	1190	385	22.6	22.0	1.04	1.506	0.313
14	31SRB	22.2	937	328	21.6	1.9	11.37	1.759	0.416
					20.9	8.1	2.56	1.788	0.424
					33.2	10.4	3.19	1.409	0.312
					29.0	6.9	4.20	1.518	0.343
					30.2	9.8	3.08	1.487	0.334
					27.3	23.9	1.14	1.564	0.358
					29.2	13.8	2.12	1.512	0.342
					25.4	17.3	1.47	1.615	0.375
					Did not buckle				
					Did not buckle				

Table 4.4e: EFFECTIVE LENGTH FACTORS USING AISC-LRFD SPECIFICATIONS
(β -panel cross-braced specimens)

Test No.	Specimen I.D.	Diagonal Diameter (mm)	c/c Length of Diagonal Ld (mm)	Measured Yield Stress Fy (MPa)	Buckling Load (kN)	Force in the Tension Diagonal (kN)	Compression/Tension Ratio	λ	Effective Length Factor K
22	P1A	15.9	1158	363					
31	P1B	15.9	1158	375	26.3 28.0	27.6 25.6	0.95 1.13	1.574 1.499	0.375 0.381
21	P2A	15.9	1158	373					
27	P2B	15.9	1158	373	29.7 23.5 30.2	6.4 7.2 6.1	4.64 3.26 4.95	1.474 1.660 1.464	0.345 0.403 0.341
30	P3A	19.1	1158	407	55.0 52.0	44.6 49.0	1.23 1.06	1.361 1.400	0.351 0.366
32	P3B	19.1	1158	409	60.0 47.3	53.2 53.6	1.13 0.88	1.305 1.470	0.327 0.383
29	P4A	19.1	1158	410					
20	P4B	19.1	1158	409	48.7 51.5 50.2	7.0 11.9	6.96 4.33	1.445 1.397 1.419	0.385 0.369 0.376
26	P5A	22.2	1158	437					
34	P5B	22.2	1158	437					
28	P6A	22.2	1158	442	68.2 91.3 82.5	69.6 21.4 6.7	0.98 4.27 9.46	1.259 1.226 1.321	0.367 0.356 0.368
15	P6B	22.2	1158	442					
16	P7A	22.2	1158	351	101.5	12.7	7.99	1.118	0.323
17	P7B	22.2	1158	343					
33	P8A	22.2	1158	346	75.4	12.5	6.03	1.172	0.363
19	P8B	22.2	1158	349	82.0	19.6	4.16	1.093	0.355

Table 4.4f: EFFECTIVE LENGTH FACTORS USING AISC-LRFD SPECIFICATIONS

(6-panel single-braced specimens)

Test No.	Specimen I.D.	Diagonal Diameter (mm)	c/c Length of Diagonal L _d (mm)	Measured Yield Stress F _y (MPa)	Buckling Load (kN)	λ	Effective Length Factor K
14	31SRB	22.2	937	328	Did not buckle		
16	33SRC	28.6	979	364	148.6	1.039	0.502
					150.5	1.025	0.495

Table 4.4g: AVERAGE EFFECTIVE LENGTH FACTORS

Effective Length Factor	Single-Braced Diagonal	Cross-Braced Diagonal		Average value
		6-Panel	8-Panel	
Effective Length Factor Using CAN/CSA -S16.1-94	0.50	0.37	0.34	0.35
Effective Length Factor Using AISC-LRFD	0.56	0.39	0.35	0.37

CHAPTER 5

CONCLUSIONS AND RECOMMENDATIONS

5.1 Conclusions

The following conclusions are based on this experimental investigation:

The distribution of forces in compression and tension diagonals and the force distribution in top half and bottom half of the compression or tension diagonals are not the same. No definite conclusions about the force distribution can be drawn.

- For identical specimens of each type, the compression diagonals for the same panel would not have the same buckling load due to minor fabrication differences.
- Diagonals in the end panels of specimens with horizontals at mid-span buckle suddenly at a high load, while the specimens without horizontals at mid-span would buckle at the middle panels at a much smaller load.
- Specimens with small leg-diagonal size ratio buckle due to the buckling of top leg, not due to the buckling of compression diagonals in any panel.
- For cross-braced specimens without horizontals, the tested compression diagonals buckle out-of-plane inside or outside with one almost half-sine wave due to the small axial force in the tension diagonals. On the other hand, for cross-braced specimens with horizontals, the tested compression diagonals buckle out-of-plane inside or outside with a symmetrical S-shape or an unsymmetrical S-shape due to the large force in the tension diagonals. For single-braced specimens, the compression diagonals buckle out-of-plane with one half-sine wave.
- The axial forces in the tension diagonals don't have a big effect on effective length factors for cross-braced compression diagonals.

- **The average effective length factor for single-braced diagonals is 0.50. The effective length factor for cross-braced diagonals is 0.35, as compared to an effective length factor of 0.75 as given by CSA-S37-94 that is much too conservative.**

5.2 Recommendation for Further Research

- **To know the distribution of forces in the members, tests should be carried out on towers with loads applied at each panel point (instead of one load at mid-span of top chord as in the current investigation) to simulate the actual distributed wind load.**
- **Tests should be carried to study the effect of fabrication differences on the effective length factor.**

REFERENCES

REFERENCES

ASTM, 1989, "ASTM A370-88a: Standard Test Methods and Definitions for Mechanical Testing of Steel Products," American Society for Testing and Materials, Philadelphia, PA.

Beedle, L. S., Galambos, T. V., and Tall, L., 1961, "New Concepts in Steel Design and Engineering: Column Strength of Construction Steels," USS Steel Corp., Pittsburgh, PA.

CSA, 1994, "Antennas, Towers, and Antenna Supporting Structures," CSA-S37-94, Canadian Standards Association, Etobicoke, Ontario.

Dewolf, J.T. and Pelliccione, J. F., 1979, "Cross-Bracing Design," Journal of the Structural Division, ASCE, ST7, July, pp.1379-1391.

El-Tayem, A. A. and Goel. S. C., 1986, "Effective Length Factor for the Design of X-bracing Systems," AISC Engineering Journal, First Quarter, Chicago, Illinois, pp. 41-45.

Euler, L., 1759 "Sur La Force des Colonnes," Memoires de L'Academie Berlin, Vol. 13.

Fujita, Y. and Driscoll, F. C., Jr., 1962, "Strength of Round Columns," Journal of the Structural Division, ASCE, Vol. 88, ST2, Proc. Paper 3090, April, pp. 43-59.

Galambos, T. V., 1965, "Strength of Round Steel Columns" Journal of the Structural Division, Proceedings of the American Society of Civil Engineers, ST1, Feb., pp. 121-139.

Jaboo K. S., 1998, "Effective Length Factors for Solid Round Diagonal Bracing Members in Lattice Towers", M. A. Sc. Thesis, Civil and Environmental Engineering Program, University of Windsor, Windsor, Ontario.

Kemp, A. R. and Behncke, H., 1998, "Behavior of Cross-bracing in Latticed Towers," Journal of Structural Engineering, ASCE, April, pp. 360-367.

Picard, A. and Beaulieu, D., 1989, "Theoretical Study of the Buckling Strength of Compression Members Connected To Coplanar Tension Members," Canadian Journal of Civil Engineering, Vol. 16, pp. 239-248.

Picard, A. and Beaulieu, D., 1989, "Experimental Study of the Buckling Strength of Compression Members Connected to Coplanar Tension Members, Canadian Journal of Civil Engineering, Vol. 16, pp. 249-257.

Picard, A. and Beaulieu, D., 1987, "Design of Diagonal Cross-Bracings: Part 1: Theoretical Study," Engineering Journal, American Institute of Steel Construction, Third Quarter, pp. 122-126.

Picard, A. and Beaulieu, D., 1988, "Design of Diagonal Cross-Bracings: Part 2: Experimental Study," Engineering Journal, American Institute of Steel Construction, Fourth Quarter, pp. 156-160.

Shanley, F. R., 1947, "Inelastic Column Theory," Journal of Aeronautical Sciences, Vol. 14, No. 5, May, pp. 261-267.

Timoshenko, S. P., 1953, "History of Strength of Materials," McGraw-Hill Book Company, Inc., New York, NY.

Timoshenko, S. P. and Gere, J. M., 1961, "Theory of Elastic Stability," Second Ed., McGraw-Hill Book Company, Inc., New York, NY.

Todhunter, I. And Pearson, K., 1960, "A History of the Theory of Elasticity and of Strength of Materials," Dover Publications, (originally published by Cambridge University press 1893).

Vickers, D. G., 1982, "Tension-Compression Cross-Bracing Using Star Angles," Master of Engineering Project Report, Dept. of Civil Engineering, McGill University, Montreal, Quebec.

Williamson, R.A. and Johnston, B.G., 1965, "Strength of Round Steel Columns," Discussion of Galambos' Paper, Journal of Structural Division , ST 5, Oct., pp. 361-366.

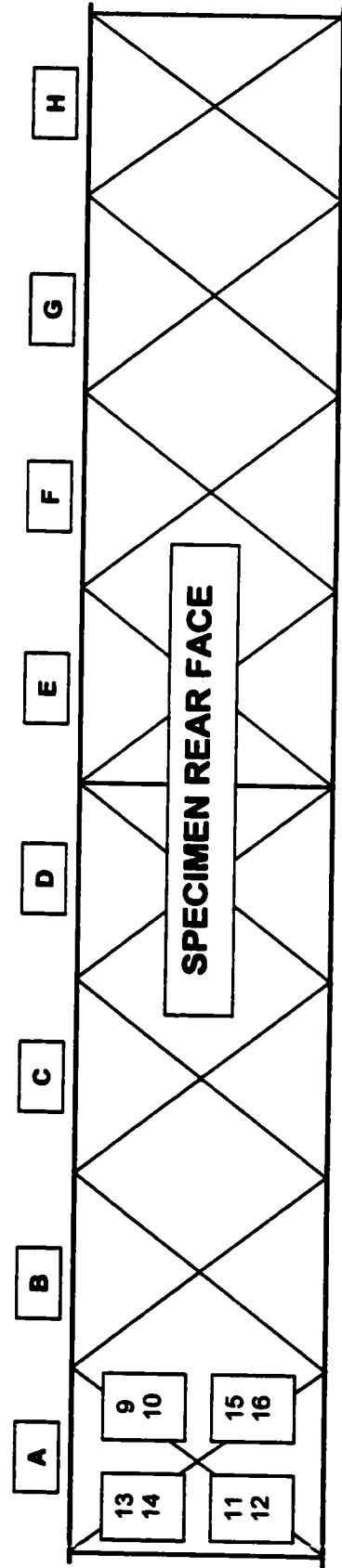
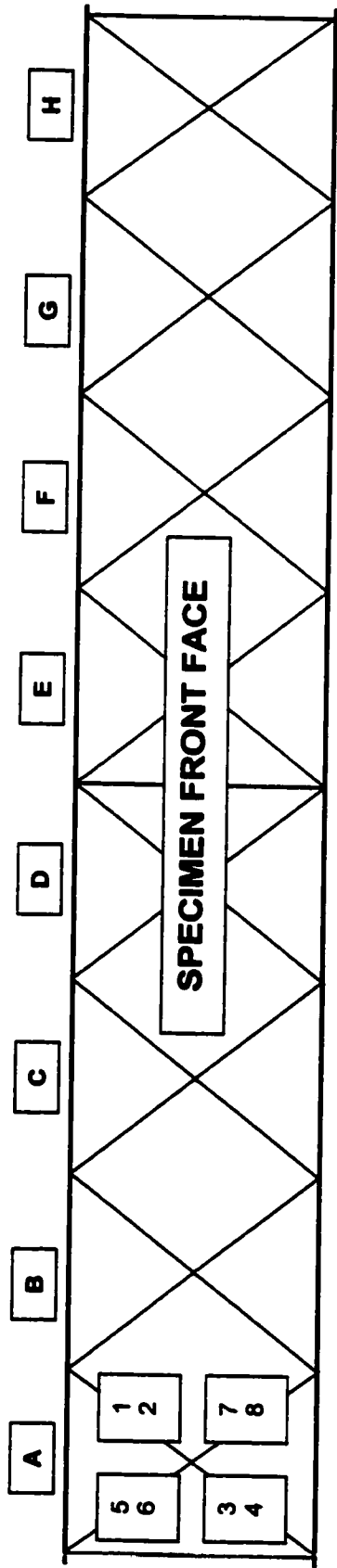
APPENDIX A

STRAIN GAUGE LOCATION

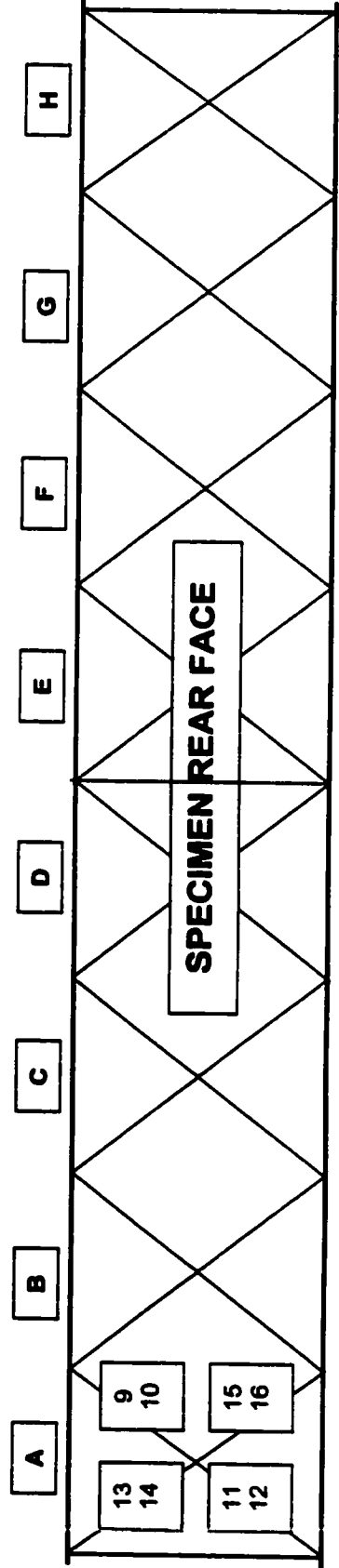
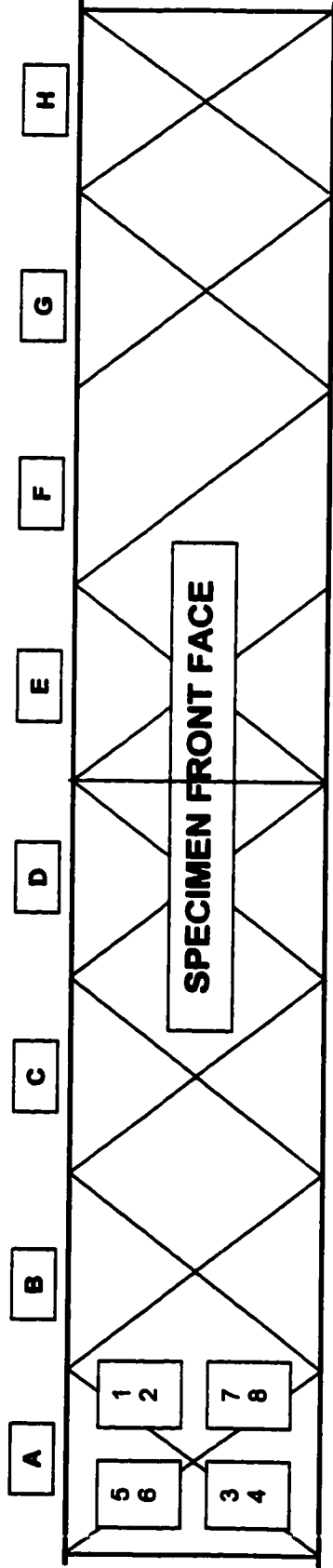
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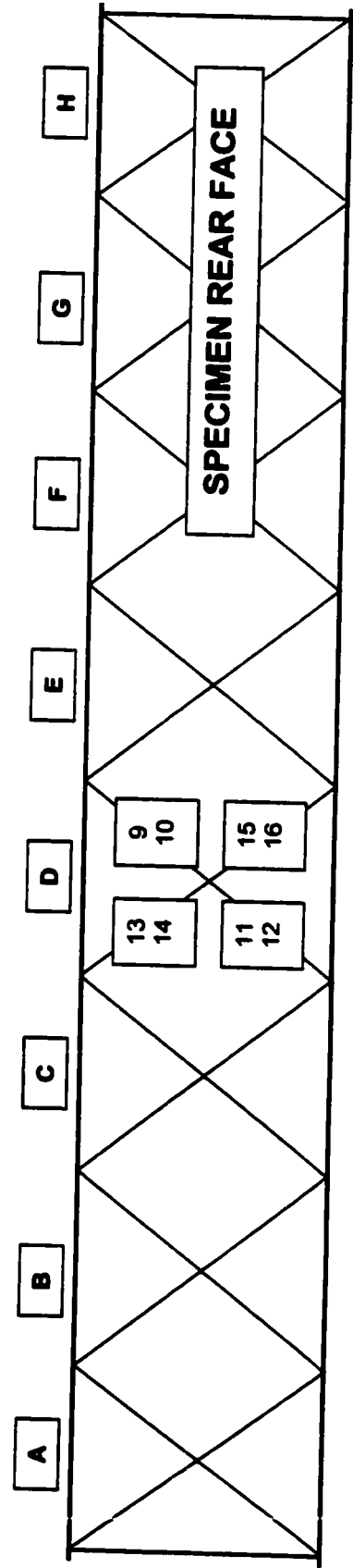
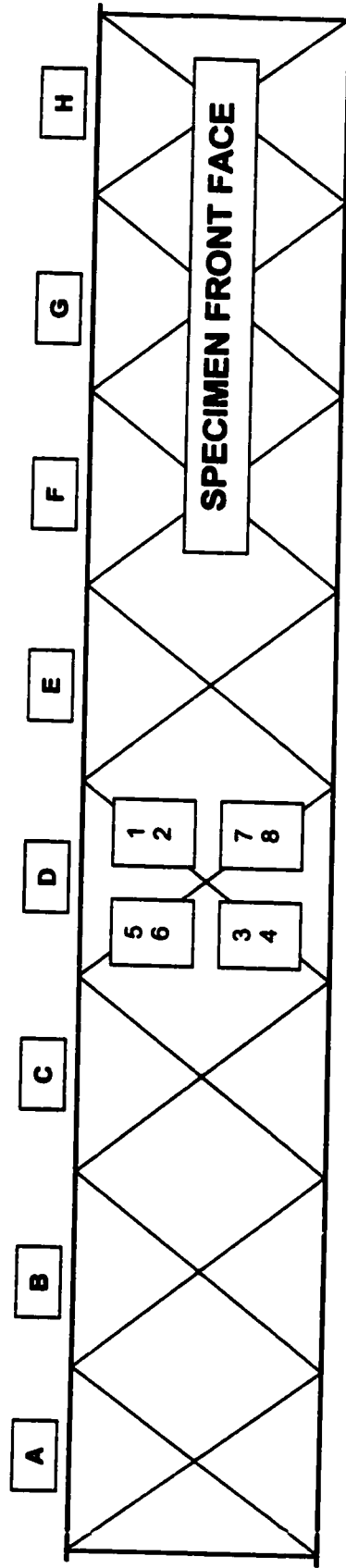
SPECIMEN P1-A (TEST # 22) STRAIN GAUGE IDENTIFICATION



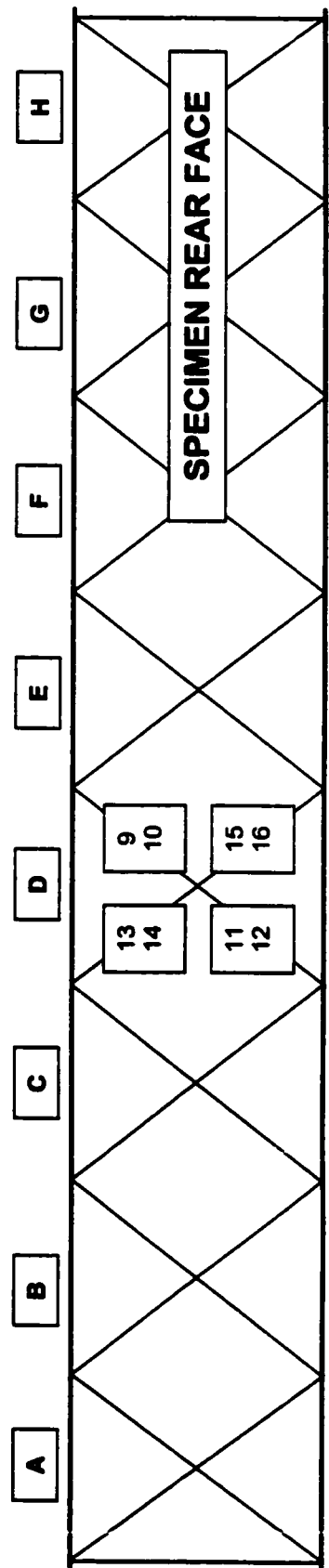
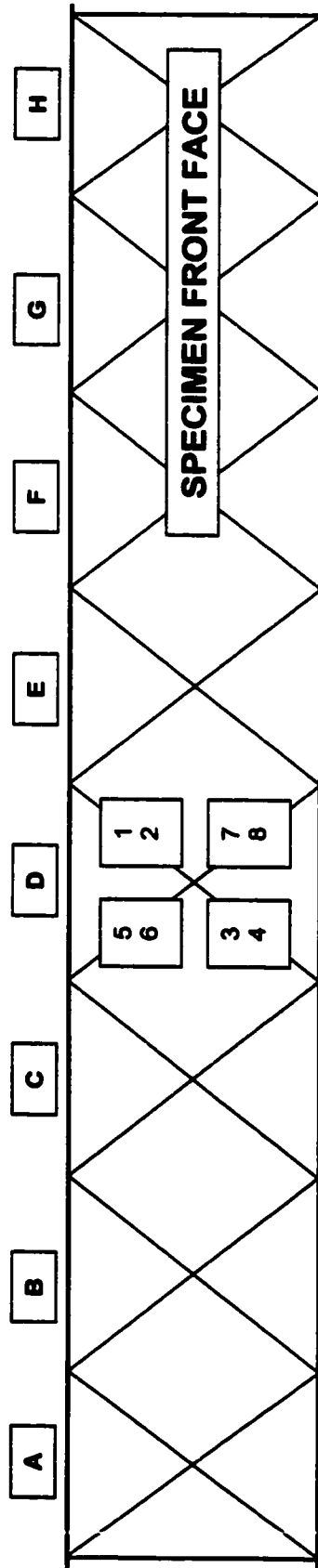
SPECIMEN P1-B (TEST # 31) STRAIN GAUGE IDENTIFICATION



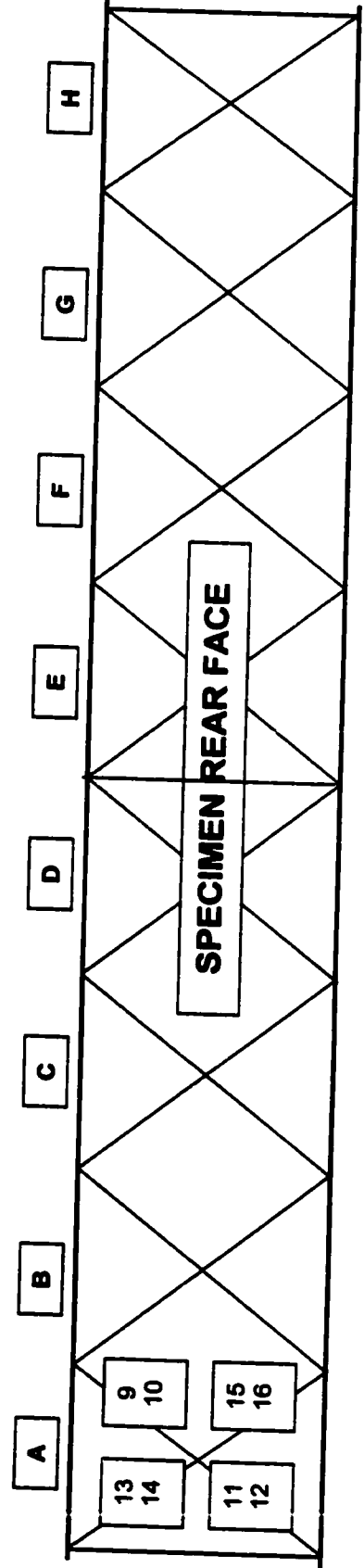
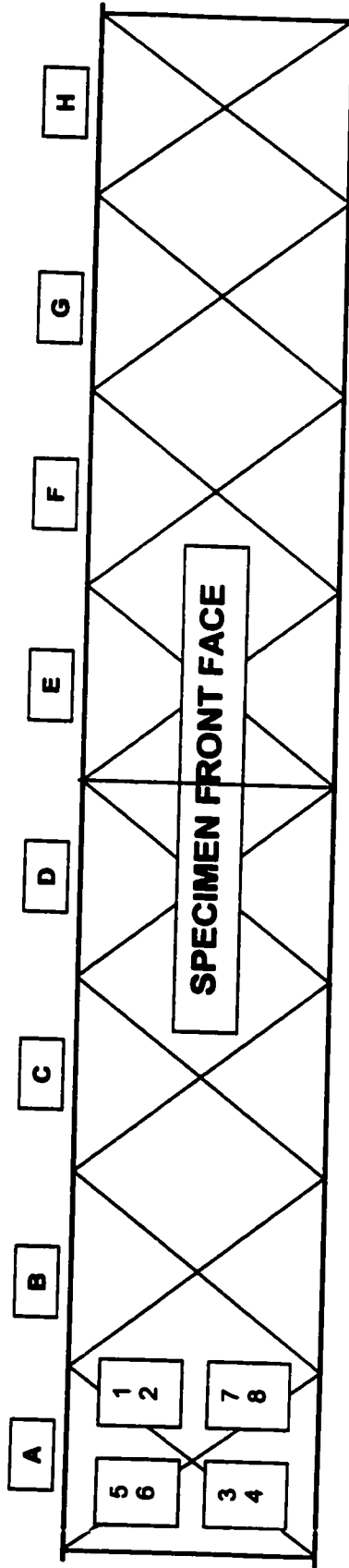
SPECIMEN P2-A (TEST # 21) STRAIN GAUGE IDENTIFICATION



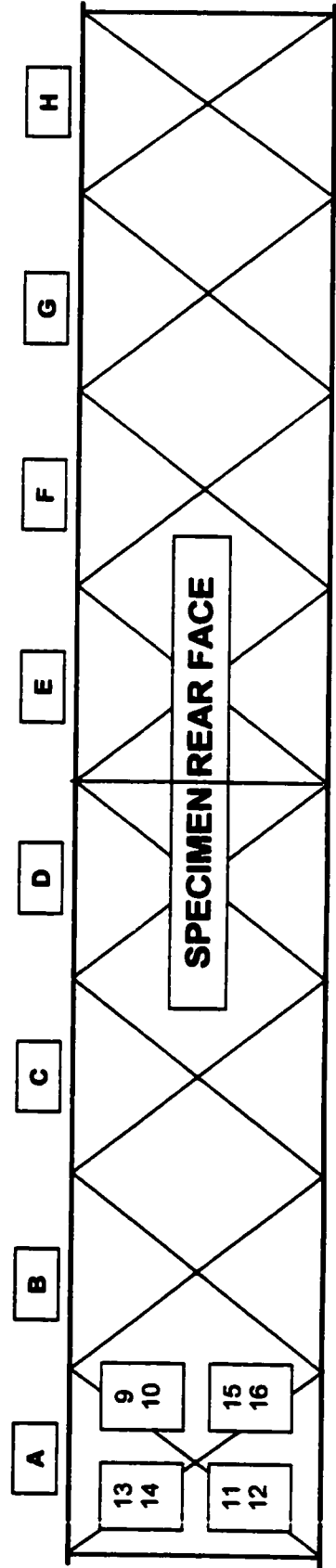
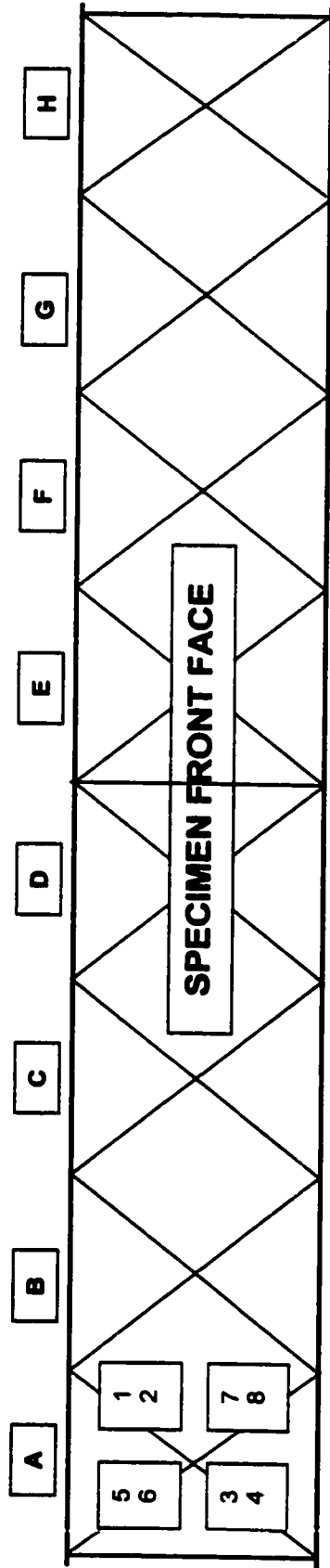
SPECIMEN P2-B (TEST # 27) STRAIN GAUGE IDENTIFICATION



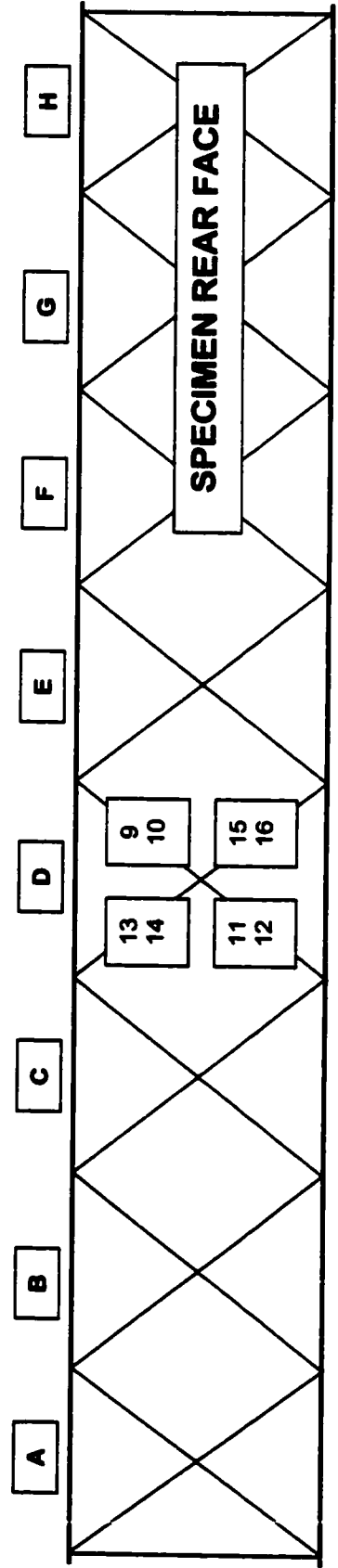
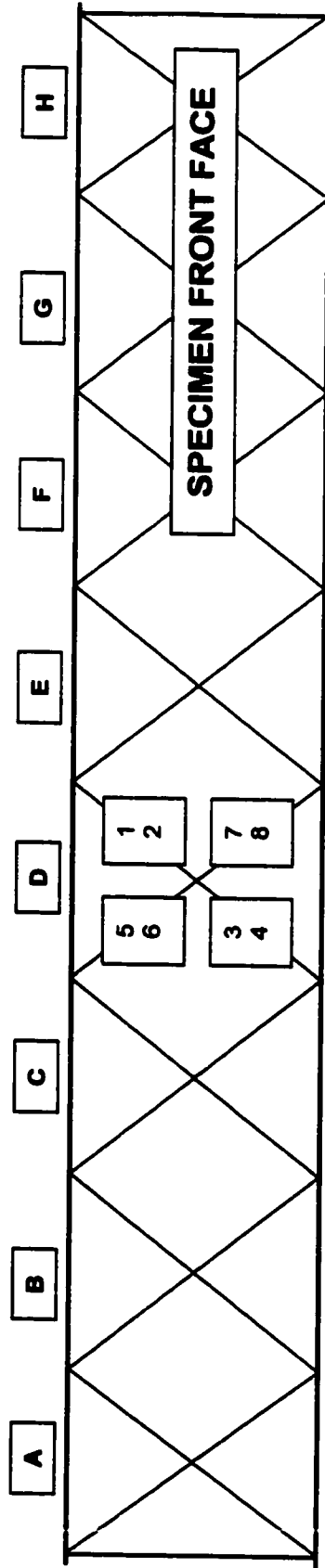
SPECIMEN P3-A (TEST # 30) STRAIN GAUGE IDENTIFICATION



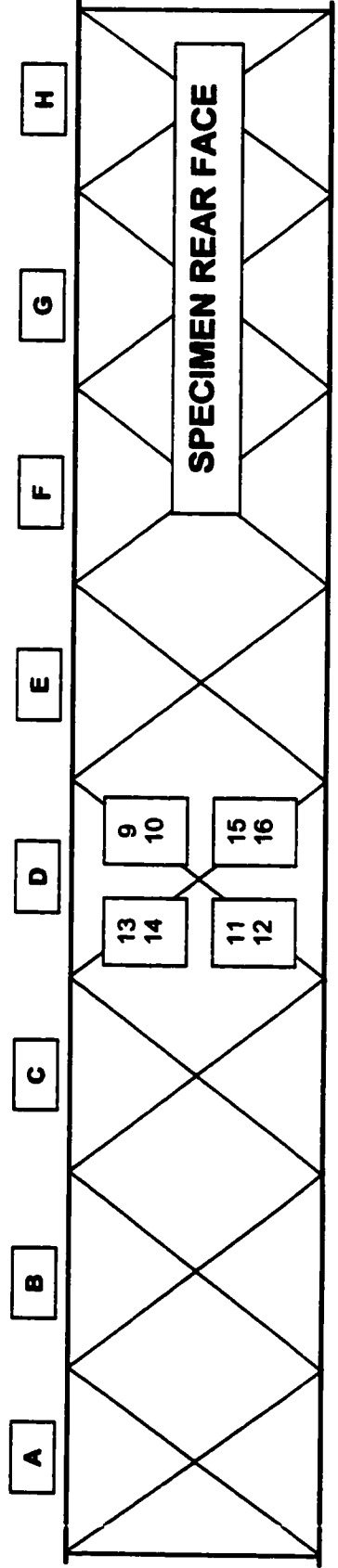
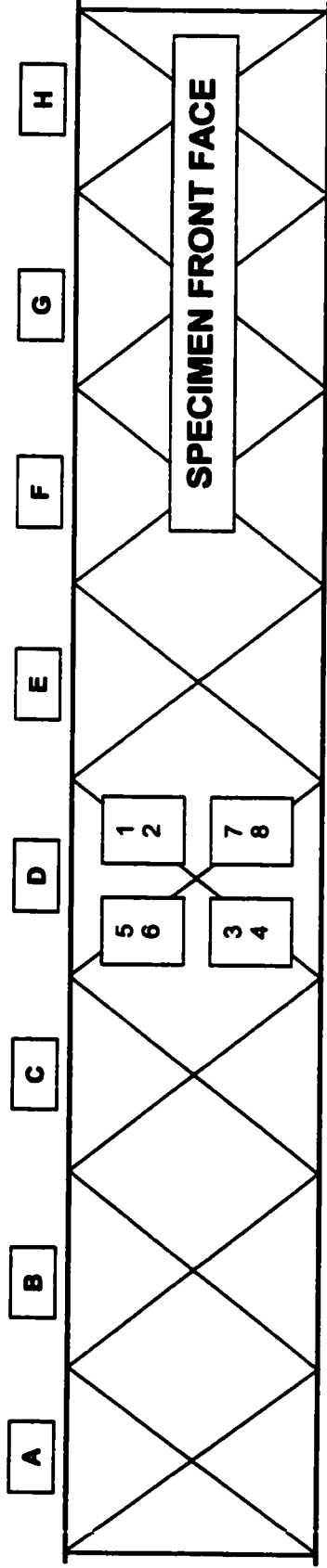
SPECIMEN P3-B (TEST # 32) STRAIN GAUGE IDENTIFICATION



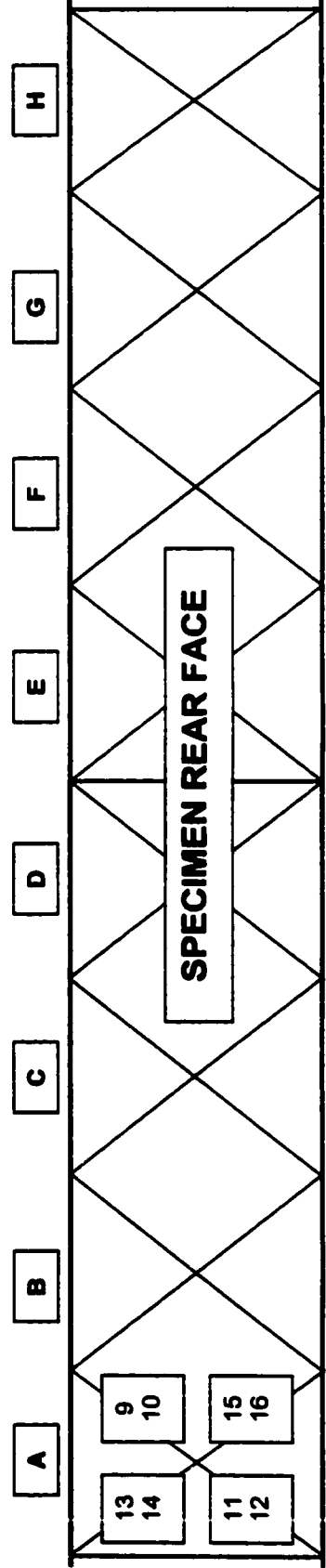
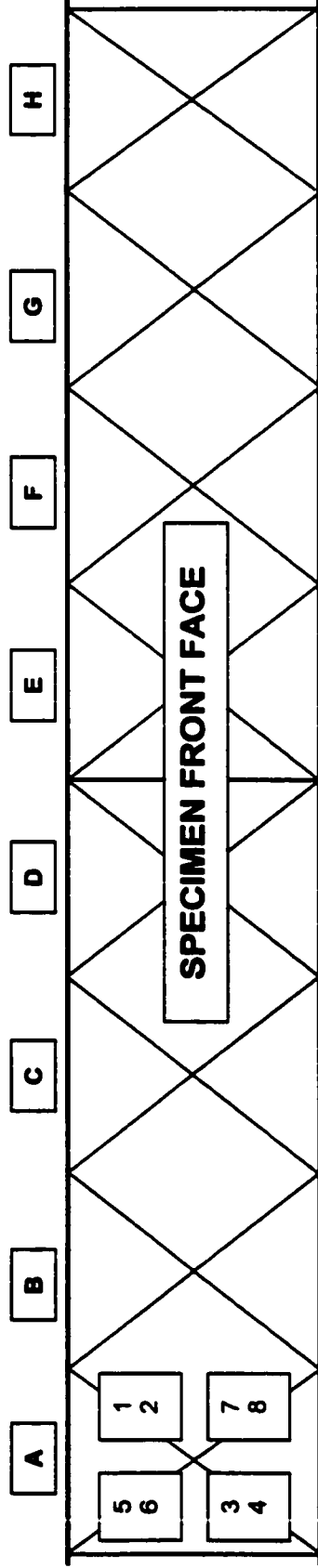
SPECIMEN P4-A (TEST # 29) STRAIN GAUGE IDENTIFICATION



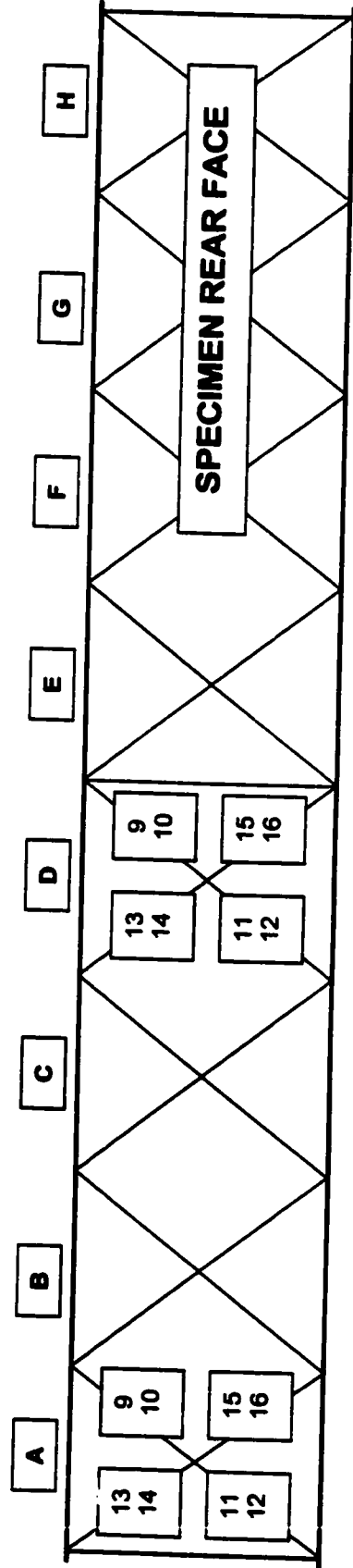
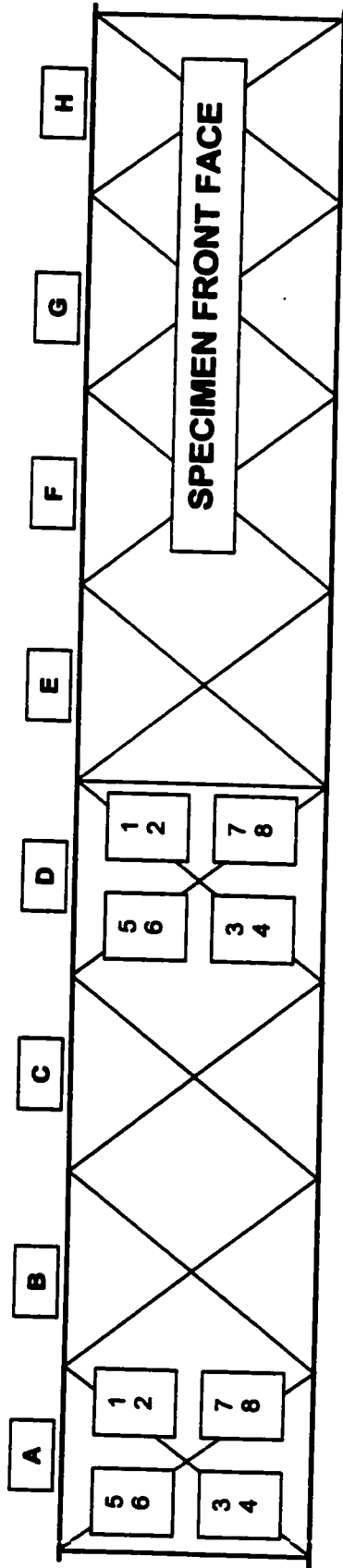
SPECIMEN P4-B (TEST # 20) STRAIN GAUGE IDENTIFICATION



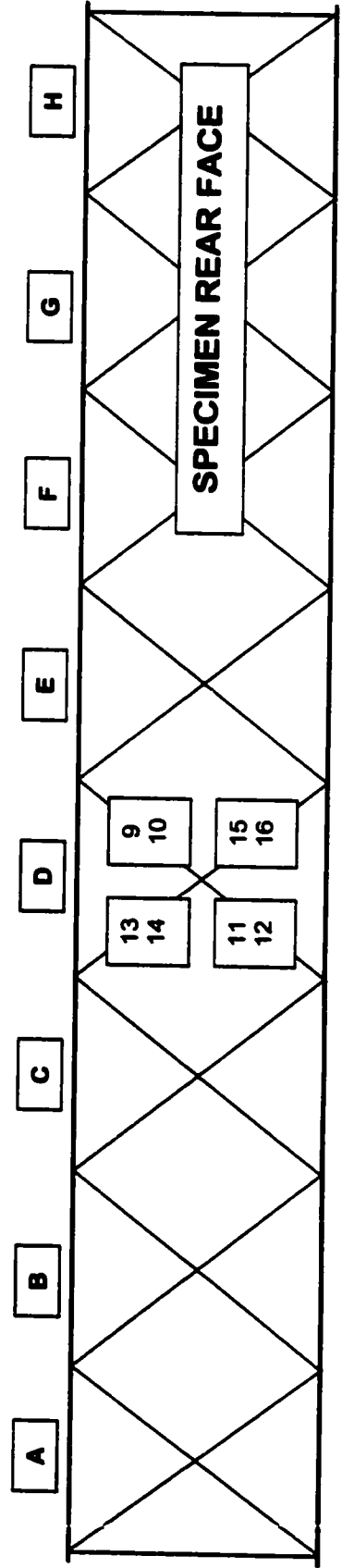
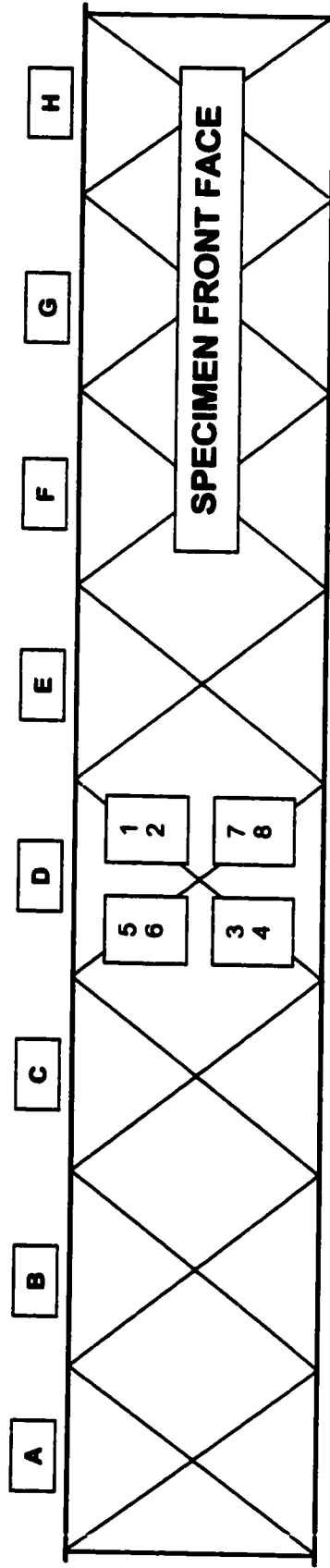
SPECIMEN P5-A (TEST # 26) STRAIN GAUGE IDENTIFICATION



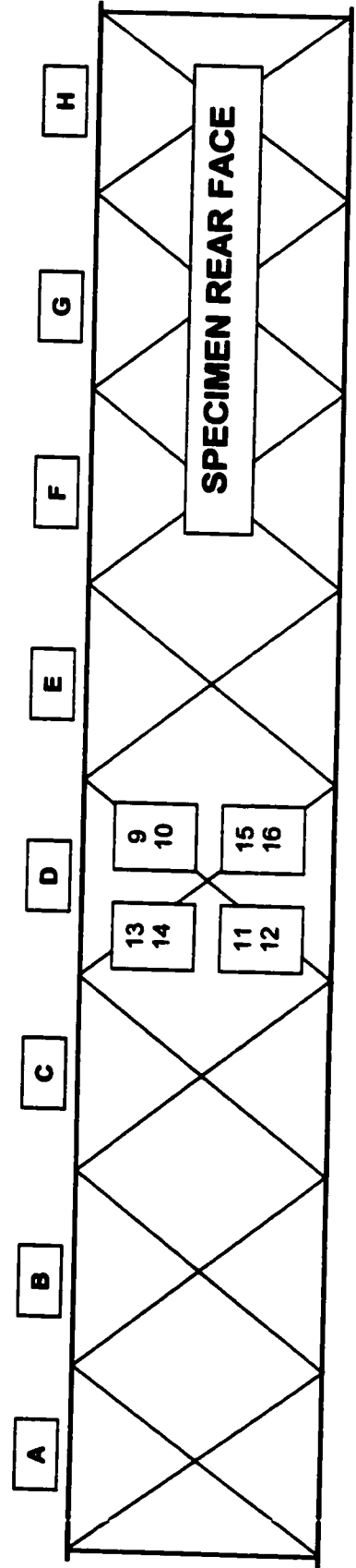
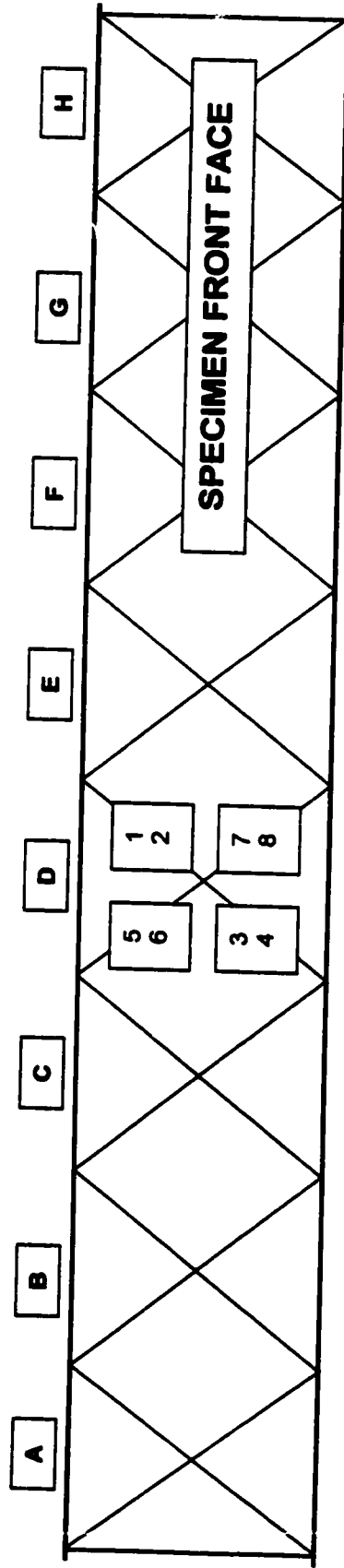
SPECIMEN P5-B (TEST # 34) STRAIN GAUGE IDENTIFICATION



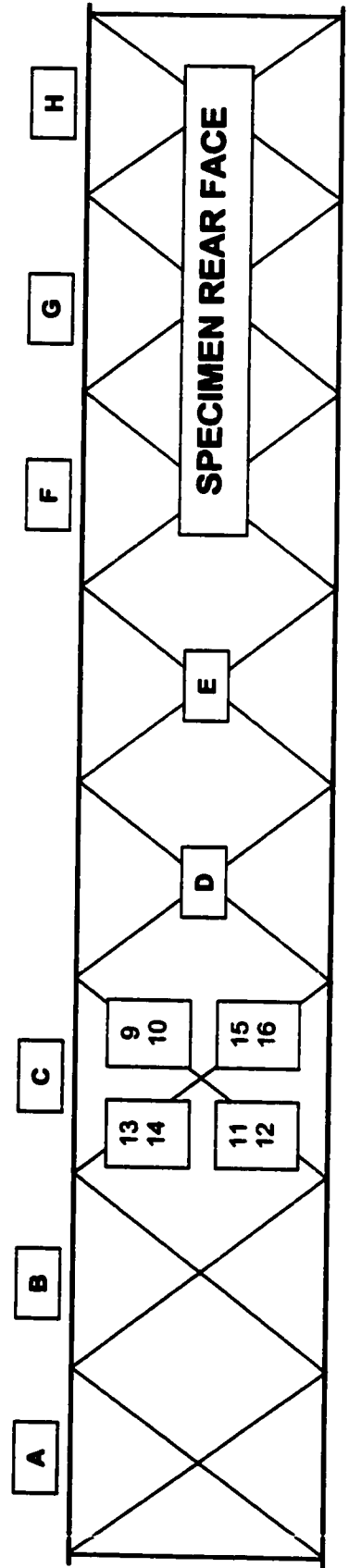
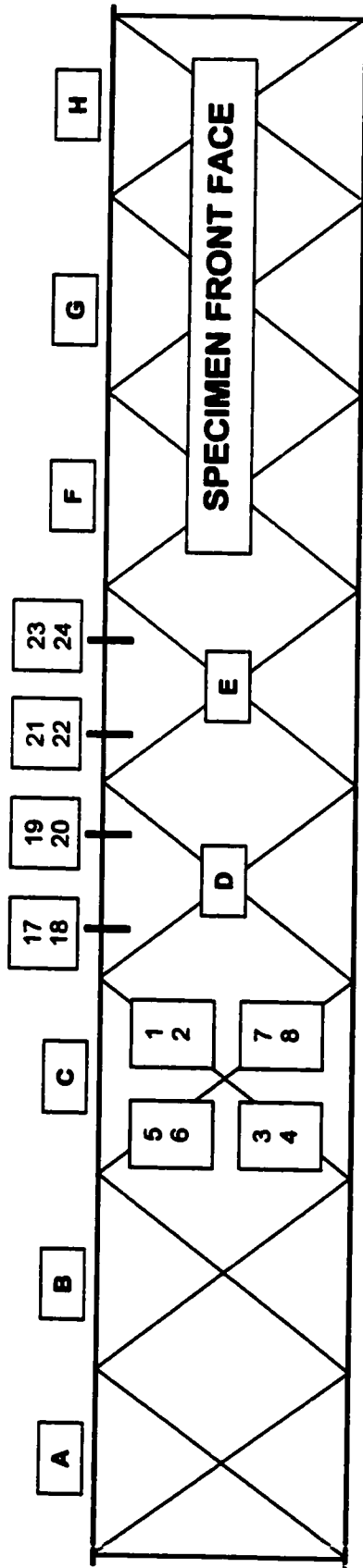
SPECIMEN P6-A (TEST # 28) STRAIN GAUGE IDENTIFICATION



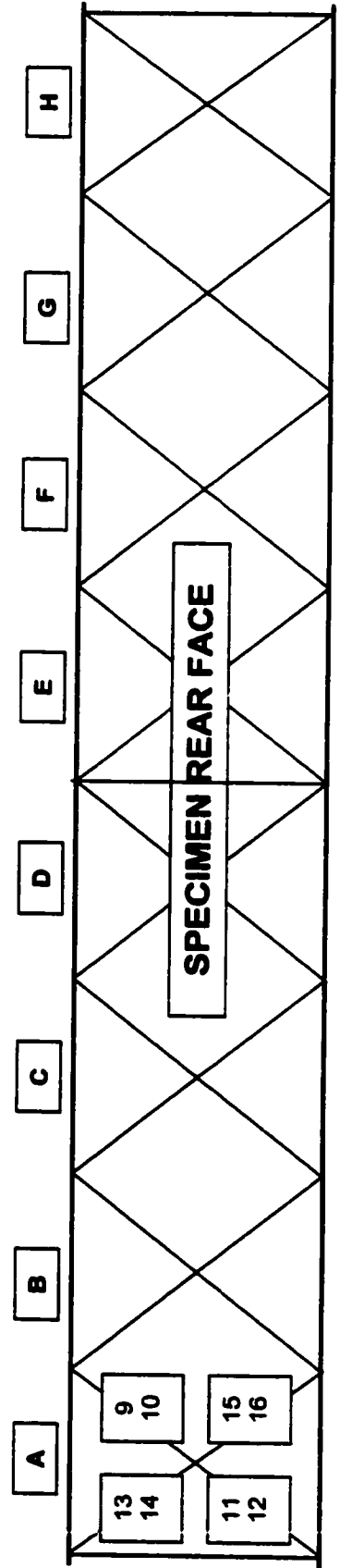
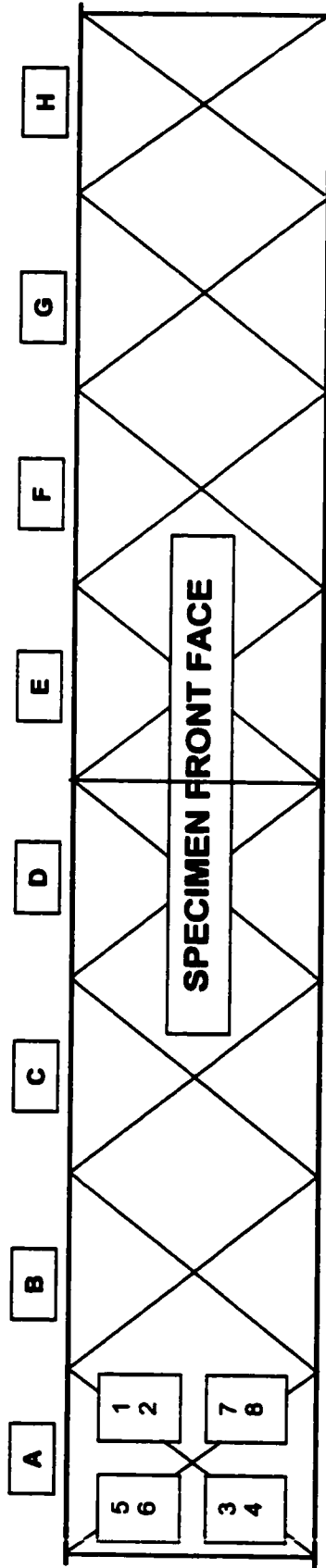
SPECIMEN P6-B (TEST # 15) STRAIN GAUGE IDENTIFICATION



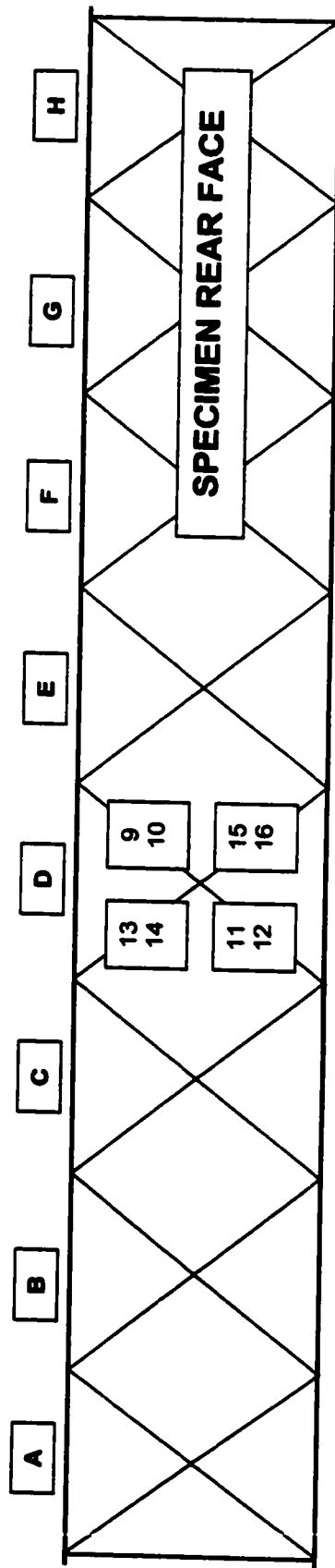
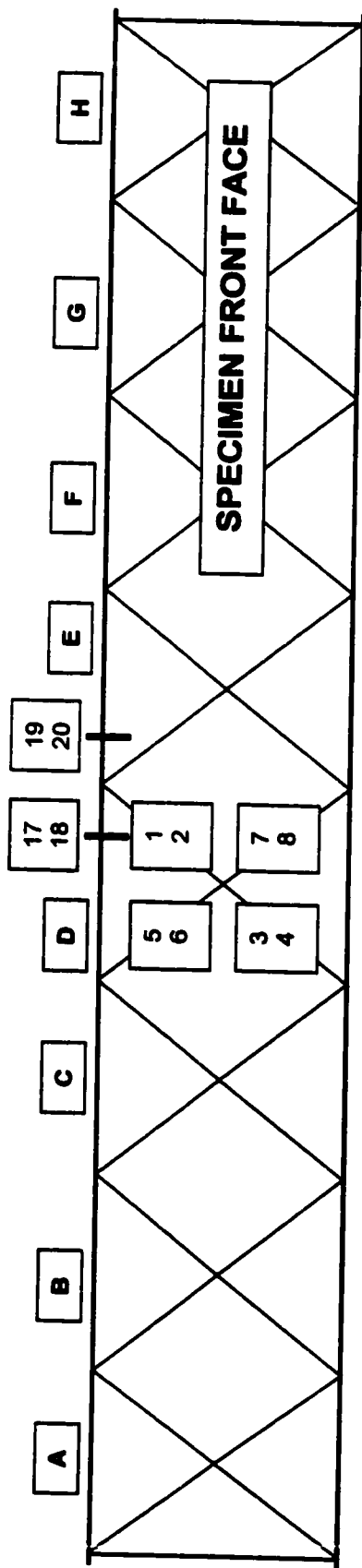
SPECIMEN P7-A (TEST # 18) STRAIN GAUGE IDENTIFICATION



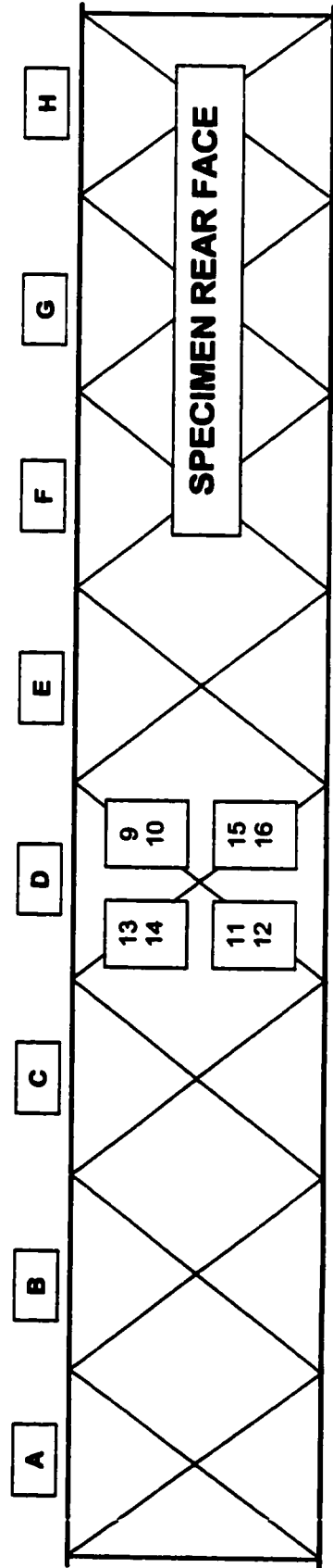
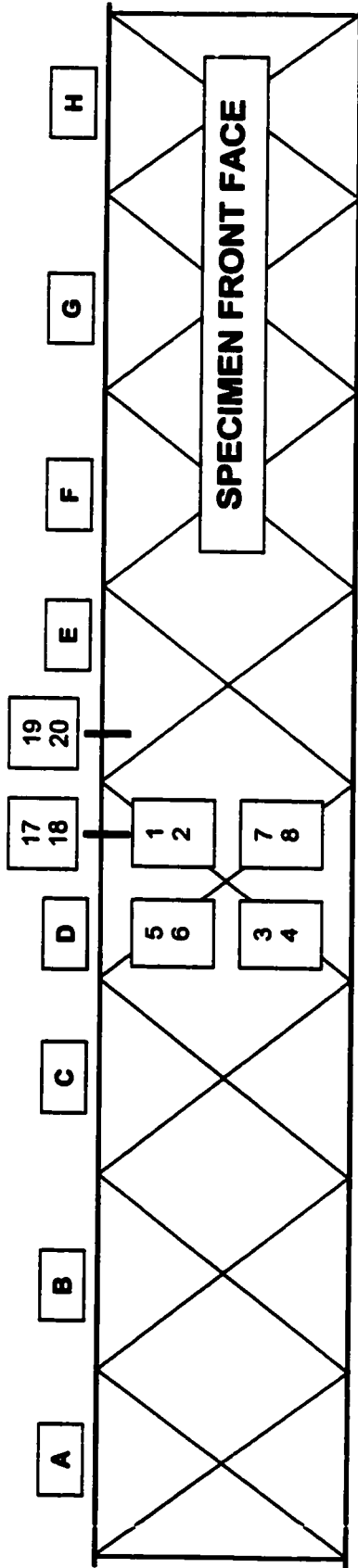
SPECIMEN P7-B (TEST # 17) STRAIN GAUGE IDENTIFICATION



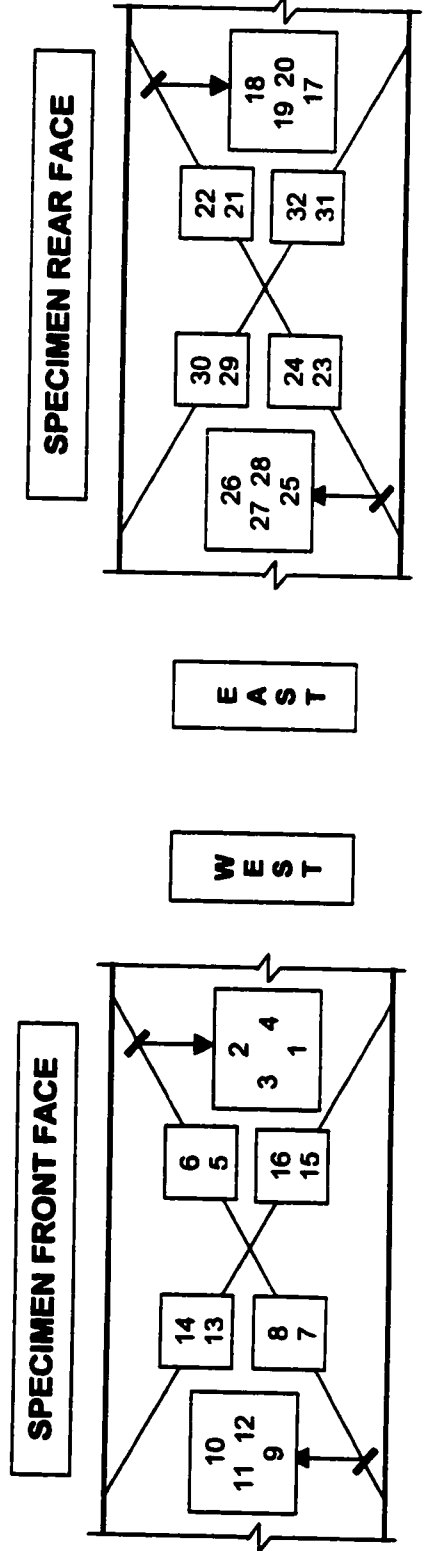
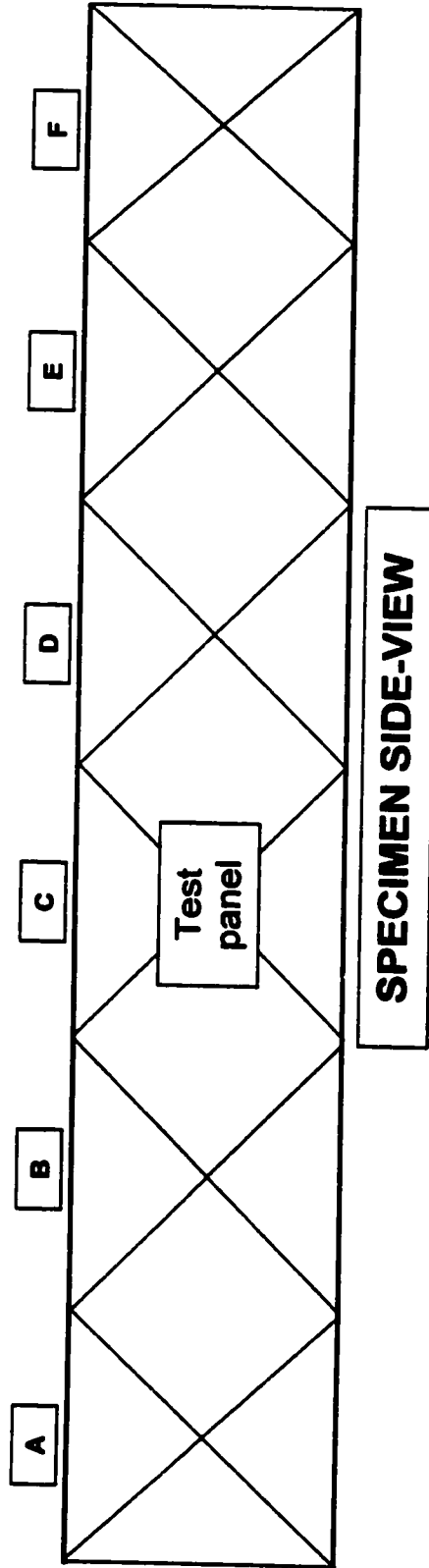
SPECIMEN P8-A (TEST # 33) STRAIN GAUGE IDENTIFICATION



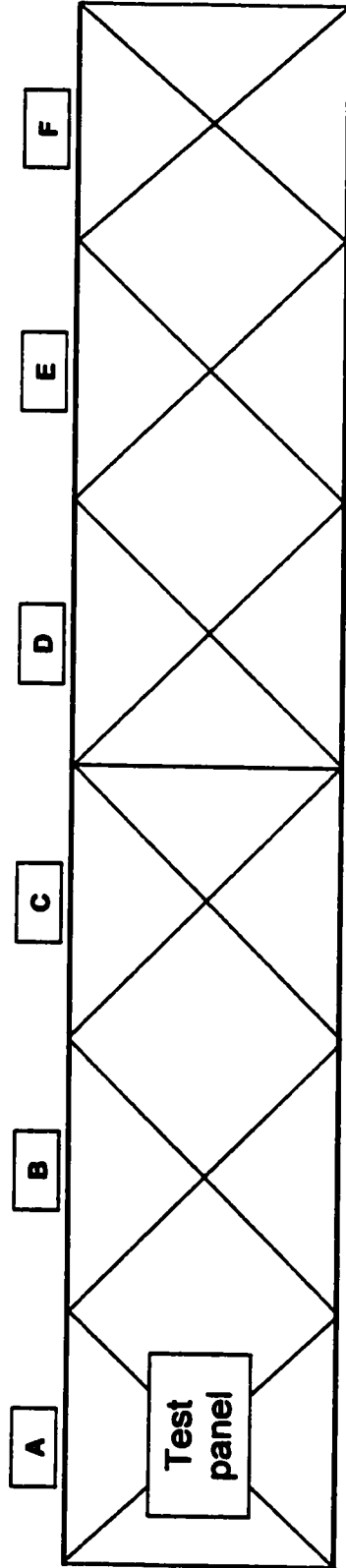
SPECIMEN P8-B (TEST # 19) STRAIN GAUGE IDENTIFICATION



SPECIMEN S1-A (TEST # 6) STRAIN GAUGE IDENTIFICATION

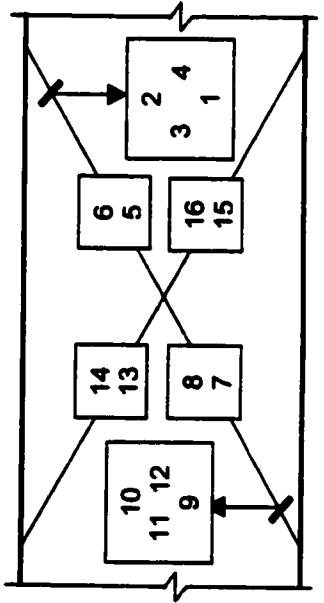


SPECIMEN S2-A (TEST # 4) STRAIN GAUGE IDENTIFICATION

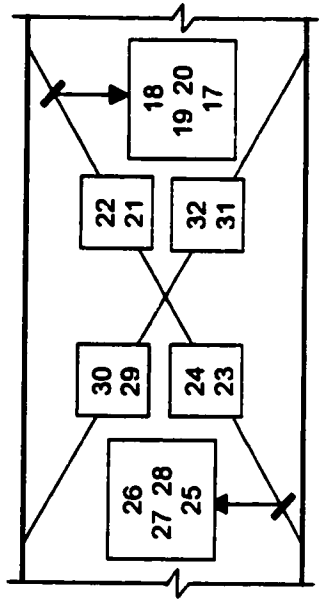


SPECIMEN SIDE-VIEW

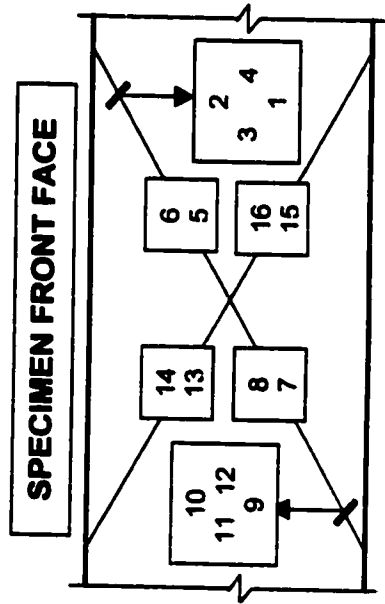
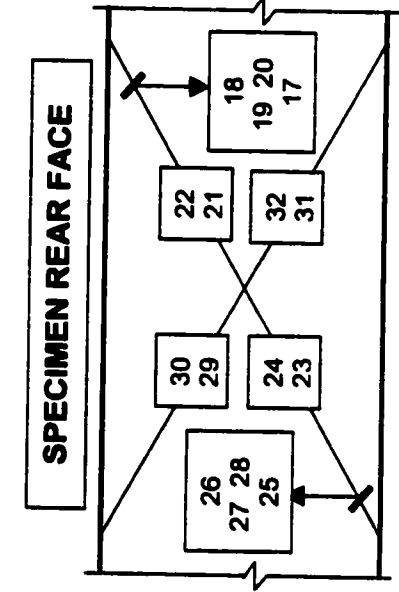
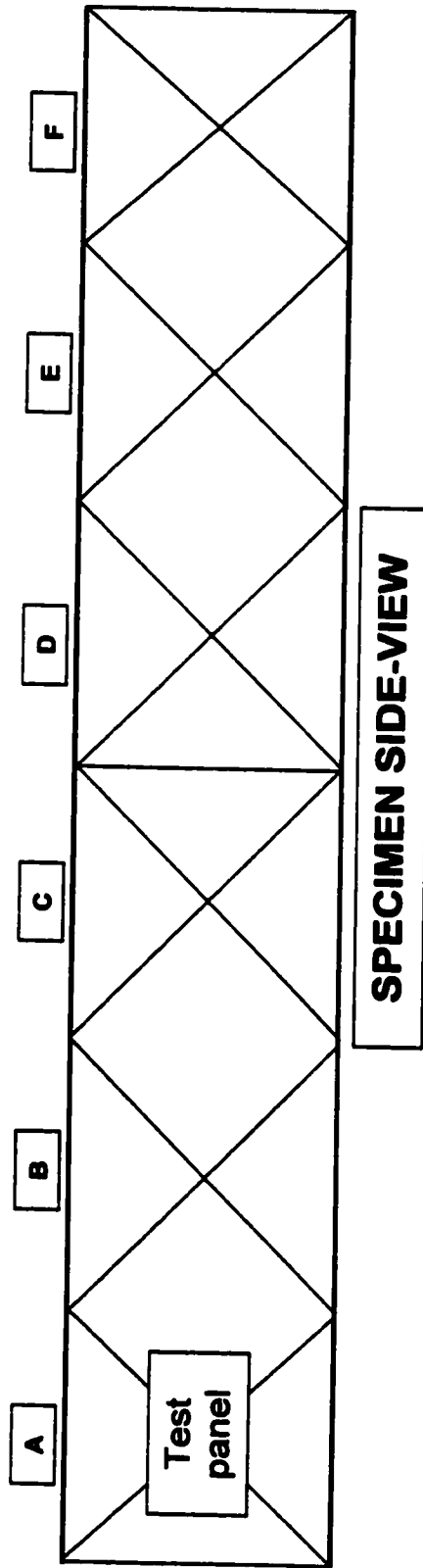
SPECIMEN FRONT FACE



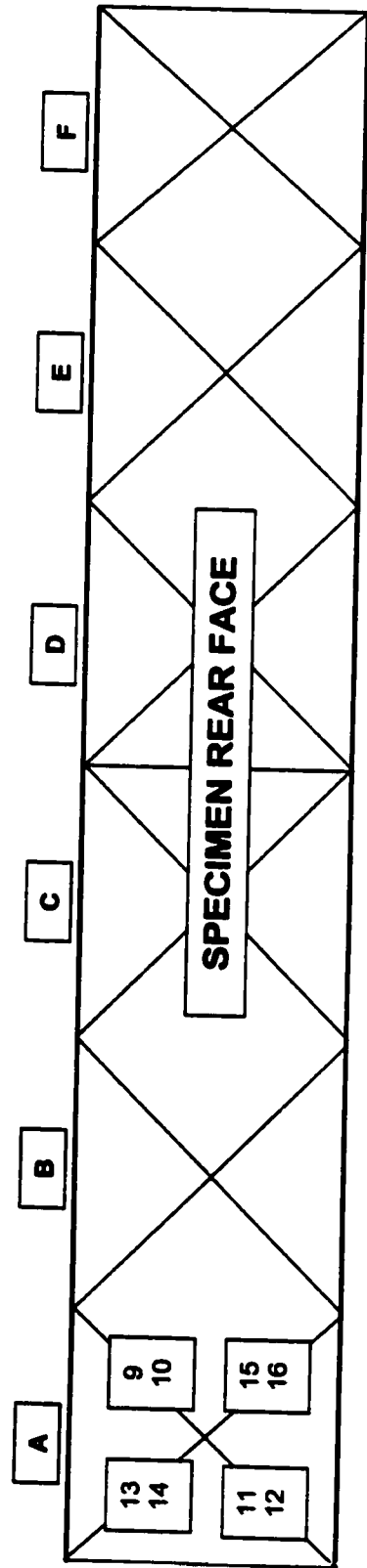
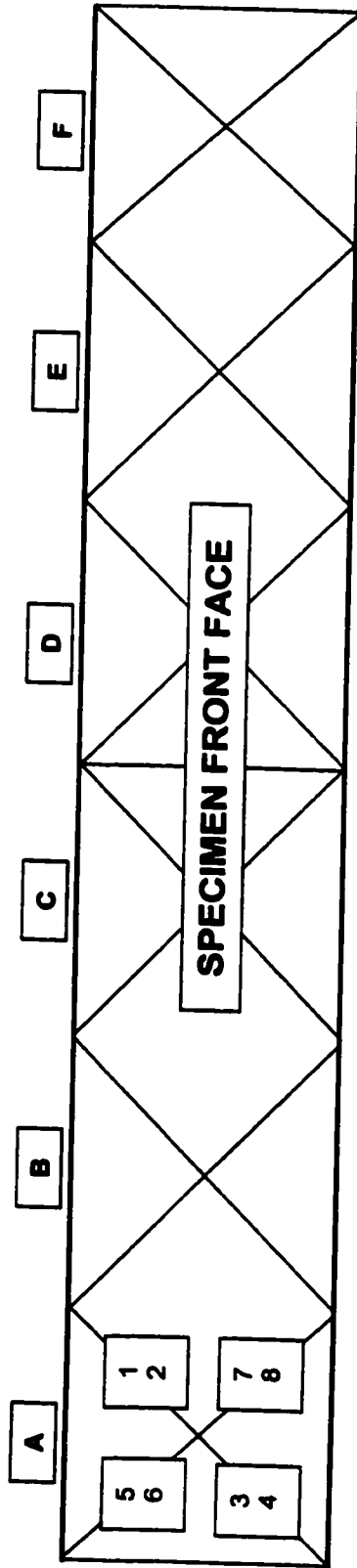
SPECIMEN REAR FACE



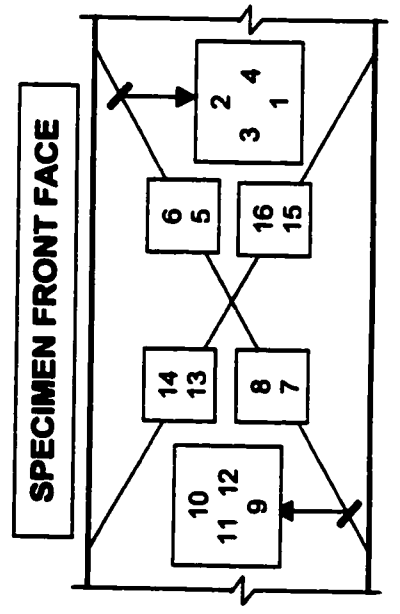
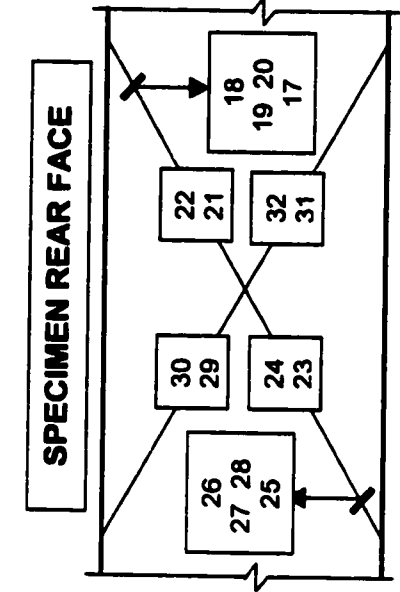
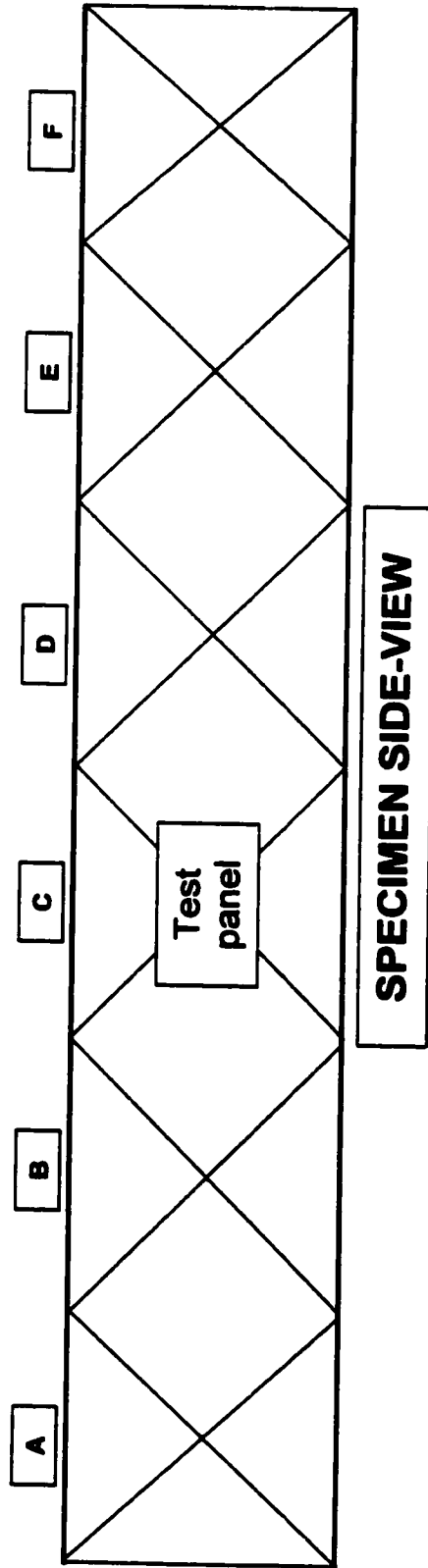
SPECIMEN S2-B (TEST # 8) STRAIN GAUGE IDENTIFICATION



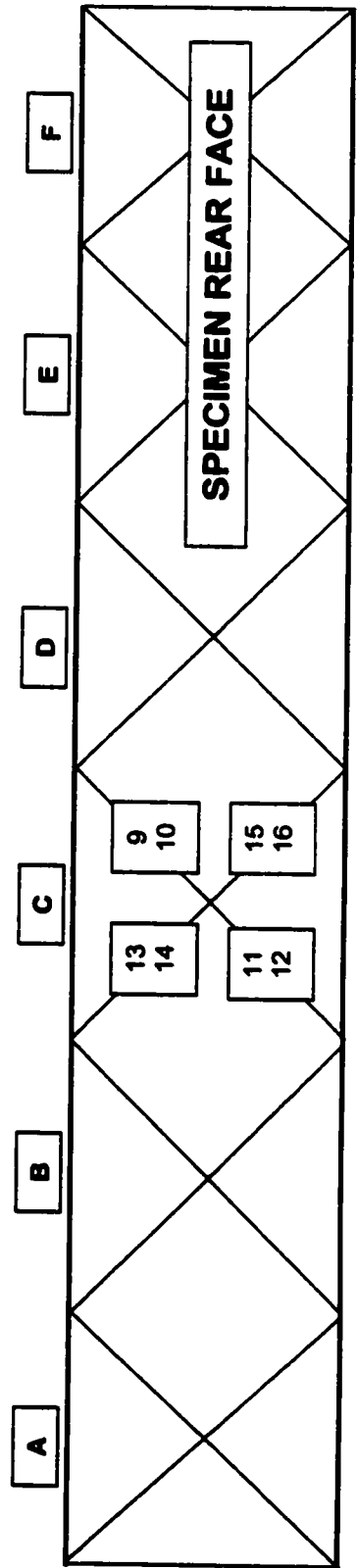
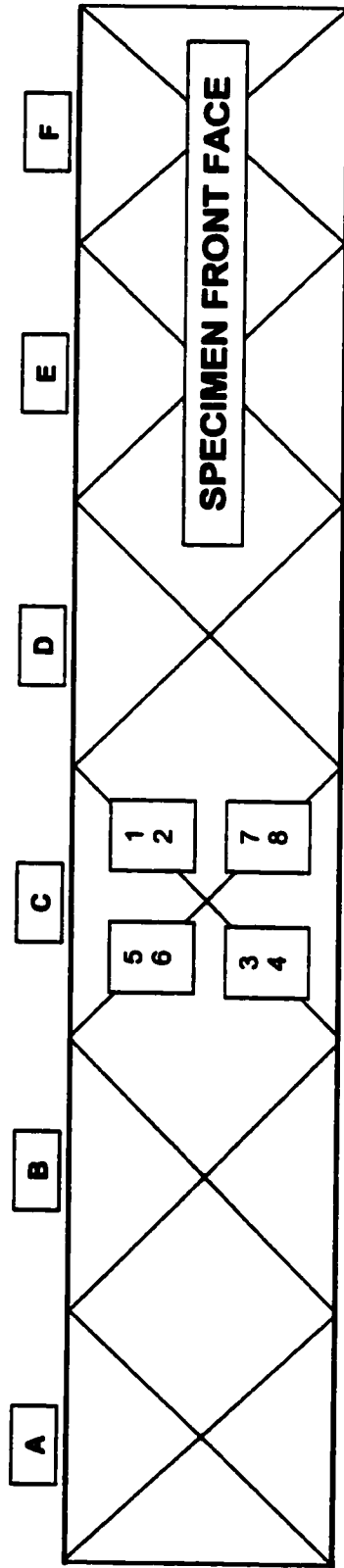
SPECIMEN S2-C (TEST # 23) STRAIN GAUGE IDENTIFICATION



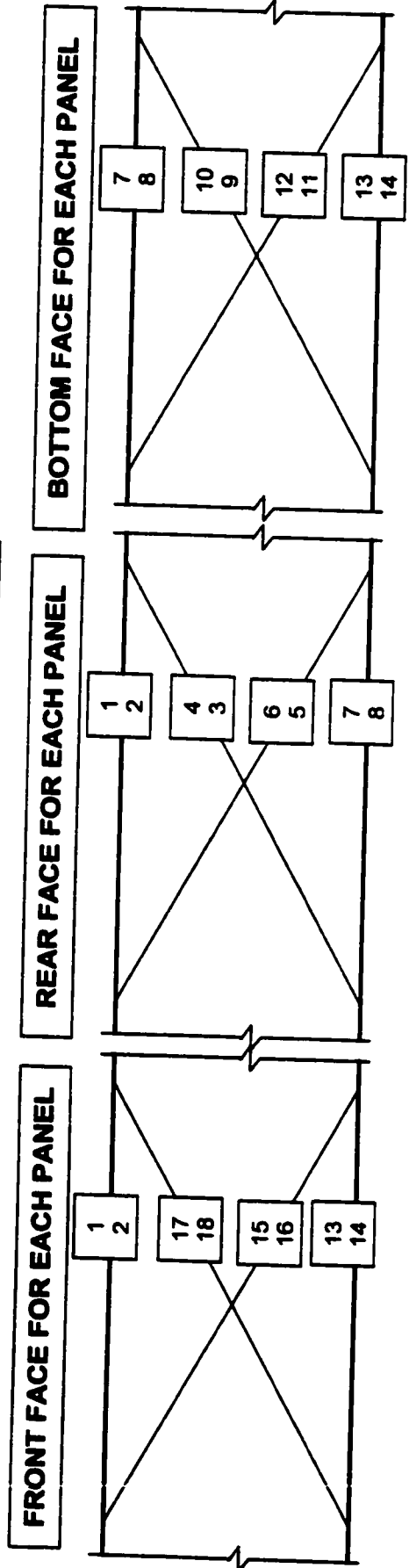
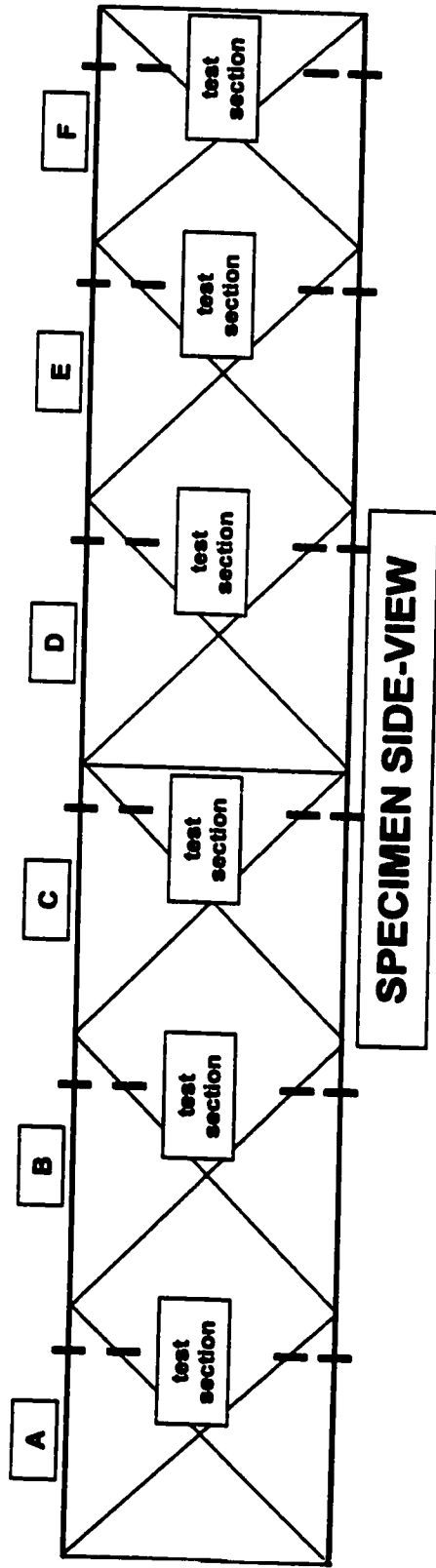
SPECIMEN S3-A (TEST # 7) STRAIN GAUGE IDENTIFICATION



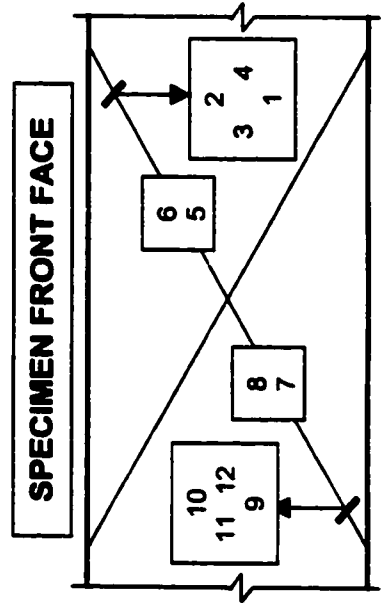
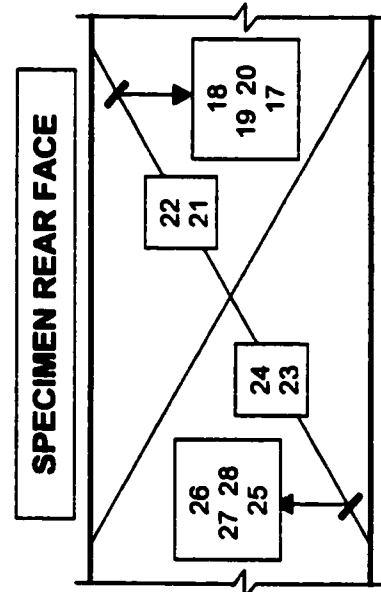
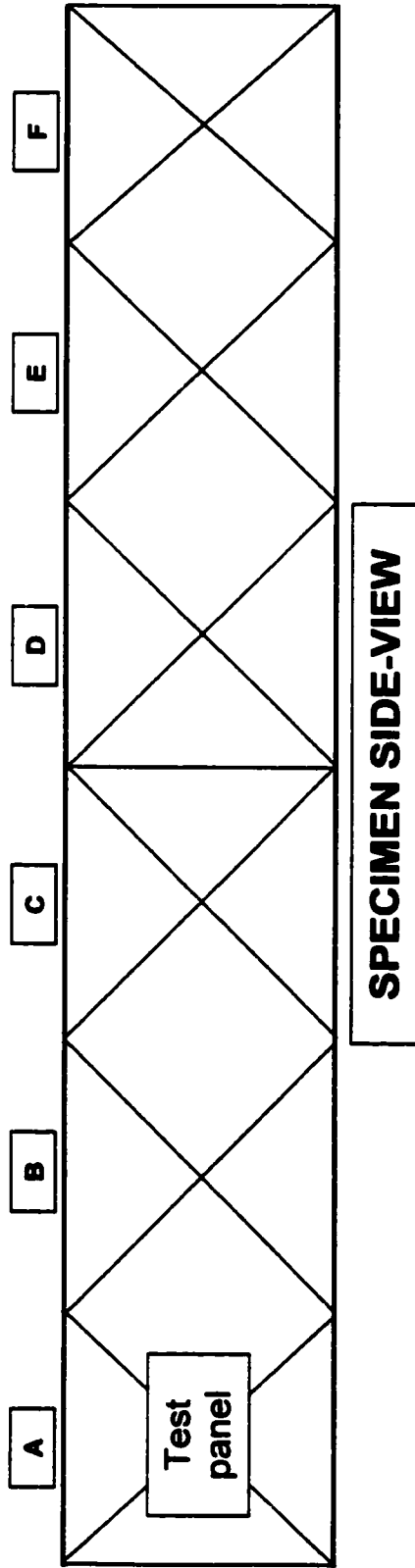
SPECIMEN S3-B (TEST # 25) STRAIN GAUGE IDENTIFICATION



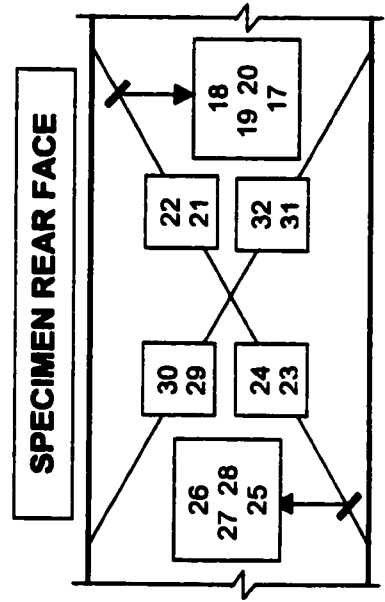
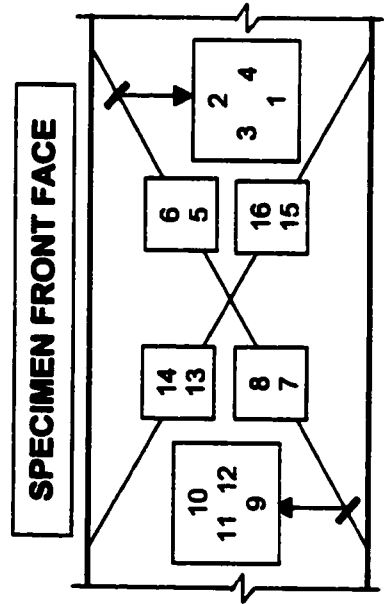
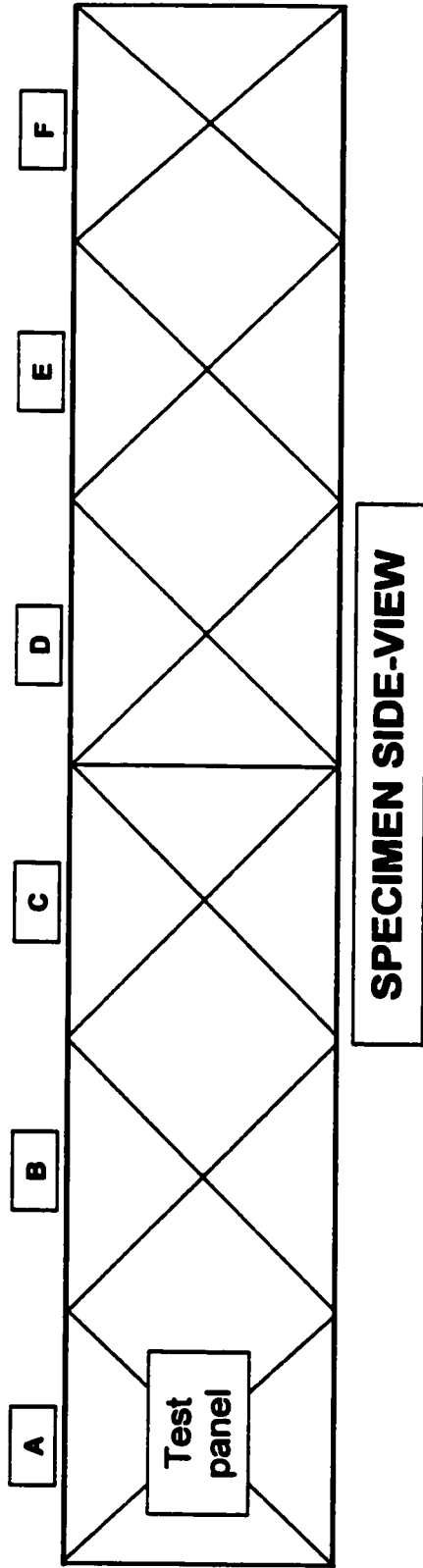
SPECIMEN S4-A (TEST # 1) STRAIN GAUGE IDENTIFICATION



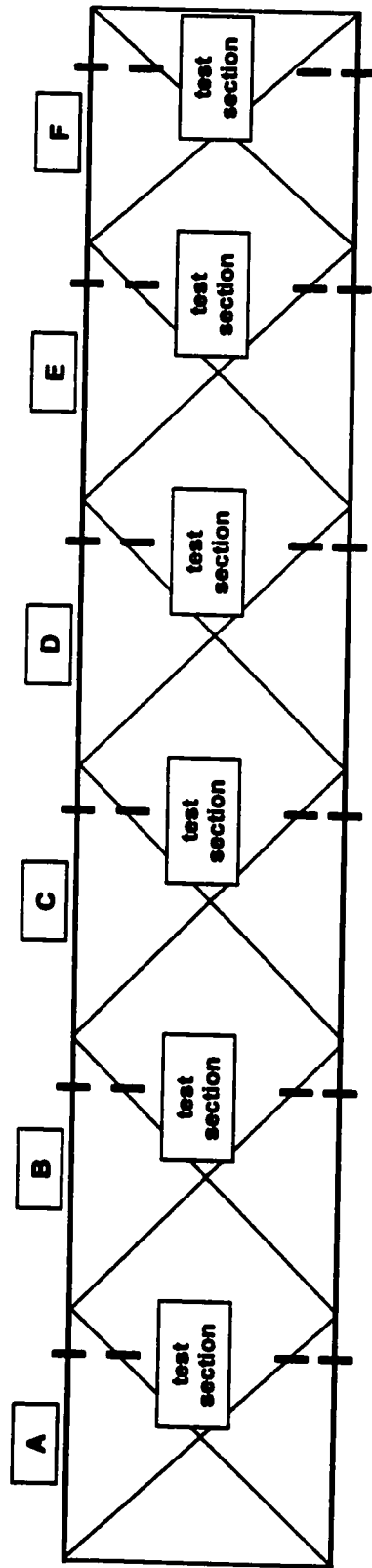
SPECIMEN S4-B (TEST # 3) STRAIN GAUGE IDENTIFICATION



SPECIMEN S4-C (TEST # 5) STRAIN GAUGE IDENTIFICATION

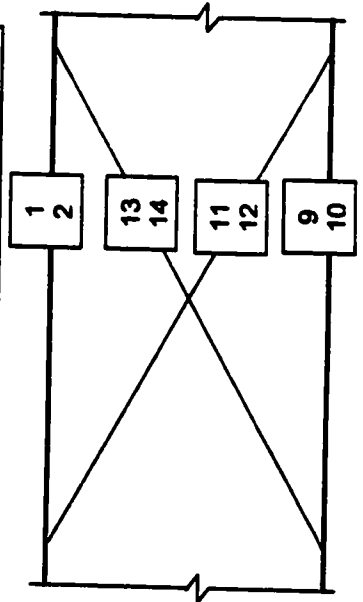


SPECIMEN S5-A (TEST # 2) STRAIN GAUGE IDENTIFICATION

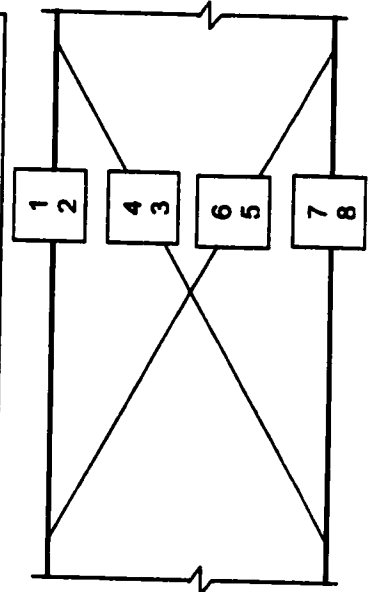


SPECIMEN SIDE-VIEW

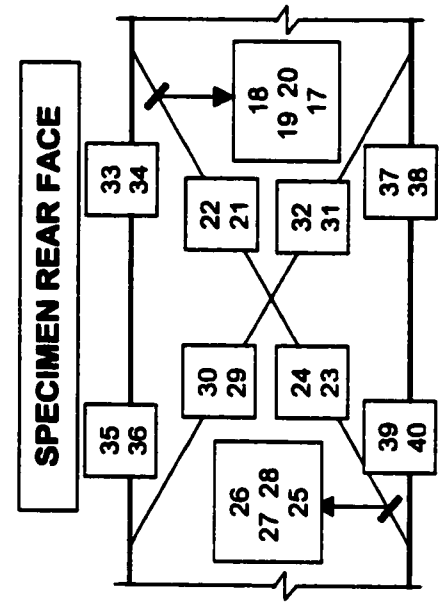
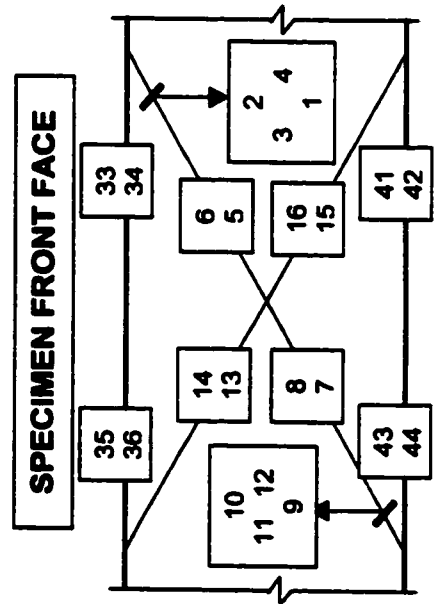
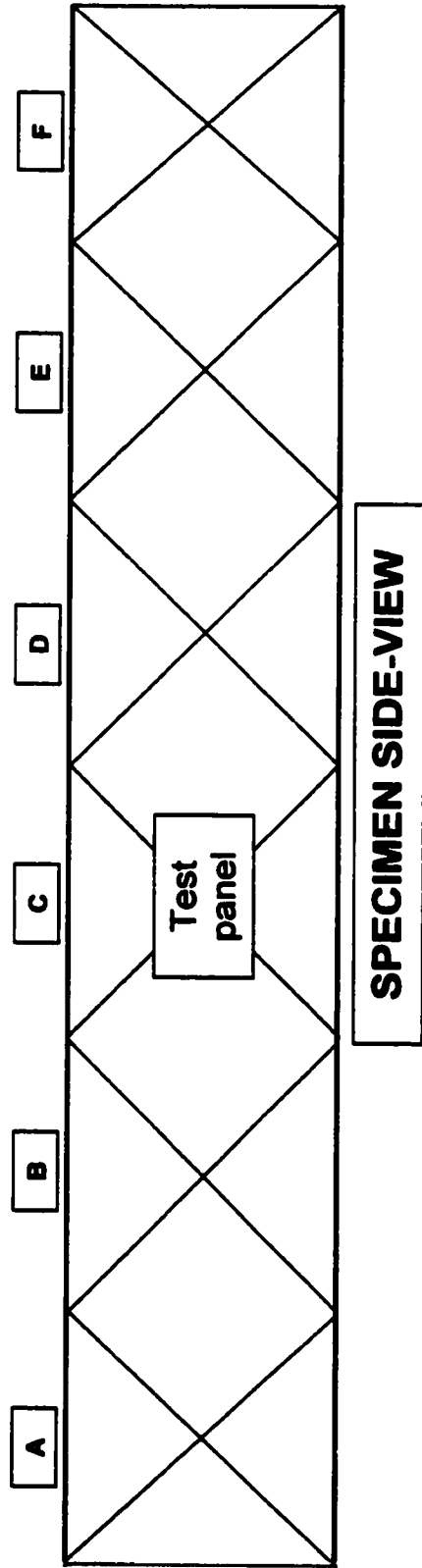
FRONT FACE FOR EACH PANEL



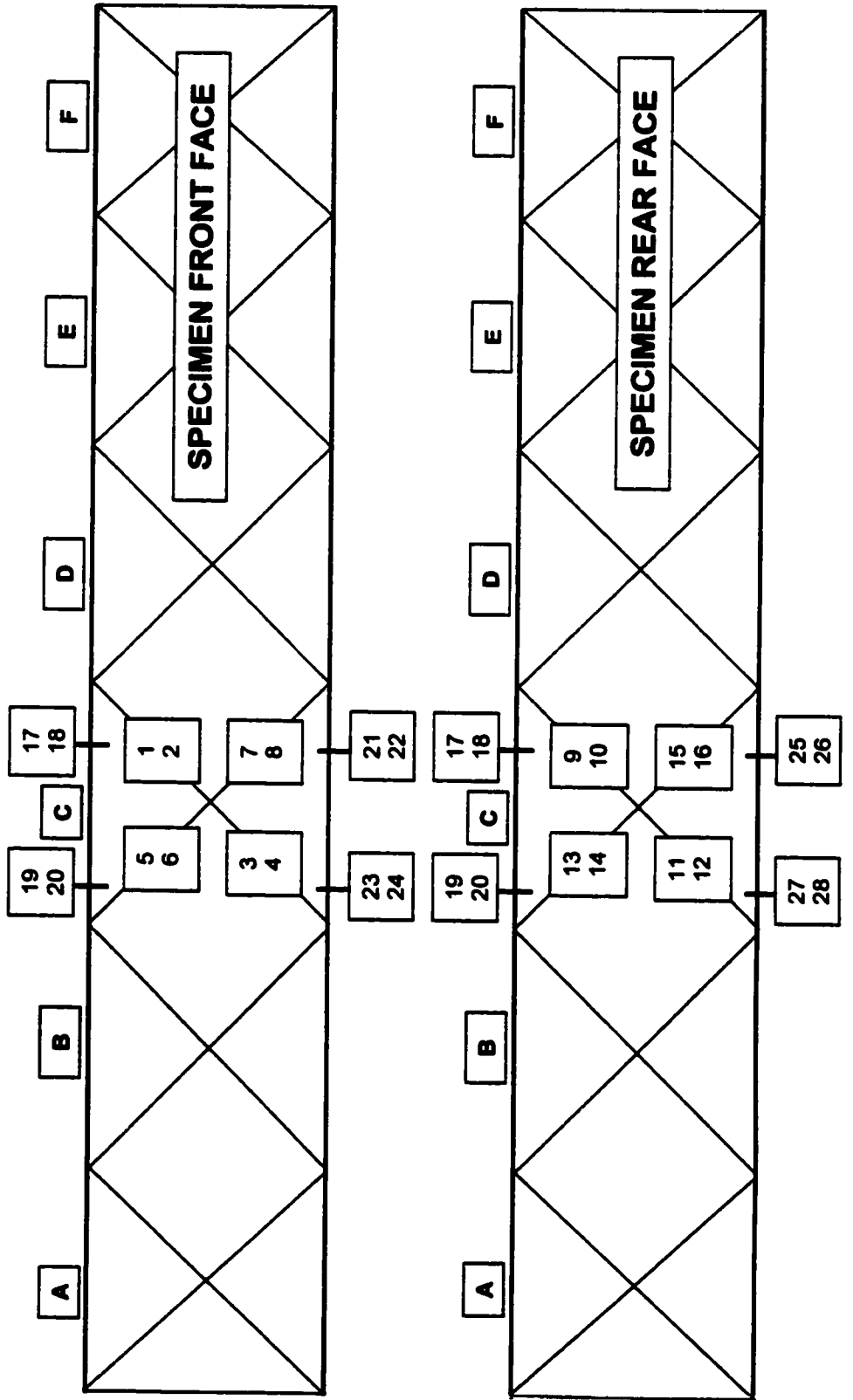
REAR FACE FOR EACH PANEL



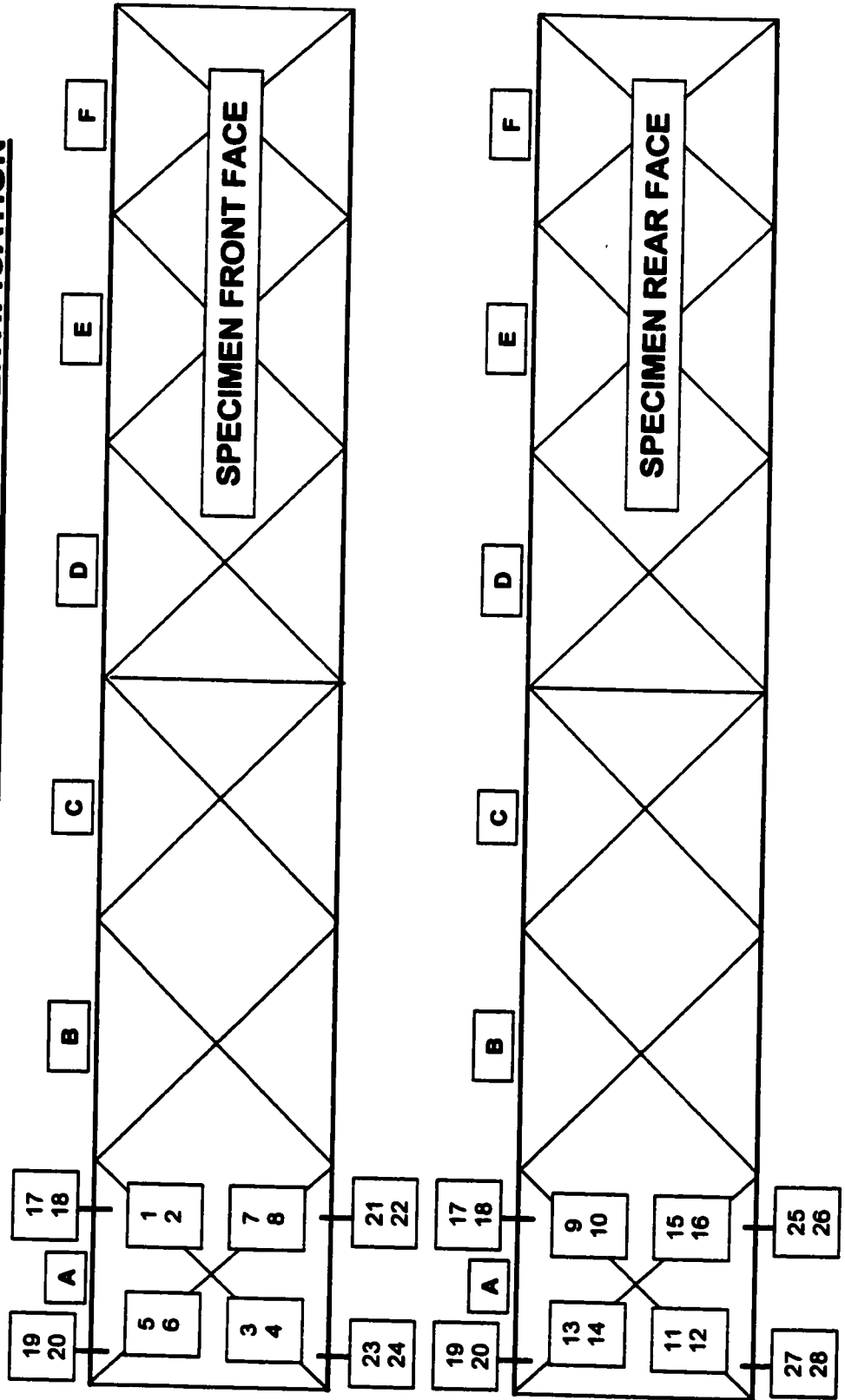
SPECIMEN S5-B (TEST # 10) STRAIN GAUGE IDENTIFICATION



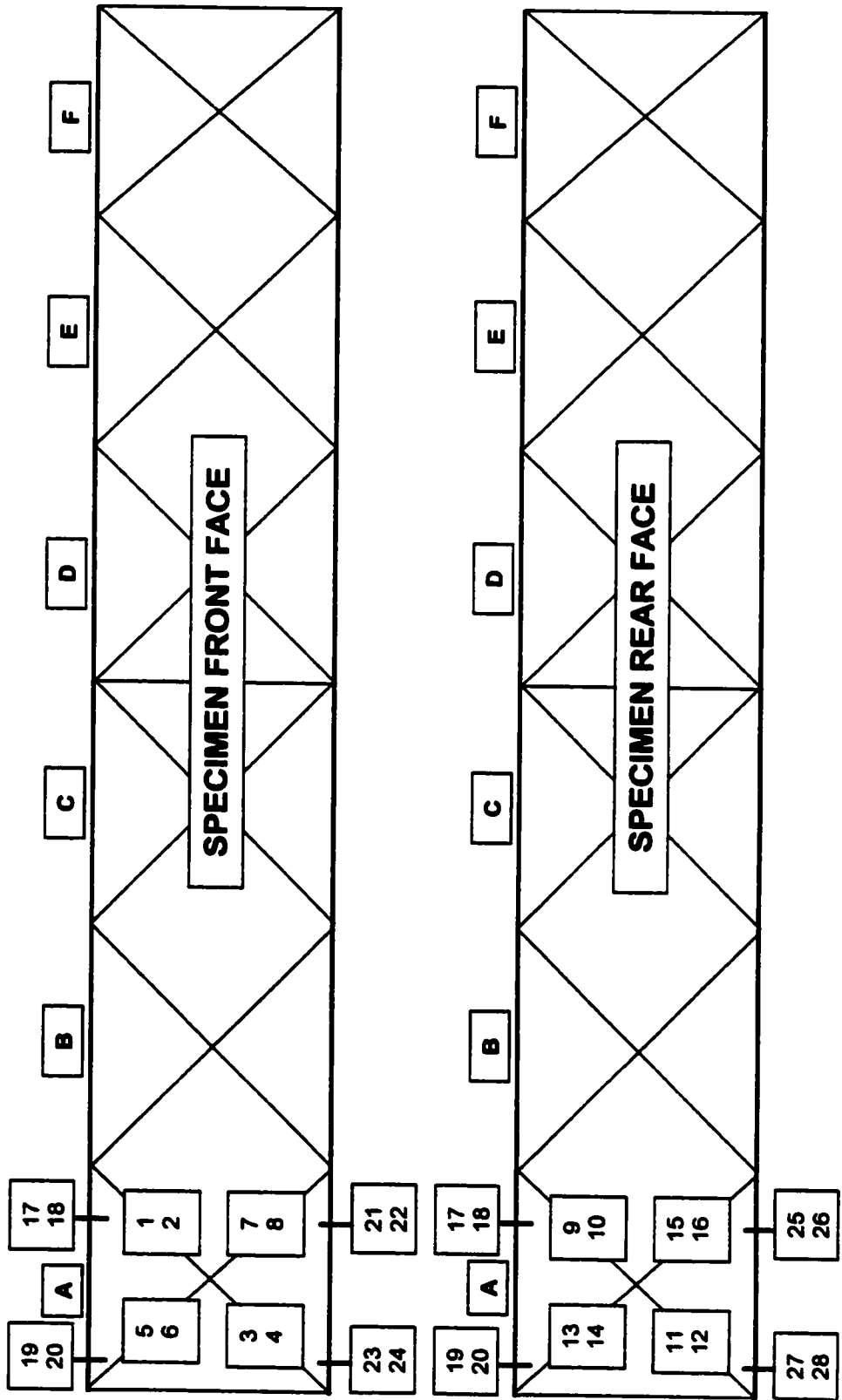
SPECIMEN S5-C (TEST # 12) STRAIN GAUGE IDENTIFICATION



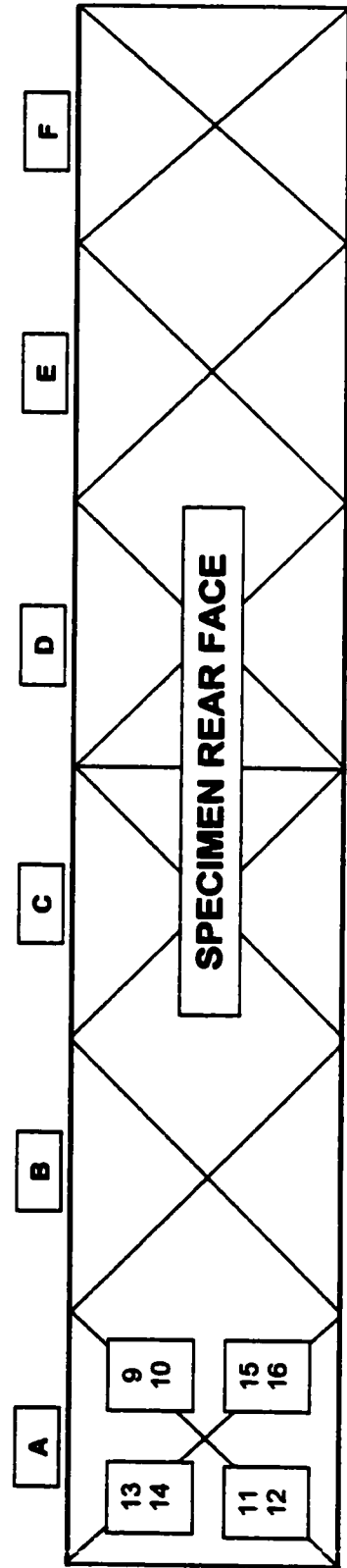
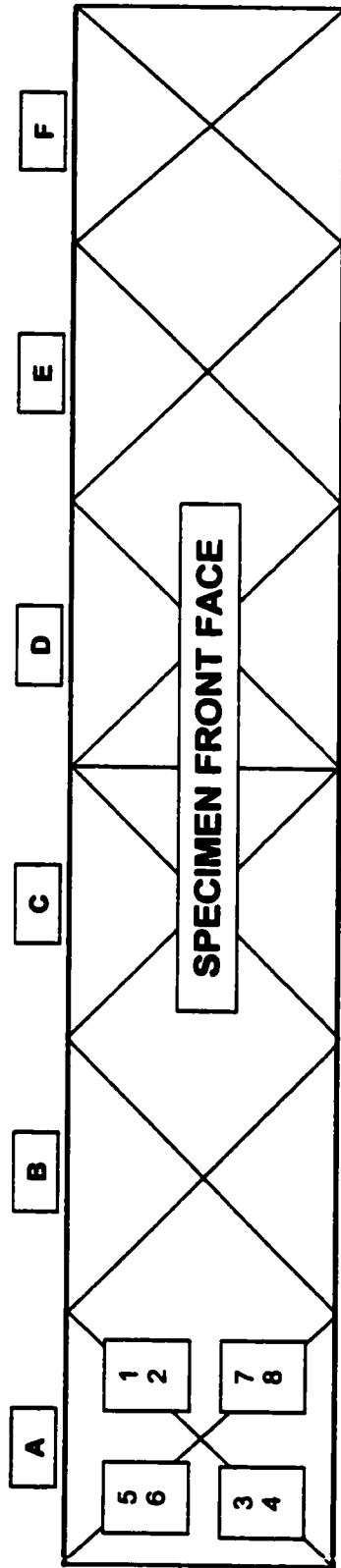
SPECIMEN S6-A (TEST # 11) STRAIN GAUGE IDENTIFICATION



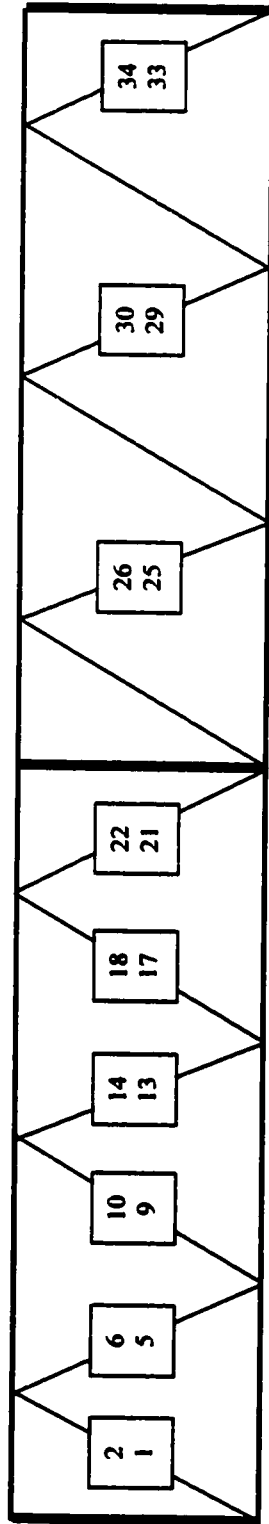
SPECIMEN S6-B (TEST # 13) STRAIN GAUGE IDENTIFICATION



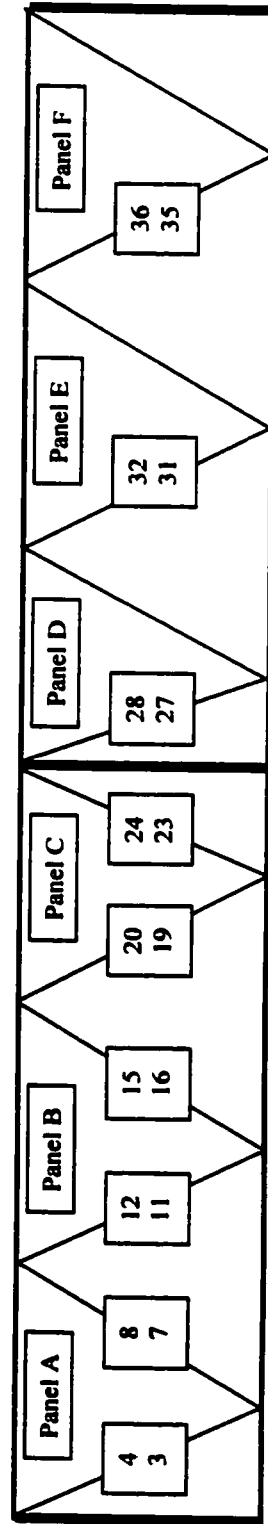
SPECIMEN S6-C (TEST # 24) STRAIN GAUGE IDENTIFICATION



SPECIMEN 31SRB (TEST # 14) STRAIN GAUGE IDENTIFICATION

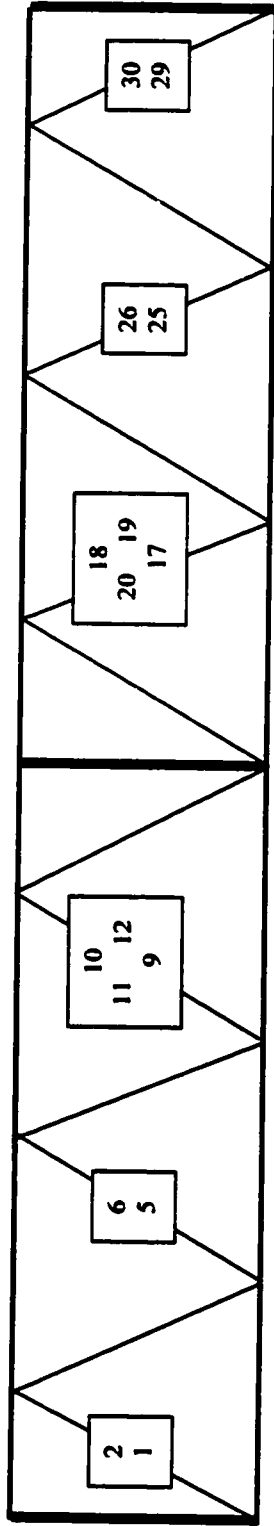


SPECIMEN FRONT FACE

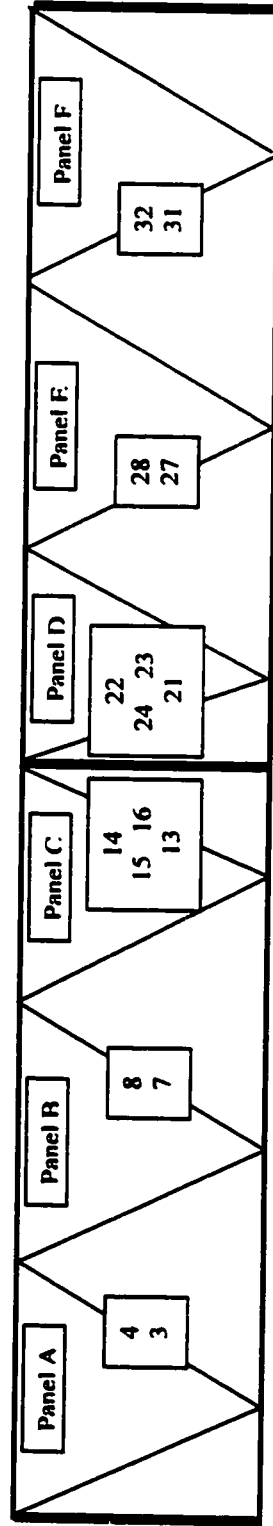


SPECIMEN REAR FACE

SPECIMEN 33SRC (TEST # 16) STRAIN GAUGE IDENTIFICATION



SPECIMEN FRONT FACE



SPECIMEN REAR FACE

APPENDIX B

**LOAD-STRAIN DATA
AND
AXIAL FORCES**

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AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P1-B

TEST PANEL A (REAR DIAGONAL TOP HALF BUCKLED INSIDE, BOTTOM A LITTLE OUTSIDE)

No.	Load (kN)	Strain										Test Serial No. 31		
		9	10	11	12	13	14	15	16	17	18	19	20	
1	-26.7	-18	-22	-33	-200	177	47	14	13	12	11	10	9	
2	-36.4	-20	-31	-46	-285	257	68	47	257	-285	-46	-31	-20	
3	-54.2	-18	-49	-81	-412	352	93	68	352	-412	-81	-49	-18	
4	-65.7	-11	-53	-71	-506	426	76	93	426	-506	-71	-53	-11	
5	-76.8	-1	-70	-81	-596	498	92	76	498	-596	-81	-70	-1	
6	-82.7	9	-768	-85	-650	538	108	92	538	-650	-85	-768	9	
7	-88.6	18	-833	-85	-701	538	141	108	538	-701	-85	-833	18	
8	-97.4	37	-940	-85	-778	575	150	141	575	-778	-85	-940	37	
9	-102.1	49	-1002	-100	-820	634	165	150	634	-820	-100	-1002	49	
10	-102.1	69	-1000	-100	-819	664	173	165	664	-819	-100	-1000	69	
11	-108.7	70	-1088	-105	-880	706	172	173	706	-880	-105	-1088	70	
12	-115.0	94	-1176	-110	-936	751	184	172	751	-936	-110	-1176	94	
13	-114.6	95	-1175	-110	-935	750	194	184	750	-935	-110	-1175	95	
14	-120.9	122	-1287	-116	-981	792	208	194	792	-981	-116	-1287	122	
15	-127.4	159	-1375	-123	-1049	835	216	208	835	-1049	-123	-1375	159	
16	-133.5	204	-1480	-132	-1103	879	228	216	879	-1103	-132	-1480	204	
17	-136.8	258	-1619	-144	-1156	925	237	228	925	-1156	-144	-1619	258	
18	-146.3	343	-1782	-166	-1199	971	245	237	971	-1199	-166	-1782	343	
19	-152.0	515	-2210	-250	-1203	1017	252	245	1017	-1203	-250	-2210	515	
20	-164.9	882	-2366	-258	-1287	1049	264	252	1049	-1287	-258	-2366	882	
21	-88.3	21091	-30290	-2462	1822	968	76	264	968	1822	-2462	-30290	21091	
AXIAL FORCE (kN)		-34.5		-29.0		25.6		26.6						

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P2-A

TEST PANEL A (FRONT DIAGONAL NOT BUCKLED)

		Test Serial No. 21								
No.	Load (kN)	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
		1	2	3	4	5	6	7	8	9
1	-27.6	-424	-48	-387	-102	-68	167	-85		149
2	-34.3	-553	-36	-476	-110	-116	222	-113		200
3	-41.1	-666	-15	-564	-109	-151	264	-145		257
4	-50.3	-807	39	-770	-86	-202	381	-194		347
5	-59.1	-1143	125	-965	-52	-259	481	-248		449
6	-70.0	-1492	283	-1204	22	-335	643	-324		592
7	-74.1	-1645	378	-1305	60	-365	707	-352		653
8	-78.1	-1809	474	-1407	97	-394	772	-382		714
9	-81.7	-1876	587	-1489	126	-420	836	-409		776
10	-83.8	-2059	641	-1520	139	-433	867	-422		805
11	-85.1	-2180	718	-1540	127	-445	900	-434		835
12	-86.4	-2278	819	-1633	83	-487	938	-444		866
13	-88.4	-2711	1018	-1434	-12	-469	964	-451		895
14	-56.6	-3023	1185	-1383	-98	-479	991	-453		915
15	-58.3	-2818	1154	-509	-641	-596	902	-375		659
16	-59.1	-2660	1196	-501	-666	-596	921	-381		675
17	-59.1	-2733	1249	-668	-708	-612	940	-385		693
18	-59.6	-2856	1390	-409	-764	-642	960	-364		704
19	-59.2	-2862	1414	-352	-830	-660	998	-397		707
20	-59.1	-2948	1473	-309	-875	-675	1017	-400		713
21	-59.2	-3047	1622	-219	-967	-704	1050	-405		724
22	-58.9	-3084	1698	-187	-1016	-718	1055	-408		728
23	-57.8	-3454	2032	45	-769	-802	1115	-409		732
24	-57.7	-3607	2198	185	-1207	-831	1150	-411		740
25	-58.8	-3705	2300	339	-1475	-830	1171	-408		732
26	-55.3	-3810	2227	344	-1450	-830	1156	-403		713
27	-54.8	-3436	2064	381	-1468	-814	1116	-385		673
28	-52.4	-3290	1954	468	-1537	-833	1108	-378		641
AXIAL FORCE (kN)		-28.5		-28.9		9.4		8.3		

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P2-A

TEST PANEL A (REAR DIAGONAL TOP HALF BUCKLED OUTSIDE)

No.	Test Serial No. 21													
	Lead (kN)	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain		
1	-27.6	-508	38	-435	10	-115	14	16	108	182				
2	-34.3	-653	73	-561	36	-155	173	225	213					
3	-41.1	-808	120	-684	76	-200	264	294	270					
4	-50.3	-1040	208	-863	158	-268	377	358						
5	-59.1	-1295	328	-1108	264	-341	483	328						
6	-70.0	-1641	517	-1367	410	-437	632	420						
7	-74.1	-1787	608	-1445	446	-477	687	457						
8	-78.1	-1948	719	-1481	481	-519	768	491						
9	-81.7	-2271	808	-1480	437	-556	834	520						
10	-83.6	-2386	985	-1508	442	-573	896	533						
11	-85.1	-2610	1138	-1462	389	-591	900	541						
12	-86.4	-3181	1428	-1318	248	-618	938	643						
13	-88.4	-5087	2490	-418	-576	-618	1014	472						
14	-95.6	-5889	2978	159	-1080	-616	1054	428						
15	-98.3	-21380	15866	938	-1500	-846	881	862						
16	-99.1	-27841	16835	883	-1625	-860	969	808						
17	-99.1	-25818	19186	1060	-1681	-868	1008	772						
18	-99.9	-28042	21688	1158	-1782	-891	1034	828						
19	-99.2	-31287	23459	1225	-1837	-898	1030	788						
20	-99.1	-32851	24584	1274	-1862	-893	1038	830						
21	-99.2	-35137	26878	1370	-1972	-900	1050	814						
22	-99.9	-38482	27823	1428	-2019	-897	1050	814						
23	-97.8	-40385	31202	1628	-2202	-899	1050	814						
24	-97.7	-42443	32987	1758	-2327	-897	1088	819						
25	-96.8	-44521	34785	1883	-2452	-899	1088	813						
26	-95.3	-45025	35212	1913	-2464	-882	1042	788						
27	-94.8	-46248	36278	2045	-2603	-888	1074	805						
28	-92.4	-48410	38043	2282	-2785	-874	1073	788						
AXIAL FORCE (kN)											-29.7	-21.2	6.5	6.4

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P3-A

TEST PANEL A (FRONT DIAGONAL TOP HALF BUCKLED INSIDE , BOTTOM A LITTLE INSIDE, FIRST)

Nt.	Load (kN)	Test Serial No. 30						
		Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-33.7	-270	-14	-247	-37	53	210	245
2	-56.0	-430	-15	-365	-53	62	332	394
3	-83.5	-634	-8	-572	-69	117	475	544
4	-102.6	-779	3	-666	-76	142	571	654
5	-126.2	-957	23	-849	-84	173	699	785
6	-149.3	-1126	47	-991	-86	201	795	904
7	-175.6	-1354	80	-1168	-86	237	924	1048
8	-199.1	-1534	133	-1300	-92	265	1022	1156
9	-208.6	-1661	170	-1377	-101	279	1080	1223
10	-215.4	-1723	186	-1420	-102	288	1113	1259
11	-223.9	-1811	223	-1486	-110	298	1154	1306
12	-223.4	-1812	226	-1462	-113	298	1152	1305
13	-227.6	-1851	259	-1489	-115	303	1174	1327
14	-232.0	-1900	261	-1512	-121	308	1194	1352
15	-236.6	-1951	288	-1532	-131	312	1215	1377
16	-235.8	-1956	299	-1517	-143	308	1213	1375
17	-239.7	-1997	312	-1541	-141	313	1233	1395
18	-244.5	-2096	345	-1563	-153	317	1256	1422
19	-249.0	-2147	363	-1575	-171	319	1278	1447
20	-253.1	-2262	465	-1573	-202	313	1301	1473
21	-251.9	-2331	515	-1544	-229	300	1300	1470
22	-256.8	-2414	557	-1565	-241	304	1324	1497
23	-269.9	-3181	1039	-1668	-422	211	1346	1616
24	-234.8	-26331	1672	1612	-2910	-80	1489	1443
25	-235.4	-27108	17449	1780	-3044	-84	1502	1428
26	-235.4	-27752	18075	1899	-3086	-73	1488	1397
27	-235.7	-26135	18456	1954	-3102	-56	1481	1379
28	-235.1	-26466	18782	2003	-3110	-37	1474	1362
29	-201.5	-26932	19340	2101	-3099	25	1441	1307
AXIAL FORCE (kN)		-55.0		-53.2		44.6		47.8

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P3-A

TEST PANEL A (REAR DIAGONAL TOP HALF BUCKLED A LITTLE INSIDE ,BOTTOM OUTSIDE,SECOND)

No.	Load (kN)	Test Serial No. 30										
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	
1	-33.7	-110	-10	14	12	13	14	15	16	17	18	19
2	-56.0	-218	-37	-281	30	10	6	119	6	119	6	124
3	-83.5	-357	-62	-482	55	18	55	347	245	245	-13	245
4	-102.8	-453	-78	-616	78	25	78	438	502	347	-21	368
5	-128.2	-65	-95	-788	113	37	113	557	37	557	-25	634
6	-148.3	-666	-106	-972	155	49	155	688	49	688	-23	759
7	-175.6	-842	-124	-1199	215	69	215	805	69	805	-17	914
8	-198.1	-853	-134	-1378	271	84	271	910	84	910	-10	1030
9	-208.6	-1024	-139	-1491	311	94	311	978	94	978	-4	1104
10	-215.4	-1056	-142	-1548	332	100	332	1010	100	1010	0	1141
11	-223.6	-1105	-146	-1629	359	106	359	1054	106	1054	4	1188
12	-223.4	-1104	-145	-1628	359	106	359	1053	106	1053	4	1187
13	-227.6	-1124	-147	-1664	374	112	374	1075	112	1075	6	1210
14	-232.0	-1150	-148	-1708	390	116	390	1099	116	1099	10	1237
15	-238.6	-1173	-151	-1750	408	120	408	1124	120	1124	14	1283
16	-235.6	-1171	-148	-1747	408	118	408	1124	118	1124	13	1281
17	-239.7	-1191	-151	-1783	420	122	420	1144	122	1144	15	1284
18	-244.5	-1217	-153	-1830	440	128	440	1170	128	1170	19	1311
19	-248.0	-1242	-155	-1890	461	132	461	1185	132	1185	22	1339
20	-253.1	-1265	-154	-1920	482	134	482	1222	134	1222	26	1367
21	-251.8	-1282	-153	-1935	484	131	484	1224	131	1224	24	1366
22	-256.8	-1289	-155	-1984	504	137	504	1250	137	1250	28	1383
23	-256.0	-1339	-128	-2055	543	141	543	1282	141	1282	35	1433
24	-234.6	-1471	-88	-2747	682	144	682	1425	144	1425	9	1457
25	-235.4	-1477	-126	-2684	1101	146	1101	1473	146	1473	5	1653
26	-235.4	-1435	-184	-3341	1330	142	1330	1527	142	1527	-10	1717
27	-298.7	-1319	-339	-3999	1643	139	1643	1688	139	1688	-29	1787
28	-235.1	-1824	-823	-5082	2277	132	2277	1625	132	1625	-70	1883
29	-201.5	1031	-2424	-23934	15182	-46	15182	1469	-46	1469	-434	1695
AXIAL FORCE (kN)		-47.0		-52.0		49.0		50.1				

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P3-B

TEST PANEL A (REAR DIAGONAL, BOTTOM BUCKLED OUTSIDE, TOP A LITTLE INSIDE)

No.	Load (kN)	Test Serial No. 32												
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-37.5	-133	-18	-139	-4	141	16	13	14	15	16			
2	-67.8	-265	-34	-277	-5	278	15	7	141	17	134			
3	-88.3	-382	-45	-390	-5	375	23	23	278	261	261			
4	-119.2	-511	-60	-541	-2	523	37	37	375	48	354			
5	-148.5	-652	-72	-693	12	659	53	53	523	72	482			
6	-194.0	-910	-86	-982	45	900	80	80	659	98	622			
7	-202.5	-956	-89	-1039	54	948	96	96	900	151	849			
8	-213.1	-1010	-91	-1068	65	991	107	107	948	182	991			
9	-222.6	-1050	-94	-1152	76	1028	116	116	991	174	935			
10	-228.4	-1070	-96	-1177	82	1048	120	120	1028	185	971			
11	-233.7	-1089	-96	-1203	87	1083	124	124	1048	190	987			
12	-236.1	-1105	-98	-1228	93	1124	128	128	1083	194	1003			
13	-234.2	-1134	-100	-1280	99	1165	134	134	1124	199	1018			
14	-243.5	-1180	-103	-1325	112	1211	144	144	1165	207	1045			
15	-252.4	-1243	-105	-1369	133	1261	156	156	1211	221	1091			
16	-260.4	-1290	-108	-1438	149	1317	168	168	1261	235	1136			
17	-267.9	-1343	-108	-1536	176	1372	182	182	1317	246	1175			
18	-272.1	-1355	-108	-1536	176	1372	182	182	1372	261	1218			
19	-221.6	-1046	-7	-2263	699	1491	226	226	1491	262	1445			
20	-225.9	-1063	-2	-2317	699	1491	226	226	1491	262	1445			
21	-279.5	-1717	-4	-2408	725	1522	230	230	1491	262	1445			
22	-232.5	-11725	-29	-2568	783	1557	235	235	1522	300	1475			
23	-244.3	-1878	-109	-2784	873	1595	241	241	1557	308	1508			
24	-203.3	1201	-2669	-25976	1043	1637	246	246	1595	312	1545			
25	0.3	944326	-787874	844212	18799	18799	100	100	1637	313	1686			
26	-6.4	383	383	944354	844212	944354	943078	943078	944354	80	1987			
				411	269	269	-60	-60	943078	-935361	4084			
									134	116	379			
AXIAL FORCE (kN)	-50.7			-47.3					53.6					54.2

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P4-A

TEST PANEL D (FRONT DIAGONAL NOT BUCKLED)

No.	Test Serial No. 29								
	Lead (kN)	Strain	Strain	Strain	Strain	Strain	Strain	Strain	
1	-29.3								
2	-40.4	-75	-324	-51	-350	160	-43	140	
3	-51.6	-91	-441	-55	-478	215	-64	188	
4	-62.7	-100	-597	-53	-614	273	-88	240	
5	-74.9	-100	-697	-41	-756	334	-116	294	
6	-86.6	-96	-850	-18	-923	405	-153	356	
7	-98.8	-86	-1003	20	-1088	478	-182	421	
8	-105.8	-27	-1172	73	-1273	561	-238	493	
9	-112.4	-1	-1272	108	-1390	609	-287	538	
10	-126.4	31	-1374	145	-1487	682	-298	583	
11	-129.5	98	-1572	224	-1700	783	-368	680	
12	-136.4	111	-1619	243	-1747	812	-385	716	
13	-139.7	145	-1710	276	-1838	880	-425	778	
14	-143.1	182	-1752	287	-1873	919	-448	811	
15	-146.1	182	-1798	306	-1917	953	-465	842	
16	-152.2	204	-1842	324	-1957	984	-483	870	
17	-155.0	256	-1931	325	-1991	1064	-527	944	
18	-159.8	296	-1985	318	-1995	1103	-547	983	
19	-164.3	405	-2119	265	-1982	1167	-571	1049	
20	-132.5	802	-2389	181	-1798	1231	-688	1127	
21	-133.5	844	-2382	-184	-1214	1088	-521	982	
22	-133.3	828	-2473	-28	-1426	1144	-539	982	
23	-132.9	971	-2586	37	-1501	1171	-548	997	
24	-133.9	1001	-2608	50	-1512	1170	-551	1003	
25	-133.8	1134	-2643	66	-1536	1192	-558	1012	
26	-134.4	1162	-2772	122	-1587	1228	-570	1040	
27	-134.0	1204	-2805	136	-1607	1238	-574	1050	
28	-133.4	1210	-2830	163	-1616	1243	-578	1051	
29	-120.8	1071	-2827	168	-1612	1241	-578	1047	
			-2458	171	-1385	1083	-507	884	
AXIAL FORCE (kN)		-49.0		-48.4		17.8		18.5	

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P4-A

No.	Load (kN)	Test Serial No. 29									
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-29.3	-56	-106	-81	-174	36	14	14	14	18	18
2	-40.4	-73	-292	-110	-257	69	64	64	64	71	71
3	-51.6	-86	-392	-136	-343	104	80	80	80	101	101
4	-62.7	-90	-503	-156	-436	144	118	118	118	133	133
5	-74.9	-84	-636	-178	-550	197	147	147	147	169	169
6	-86.6	-69	-775	-188	-661	254	192	192	192	215	215
7	-98.9	-36	-836	-184	-767	326	254	254	254	287	287
8	-105.8	-16	-1032	-193	-862	399	286	286	286	315	315
9	-112.4	14	-1136	-190	-937	417	316	316	316	347	347
10	-120.4	85	-1370	-173	-1102	526	349	349	349	382	382
11	-129.5	116	-1426	-167	-1141	554	371	371	371	418	418
12	-136.4	168	-1553	-155	-1224	614	420	420	420	458	458
13	-139.7	200	-1622	-151	-1263	649	437	437	437	475	475
14	-143.1	230	-1686	-146	-1301	681	468	468	468	516	516
15	-146.1	260	-1748	-144	-1332	714	498	498	498	559	559
16	-152.2	340	-1887	-167	-1363	795	534	534	534	578	578
17	-155.0	399	-1957	-168	-1385	835	579	579	579	627	627
18	-159.8	501	-2102	-244	-1344	912	602	602	602	651	651
19	-164.3	711	-2443	-346	-1271	967	642	642	642	683	683
20	-132.5	5180	-6246	-1506	368	1190	1006	1006	1006	878	878
21	-133.5	12653	-17171	-1824	799	1253	1101	1101	1101	1059	1059
22	-133.3	16185	-21374	-1944	964	1253	1099	1099	1099	1131	1131
23	-132.9	17247	-22565	-1974	1036	1250	1070	1070	1070	1136	1136
24	-133.9	17781	-23180	-2007	1067	1262	1067	1067	1067	1136	1136
25	-133.8	21300	-21129	-2136	1247	1271	1075	1075	1075	1146	1146
26	-134.4	21925	-21824	-2173	1282	1282	1081	1081	1081	1149	1149
27	-134.0	24275	-30425	-2265	1400	1286	1087	1087	1087	1156	1156
28	-133.4	25108	-31330	-2295	1442	1287	1087	1087	1087	1154	1154
29	-120.8	26203	-35787	-2431	1626	1302	1095	1095	1095	1165	1165
							1073	1073	1073	1132	1132
AXIAL FORCE (kN)		-46.7	-46.1	8.9	7.0						

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P4-B

TEST PANEL D (FRONT DIAGONAL TOP HALF BUCKLED INSIDE)

No.	Load (kN)	Test Serial No. 20						
		1	2	3	4	5	6	7
1	-24.1	-273	-41	-238	-80	-26	100	121
2	-41.1	-483	-51	-407	-125	-55	184	184
3	-58.9	-715	-41	-582	-160	-92	288	208
4	-77.5	-875	-6	-782	-184	-140	300	300
5	-82.8	-1201	45	-858	-182	-184	350	403
6	-107.8	-1445	119	-1126	-191	-236	439	484
7	-118.9	-1839	186	-1246	-182	-279	525	564
8	-128.3	-1772	245	-1316	-197	-309	588	647
9	-132.4	-1809	308	-1382	-210	-335	682	739
10	-135.9	-2025	355	-1378	-225	-354	722	785
11	-136.5	-2119	410	-1365	-239	-371	751	833
12	-142.9	-2254	503	-1381	-278	-381	784	870
13	-146.1	-2402	608	-1382	-320	-408	814	915
14	-148.3	-2627	746	-1389	-360	-439	848	958
15	-152.2	-3097	1108	-1136	-568	-452	879	1003
16	-153.5	-4781	2159	-412	-1197	-481	918	1082
17	-153.5	-9198	4882	103	-1620	-480	926	1183
18	-154.8	-14028	7650	481	-1913	-468	936	1218
19	-156.3	-17018	9849	897	-2108	-457	944	1282
20	-157.8	-19293	11648	889	-2268	-445	952	1288
21	-159.5	-20717	12735	1018	-2384	-438	959	1329
22	-161.3	-22378	14043	1177	-2514	-422	987	1354
23	-162.7	-23377	14816	1280	-2607	-412	973	1379
24	-163.4	-24564	15782	1415	-2719	-398	978	1386
25	-165.3	-25840	16717	1612	-2807	-375	987	1416
26	-165.5	-28873	18874	1743	-3043	-359	980	1450
27	-166.3	-27177	17137	1844	-3163	-347	983	1485
28	-168.3	-27881	17432	1952	-3274	-336	986	1477
29	-165.5	-28078	17707	2055	-3382	-324	985	1489
30	-131.2	-27158	17019	1973	-3086	-408	972	1493
AXIAL FORCE (kN)		-51.5		-48.5		11.9		15.2

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P4-B

TEST PANEL D (REAR DIAGONAL TOP HALF BUCKLED OUTSIDE)

No.	Test Serial No. 20									
	Lead (kN)	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-24.1	-163	63	-150	-73	14	13	14	14	18
2	-41.1	-262	-119	-270	-119	4	-56	4	4	-34
3	-58.0	-447	-138	-413	-163	14	-91	36	12	-61
4	-77.5	-636	-162	-581	-186	70	-137	70	33	-100
5	-82.6	-781	-170	-737	-211	112	-184	112	70	-146
6	-107.8	-977	-183	-914	-208	167	-245	167	115	-201
7	-118.9	-1130	-141	-1084	-190	222	-269	222	177	-266
8	-126.3	-1240	-119	-1173	-166	284	-340	284	237	-328
9	-132.4	-1336	-82	-1280	-139	303	-378	303	285	-378
10	-135.9	-1387	-73	-1331	-119	330	-404	330	331	-421
11	-139.5	-1460	-52	-1395	-96	359	-431	359	359	-450
12	-142.9	-1524	-27	-1458	-73	422	-461	422	384	-482
13	-146.1	-1583	-5	-1517	-49	490	-490	490	430	-517
14	-149.3	-1647	23	-1579	-22	458	-523	458	485	-551
15	-152.2	-1710	54	-1642	6	500	-559	500	508	-599
16	-153.5	-1768	81	-1709	44	543	-593	543	557	-632
17	-154.8	-1800	94	-1760	76	568	-610	568	612	-679
18	-156.3	-1842	117	-1805	100	610	-636	610	652	-711
19	-157.8	-1886	147	-1856	118	654	-669	654	705	-751
20	-159.5	-1937.8	186	-1909	127	708	-704	708	761	-785
21	-168.8	-1991.8	229	-1974	131	764	-738	764	820	-838
22	-181.3	-2083	304	-1954	109	818	-780	818	874	-877
23	-182.7	-2144	381	-1827	60	873	-817	873	938	-910
24	-183.4	-2323	543	-1886	-37	954	-871	954	998	-951
25	-185.3	-2748	654	-1286	-360	1124	-879	1124	1050	-983
26	-185.5	-3146	1256	-853	-653	1222	-1035	1222	1163	-1026
27	-186.3	-3480	1526	-700	-681	1293	-1073	1293	1218	-1035
28	-186.3	-3842	1815	-487	-1050	1351	-1101	1351	1260	-1039
29	-185.5	-4442	2209	-214	-1242	1401	-1136	1401	1297	-1040
30	-131.2	-9251	5889	392	-1502	1410	-1440	1410	1328	-1052
AXIAL FORCE (kN)										
-50.2										
-49.7										
0.5										
-0.1										

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P5-A

TEST PANEL A (FRONT HORIZONTAL BUCKLED FIRST, THEN FRONT PANEL D&E DIAGONAL BUCKLED OUTSIDE FRONT DIAGONAL NOT BUCKLED)

No.	Load (kN)	Test Serial No. 26											
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain		
1	-35.3	-179	2	-145	43	135	27	153	7	153	6		
2	-95.3	-469	16	-375	111	347	73	347	111	347	73		
3	-127.6	-633	29	-501	149	461	97	461	149	461	97		
4	-161.4	-807	49	-633	199	575	124	575	199	575	124		
5	-196.4	-994	77	-767	230	693	152	693	230	693	152		
6	-230.9	-1188	112	-902	273	810	180	810	273	810	180		
7	-265.3	-1407	160	-1038	315	924	211	924	315	924	211		
8	-299.4	-1674	239	-1159	360	1036	242	1036	360	1036	242		
9	-321.6	-1893	311	-1235	390	1116	264	1116	390	1116	264		
10	-343.2	-2137	401	-1301	421	1181	285	1181	421	285	1357		
11	-354.0	-2271	453	-1330	437	1229	296	1229	437	296	1401		
12	-357.4	-2346	482	-1335	443	1242	300	1242	443	300	1418		
13	-365.0	-2443	519	-1355	453	1269	307	1269	453	307	1449		
14	-376.0	-2597	565	-1383	469	1309	318	1309	469	318	1485		
15	-378.6	-2687	616	-1397	476	1321	324	1321	476	324	1506		
16	-383.7	-2739	646	-1393	481	1336	326	1336	481	326	1525		
17	-382.3	-2766	674	-1378	481	1332	327	1332	481	327	1524		
18	-381.6	-2821	727	-1366	484	1365	336	1365	484	336	1560		
19	-384.5	-2860	759	-1400	499	1375	340	1375	499	340	1572		
20	-386.6	-3125	818	-1362	502	1382	343	1382	502	343	1581		
21	-389.0	-3171	840	-1360	507	1384	346	1384	507	346	1585		
22	-402.6	-3233	899	-1366	512	1407	351	1407	512	351	1609		
23	-405.9	-3334	916	-1363	522	1425	356	1425	516	356	1622		
24	-408.2	-3452	966	-1352	523	1437	355	1437	523	355	1632		
25	-399.8	-3368	896	-1366	528	1425	356	1425	528	356	1632		
26	-404.3	-3411	908	-1371	528	1425	356	1425	908	356	1572		
27	-408.0	-3436	916	-1371	528	1425	356	1425	916	356	1583		
28	-409.9	-3474	929	-1381	534	1391	356	1391	929	356	1606		
29	-412.5	-3508	941	-1386	536	1366	356	1366	941	356	1615		
30	-413.7	-3529	947	-1391	539	1366	356	1366	947	356	1617		
AXIAL FORCE (kN)		-69.3										75.0	76.7

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P5-A

TEST PANEL A (REAR DIAGONAL NOT BUCKLED)

No.	Load (kN)	Test Serial No. 26									
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-35.3	-129	-29	-125	19	13	14	18	19	10	117
2	95.3	-372	-80	-381	56	56	300	332	336	332	117
3	-127.6	-504	-104	-491	80	100	401	445	445	445	445
4	-181.4	-648	-128	-634	104	104	512	567	567	567	567
5	-198.4	-802	-145	-784	131	131	628	682	682	682	682
6	-230.8	-857	-162	-838	158	158	740	815	815	815	815
7	-265.3	-1116	-171	-1088	186	186	853	942	942	942	942
8	-298.4	-1282	-178	-1254	214	214	967	1069	1069	1069	1069
9	-321.6	-1397	-173	-1361	233	233	1042	1116	1116	1116	1116
10	-343.2	-1513	-168	-1489	251	251	1116	1184	1184	1184	1184
11	-364.0	-1574	-163	-1523	269	269	1184	1237	1237	1237	1237
12	-357.4	-1587	-158	-1543	281	281	1153	1278	1278	1278	1278
13	-385.0	-1642	-156	-1583	283	283	1187	1293	1293	1293	1293
14	-376.0	-1703	-151	-1640	280	280	1192	1311	1311	1311	1311
15	-379.6	-1724	-147	-1658	283	283	1192	1320	1320	1320	1320
16	-383.7	-1747	-145	-1680	286	286	1242	1329	1329	1329	1329
17	-382.3	-1749	-142	-1676	284	284	1257	1353	1353	1353	1353
18	-391.6	-1801	-138	-1725	284	284	1253	1362	1362	1362	1362
19	-384.5	-1817	-135	-1736	288	288	1284	1382	1382	1382	1382
20	-385.6	-1836	-128	-1750	288	288	1284	1426	1426	1426	1426
21	-399.0	-1854	-128	-1767	288	288	1285	1438	1438	1438	1438
22	-402.6	-1873	-127	-1786	300	300	1301	1446	1446	1446	1446
23	-405.9	-1882	-124	-1786	304	304	1313	1450	1450	1450	1450
24	-408.2	-1885	-124	-1803	308	308	1324	1472	1472	1472	1472
25	-399.9	-1822	-171	-1817	312	312	1335	1485	1485	1485	1485
26	-404.3	-1833	-181	-1791	332	332	1342	1483	1483	1483	1483
27	-408.0	-1836	-184	-1803	343	343	1385	1488	1488	1488	1488
28	-408.9	-1854	-187	-1822	346	346	1317	1481	1481	1481	1481
29	-412.5	-1882	-184	-1836	351	351	1324	1489	1489	1489	1489
30	-413.7	-1889	-280	-1843	357	357	1335	1501	1501	1501	1501
					363	363	1343	1512	1512	1512	1512
					363	363	1344	1516	1516	1516	1516
AXIAL FORCE (kN)		-79.8									
		66.3									
		67.9									

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P5-B

TEST PANEL A (FRONT DIAGONAL NOT BUCKLED), PANEL H HORIZONTAL AND REAR DIAGONAL BUCKLED

FIRST

No.	Load (kN)	Test Serial No. 34									
		Strain	1	2	3	4	5	6	7	8	9
1	-40.2	-57	-187	-29	-214	185	174	0	0	0	
2	-88	-310	-39	-357	273	284	15	15	15	15	
3	-86.6	-116	-441	-45	-509	384	387	25	25	25	
4	-133.5	-143	-590	-43	-685	512	524	38	38	38	
5	-226.7	-182	-880	-3	-1156	847	846	78	78	78	
6	-281.2	-208	-1125	26	-1338	970	981	82	82	82	
7	-283.6	-210	-1264	65	-1518	1086	1071	107	107	107	
8	-323.5	-213	-1399	111	-1700	1200	1174	121	121	121	
9	-354.4	-216	-1535	187	-1885	1314	1281	139	139	139	
10	-385.9	-215	-1599	191	-1973	1359	1320	146	146	146	
11	-377.0	-218	-1638	217	-2046	1400	1359	153	153	153	
12	-389.6	-215	-1700	252	-2148	1453	1408	158	158	158	
13	-397.3	-218	-1732	272	-2202	1481	1432	163	163	163	
14	-406.3	-216	-1780	316	-2315	1533	1478	167	167	167	
15	-422.6	-218	-1851	359	-2423	1587	1528	177	177	177	
16	-428.8	-220	-1970	381	-2468	1608	1544	179	179	179	
17	-433.8	-228	-1991	422	-2567	1838	1874	188	188	188	
18	-440.3	-226	-1830	450	-2819	1888	1870	184	184	184	
19	-442.4	-228	-1941	469	-2858	1878	1810	186	186	186	
20	-445.8	-230	-1855	483	-2880	1882	1822	187	187	187	
21	-449.1	-234	-1868	505	-2731	1707	1634	189	189	189	
22	-446.3	-238	-1856	518	-2759	1705	1631	184	184	184	
23	-450.3	-239	-1889	534	-2784	1721	1648	187	187	187	
24	-449.9	-241	-1970	535	-2787	1719	1645	185	185	185	
25	-458.3	-246	-1999	575	-2807	1751	1673	192	192	192	
26	-461.9	-253	-2009	607	-2827	1768	1667	192	192	192	
27	-463.5	-266	-2008	655	-3009	1770	1688	199	199	199	
28	-459.5	-284	-1978	684	-3051	1773	1693	183	183	183	
29	-463.7	-287	-1980	704	-3081	1789	1708	185	185	185	
30	-380.0	-205	-1218	335	-1886	1035	1030	85	85	85	
AXIAL FORCE (kN)		-82.5		-82.0		63.2		66.1			

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P5-B

TEST PANEL A (REAR DIAGONAL TOP HALF BUCKLED INSIDE, AND HORIZONTAL BUCKLED OUTSIDE INPLANE ND OUTPLANE)

No.	Load (kN)	Test Serial No. 34													
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-40.2	0	10	11	12	13	14	15	16	17	18	19	20		
2	-88.7	20	-163	-14	-124	129	22	144	144	144	144	144	144	144	
3	-98.6	41	-299	-25	-222	232	41	260	260	260	260	260	260	260	
4	-133.5	68	-449	-38	-327	343	63	303	303	303	303	303	303	303	
5	-228.7	106	-833	-49	-453	475	91	529	529	529	529	529	529	529	
6	-261.2	252	-1197	-86	-610	646	182	940	940	940	940	940	940	940	
7	-293.6	323	-1426	-105	-840	887	222	1093	1093	1093	1093	1093	1093	1093	
8	-323.5	402	-1882	-128	-1054	1121	265	1236	1236	1236	1236	1236	1236	1236	
9	-354.4	487	-1927	-163	-1138	1248	308	1379	1379	1379	1379	1379	1379	1379	
10	-365.9	635	-2271	-212	-1210	1379	360	1524	1524	1524	1524	1524	1524	1524	
11	-377.0	863	-2415	-233	-1237	1430	360	1579	1579	1579	1579	1579	1579	1579	
12	-399.6	789	-2569	-256	-1255	1476	400	1630	1630	1630	1630	1630	1630	1630	
13	-397.3	844	-2655	-302	-1253	1536	429	1697	1697	1697	1697	1697	1697	1697	
14	-408.3	1188	-3000	-324	-1257	1566	445	1733	1733	1733	1733	1733	1733	1733	
15	-422.8	1338	-3466	-411	-1191	1626	481	1796	1796	1796	1796	1796	1796	1796	
16	-429.8	1477	-3602	-465	-1191	1690	512	1862	1862	1862	1862	1862	1862	1862	
17	-433.8	1771	-4231	-512	-1148	1713	527	1864	1864	1864	1864	1864	1864	1864	
18	-440.3	1825	-4631	-607	-1047	1744	646	1928	1928	1928	1928	1928	1928	1928	
19	-442.4	2105	-5256	-657	-1028	1787	575	1960	1960	1960	1960	1960	1960	1960	
20	-445.6	2222	-5655	-706	-975	1803	587	1872	1872	1872	1872	1872	1872	1872	
21	-449.1	2432	-5998	-739	-951	1820	599	1969	1969	1969	1969	1969	1969	1969	
22	-448.3	2749	-6378	-791	-900	1836	613	2008	2008	2008	2008	2008	2008	2008	
23	-450.3	2831	-6997	-854	-807	1839	621	2004	2004	2004	2004	2004	2004	2004	
24	-449.9	2970	-7152	-877	-787	1857	632	2022	2022	2022	2022	2022	2022	2022	
25	-458.3	3202	-7226	-885	-787	1857	632	2022	2022	2022	2022	2022	2022	2022	
26	-461.9	3664	-7836	-982	-727	1898	658	2080	2080	2080	2080	2080	2080	2080	
27	-483.5	4319	-8682	-1050	-633	1919	678	2079	2079	2079	2079	2079	2079	2079	
28	-459.5	4807	-9665	-1165	-499	1940	704	2098	2098	2098	2098	2098	2098	2098	
29	-483.7	5036	-10862	-1251	-369	1942	721	2092	2092	2092	2092	2092	2092	2092	
30	-390.0	8620	-11108	-1285	-348	1983	732	2111	2111	2111	2111	2111	2111	2111	
			-18682	-1956	452	2083	945	2221	2221	2221	2221	2221	2221	2221	
AXIAL FORCE (kN)		-88.2		-64.6		89.6		92.7							

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN PS-B

TEST PANEL D (FRONT DIAGONAL NOT BUCKLED)

No.	Load (kN)	Test Serial No. 34														
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	
1	-40.2	1	-173	-55	15	170	14	174								
2	-66.7	-43	-82	-287	26	287	23	281								
3	-88.0	-60	-408	-45	45	390	37	365								
4	-133.5	-81	-546	-171	71	525	56	524								
5	-228.7	-136	-805	-282	131	832	97	830								
6	-281.2	-160	-1036	-322	151	840	113	854								
7	-293.6	-183	-1156	-381	175	1042	128	1053								
8	-323.5	-207	-1289	-402	186	1143	142	1170								
9	-354.4	-234	-1390	-446	225	1245	159	1261								
10	-385.9	-244	-1423	-481	236	1285	166	1328								
11	-377.0	-255	-1481	-477	247	1324	172	1367								
12	-389.6	-271	-1501	-504	257	1369	178	1418								
13	-397.3	-279	-1529	-517	265	1385	181	1446								
14	-409.3	-302	-1549	-558	283	1446	188	1498								
15	-427.6	-319	-1579	-581	300	1498	202	1564								
16	-429.8	-329	-1577	-611	309	1519	209	1587								
17	-433.9	-364	-1586	-640	318	1549	208	1625								
18	-440.3	-387	-1581	-689	329	1578	216	1656								
19	-442.4	-378	-1545	-712	332	1598	219	1671								
20	-445.6	-387	-1540	-732	338	1603	222	1689								
21	-449.1	-401	-1524	-781	343	1618	224	1704								
22	-448.3	-421	-1471	-789	345	1620	225	1709								
23	-459.3	-429	-1470	-814	349	1633	227	1725								
24	-449.9	-430	-1484	-818	348	1631	228	1722								
25	-459.3	-453	-1449	-866	342	1606	237	1758								
26	-411.9	-482	-1404	-823	349	1662	242	1781								
27	-483.5	-545	-1292	-809	340	1605	240	1805								
28	-496.5	-609	-1172	-1109	344	1707	252	1812								
29	-483.7	-621	-1181	-1135	390	1721	257	1827								
30	-380.0	-507	-664	-814	92	1189	92	1219								
AXIAL FORCE (kN)		8.8			-85.0			72.3			74.1					

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P5-B

TEST PANEL D (REAR DIAGONAL NOT BUCKLED)

No.	Load (kN)	Test Serial No. 34									
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-40.2	-186	22	-143	12	13	14	16	16		
2	-68.7	-327	31	-257	-28	-9	141	11	116		
3	-88.6	-477	41	-378	-47	-6	245	24	202		
4	-133.5	-656	52	-526	-68	-6	351	37	285		
5	-226.7	-1207	116	-960	-82	-2	476	56	402		
6	-261.2	-1434	153	-1126	-147	22	616	104	704		
7	-283.6	-1666	205	-1268	-164	33	944	124	816		
8	-323.5	-1916	267	-1433	-178	42	1066	140	929		
9	-354.4	-2162	351	-1574	-188	49	1183	155	1034		
10	-365.9	-2301	396	-1628	-220	57	1303	171	1143		
11	-377.0	-2411	429	-1675	-241	61	1350	177	1168		
12	-389.6	-2562	500	-1722	-264	64	1385	183	1226		
13	-397.3	-2673	540	-1751	-274	68	1450	189	1276		
14	-406.3	-2845	673	-1796	-324	69	1481	194	1304		
15	-422.6	-3142	766	-1804	-324	74	1539	201	1352		
16	-426.6	-3278	838	-1797	-350	75	1598	210	1404		
17	-433.8	-3491	1007	-1766	-376	77	1619	214	1424		
18	-440.3	-3776	1097	-1764	-437	77	1669	222	1468		
19	-442.4	-3853	1186	-1737	-466	80	1691	227	1485		
20	-445.6	-4074	1252	-1731	-489	80	1705	231	1497		
21	-449.1	-4281	1356	-1707	-519	84	1721	234	1512		
22	-446.3	-4565	1512	-1633	-555	84	1736	238	1527		
23	-450.3	-4879	1560	-1634	-606	83	1739	240	1524		
24	-449.8	-4715	1579	-1634	-620	86	1756	243	1540		
25	-458.3	-5023	1740	-1626	-626	86	1756	243	1540		
26	-461.8	-5464	1969	-1611	-677	82	1783	254	1573		
27	-463.5	-6332	2418	-1550	-743	95	1815	281	1591		
28	-459.5	-7066	2620	-1415	-861	100	1836	273	1609		
29	-463.7	-7253	2810	-1283	-955	103	1842	278	1609		
30	-380.0	-10645	5126	-1275	-960	108	1862	285	1626		
				-672	-1531	237	2050	445	1779		
AXIAL FORCE (kN)		-88.8	-85.3	67.4	65.2						

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P6-A

TEST PANEL D (FRONT DIAGONAL BUCKLED INSIDE TOTALLY)

No.	Test Serial No. 28									
	Loos (kN)	Strain	1	2	3	4	5	6	7	8
1	-34.7		-237	-68	-233	-73	-49	114		
2	-64.0		-445	-116	-434	-126	-99	207		
3	-86.9		-619	-143	-588	-160	-142	287		
4	-110.6		-806	-161	-776	-189	-182	374		
5	-134.6		-1013	-188	-984	-209	-250	468		
6	-159.0		-1233	-153	-1159	-220	-316	569		
7	-183.2		-1481	-116	-1380	-276	-398	688		
8	-202.5		-1711	-64	-1522	-230	-480	802		
9	-221.1		-1973	22	-1674	-232	-575	936		
10	-230.7		-2113	74	-1756	-227	-622	1004		
11	-239.0		-2280	157	-1808	-242	-663	1063		
12	-243.3		-2377	200	-1836	-245	-711	1132		
13	-251.0		-2544	288	-1885	-255	-761	1207		
14	-254.9		-2629	334	-1888	-266	-787	1246		
15	-252.7		-2780	439	-1809	-324	-812	1277		
16	-260.7		-2807	507	-1841	-331	-842	1326		
17	-263.1		-3216	705	-1783	-405	-892	1403		
18	-267.9		-3664	1183	-1873	-411	-871	1433		
19	-246.3		-4028	1256	-1277	-798	-1009	1519		
20	-244.6		-4132	1327	-1220	-864	-1036	1551		
21	-241.8		-4240	1406	-1152	-940	-1068	1588		
22	-237.9		-4745	1699	-1026	-1080	-1129	1654		
23	-236.0		-5303	1971	-936	-1189	-1159	1689		
24	-234.9		-6018	2374	-805	-1312	-1186	1720		
25	-233.6		-6808	2783	-699	-1434	-1213	1756		
26	-232.3		-8055	3379	-524	-1594	-1234	1785		
27	-228.9		-10564	4785	-186	-1895	-1274	1841		
28	-224.7		-12762	6100	68	-2144	-1306	1885		
29	-221.9		-14124	7001	292	-2307	-1338	1921		
30	-221.9		-18297	8502	639	-2613	-1369	1964		
AXIAL FORCE (kN)			-91.3		-84.8		21.4		24.0	

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P6-A

TEST PANEL D (REAR DIAGONAL BUCKLED OUTSIDE TOTALLY)

No.	TEST SERIAL NO. 28														
	Lead (kN)	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-34.7	-207	-47	-22	-22	-22	-22	-22	-22	-22	-22	-22	-22	-22	-22
2	-64.0	-401	-70	-25	-25	-25	-25	-25	-25	-25	-25	-25	-25	-25	-25
3	-86.9	-562	-96	-18	-18	-18	-18	-18	-18	-18	-18	-18	-18	-18	-18
4	-110.6	-739	-104	8	8	8	8	8	8	8	8	8	8	8	8
5	-134.8	-935	-90	49	49	49	49	49	49	49	49	49	49	49	49
6	-159.0	-1142	-70	115	115	115	115	115	115	115	115	115	115	115	115
7	-183.2	-1367	-44	202	202	202	202	202	202	202	202	202	202	202	202
8	-202.5	-1556	-5	288	288	288	288	288	288	288	288	288	288	288	288
9	-221.1	-1748	41	366	366	366	366	366	366	366	366	366	366	366	366
10	-230.7	-1850	73	450	450	450	450	450	450	450	450	450	450	450	450
11	-239.0	-1933	97	510	510	510	510	510	510	510	510	510	510	510	510
12	-243.3	-1987	116	543	543	543	543	543	543	543	543	543	543	543	543
13	-251.0	-2073	149	610	610	610	610	610	610	610	610	610	610	610	610
14	-254.6	-2114	186	645	645	645	645	645	645	645	645	645	645	645	645
15	-262.7	-2085	156	642	642	642	642	642	642	642	642	642	642	642	642
16	-268.3	-2164	188	682	682	682	682	682	682	682	682	682	682	682	682
17	-283.1	-2231	220	762	762	762	762	762	762	762	762	762	762	762	762
18	-287.6	-2267	269	812	812	812	812	812	812	812	812	812	812	812	812
19	-246.3	-2527	563	1136	1136	1136	1136	1136	1136	1136	1136	1136	1136	1136	1136
20	-244.6	-2602	671	1233	1233	1233	1233	1233	1233	1233	1233	1233	1233	1233	1233
21	-241.8	-2607	751	1320	1320	1320	1320	1320	1320	1320	1320	1320	1320	1320	1320
22	-237.6	-2581	834	1446	1446	1446	1446	1446	1446	1446	1446	1446	1446	1446	1446
23	-236.0	-2550	842	1520	1520	1520	1520	1520	1520	1520	1520	1520	1520	1520	1520
24	-234.8	-2487	828	1605	1605	1605	1605	1605	1605	1605	1605	1605	1605	1605	1605
25	-233.6	-2422	766	1713	1713	1713	1713	1713	1713	1713	1713	1713	1713	1713	1713
26	-232.3	-2310	711	1850	1850	1850	1850	1850	1850	1850	1850	1850	1850	1850	1850
27	-228.9	-2103	554	2236	2236	2236	2236	2236	2236	2236	2236	2236	2236	2236	2236
28	-227.2	-1854	441	2660	2660	2660	2660	2660	2660	2660	2660	2660	2660	2660	2660
29	-224.7	-1691	388	3007	3007	3007	3007	3007	3007	3007	3007	3007	3007	3007	3007
30	-221.9	-1776	307	3908	3908	3908	3908	3908	3908	3908	3908	3908	3908	3908	3908
AXIAL FORCE (kN)			-79.8		-82.5		8.7		9.4						

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P6-B

TEST PANEL D (FRONT DIAGONAL NOT BUCKLED)

No.	Load (kN)	Test Serial No. 15							
		1	2	3	4	5	6	7	8
1	-26.6	-61	-176	-49	-162	66	-6		
2	-45.5	-88	-297	-77	-308	106	-14	63	-14
3	-69.6	-135	-449	-104	-464	159	-20	113	-27
4	-83.6	-156	-549	-117	-567	191	-40	165	-48
5	-101.1	-177	-688	-127	-683	233	-54	200	-64
6	-119.5	-193	-800	-131	-830	277	-73	243	-83
7	-136.3	-203	-937	-130	-975	326	-96	291	-106
8	-156.7	-207	-1077	-120	-1124	379	-121	343	-135
9	-169.0	-208	-1175	-106	-1226	414	-136	368	-167
10	-183.1	-204	-1282	-91	-1354	459	-187	437	-186
11	-203.2	-183	-1455	-55	-1537	523	-197	494	-216
12	-210.9	-186	-1520	-37	-1612	547	-210	553	-259
13	-221.9	-181	-1611	-8	-1718	564	-230	583	-277
14	-229.9	-179	-1687	13	-1783	606	-241	622	-302
15	-237.4	-182	-1746	42	-1865	641	-262	648	-319
16	-243.1	-151	-1788	65	-1946	666	-276	687	-343
17	-250.5	-143	-1855	86	-2045	699	-283	714	-369
18	-261.9	-126	-1947	144	-2178	769	-337	754	-383
19	-267.9	-120	-1991	170	-2246	809	-363	829	-431
20	-272.1	-118	-2015	186	-2299	849	-360	872	-457
21	-278.4	-103	-2056	217	-2373	897	-421	914	-485
22	-283.9	-66	-2118	246	-2449	964	-487	965	-519
23	-285.7	-71	-2140	261	-2472	987	-479	1023	-549
24	-287.1	-60	-2180	278	-2503	997	-492	1057	-577
25	-282.1	-56	-2154	317	-2538	1010	-486	1083	-591
26	-269.7	-102	-1739	167	-2039	1018	-371	1101	-597
27	-247.7	-112	-1717	182	-2043	795	-357	853	-450
28	-215.2	-81	-1467	185	-1776	779	-257	846	-440
29	-208.6	-87	-1409	217	-1796	627	-218	691	-335
30	-196.9	-72	-1340	248	-1705	583	-174	659	-304
						530		613	-284
AXIAL FORCE (kN)	-65.5		-84.9		19.3			18.1	

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN PG-B

TEST PANEL D (REAR DIAGONAL TOP HALF BUCKLED INSIDE)

No.	Load (kN)	Test Serial No. 15																																																							
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain																																										
1	-26.6	-79	-128	-84	-119	-26	14	13	12	11	10	9	8	7	6	5	4	3	2	1																																					
2	-45.5	-131	-227	-143	-227	-53	-20	26	-213	-53	-362	-453	-311	-410	-362	-311	-274	-233	-195	-131																																					
3	-69.6	-195	-362	-218	-362	-95	-55	95	-338	-95	-638	-523	-362	-452	-362	-311	-274	-233	-195	-131																																					
4	-83.6	-233	-453	-281	-453	-125	-80	125	-422	-125	-685	-523	-362	-452	-362	-311	-274	-233	-195	-131																																					
5	-101.1	-274	-523	-311	-523	-161	-95	161	-523	-161	-751	-523	-362	-452	-362	-311	-274	-233	-195	-131																																					
6	-119.5	-315	-563	-311	-563	-202	-111	202	-563	-202	-751	-563	-362	-452	-362	-311	-274	-233	-195	-131																																					
7	-136.3	-350	-604	-311	-604	-240	-118	240	-604	-240	-751	-604	-362	-452	-362	-311	-274	-233	-195	-131																																					
8	-156.7	-381	-649	-311	-649	-281	-138	281	-649	-281	-751	-649	-362	-452	-362	-311	-274	-233	-195	-131																																					
9	-199.0	-398	-698	-311	-698	-332	-143	332	-698	-332	-751	-698	-362	-452	-362	-311	-274	-233	-195	-131																																					
10	-183.1	-413	-704	-311	-704	-332	-143	332	-704	-332	-751	-704	-362	-452	-362	-311	-274	-233	-195	-131																																					
11	-203.2	-430	-730	-311	-730	-332	-143	332	-730	-332	-751	-730	-362	-452	-362	-311	-274	-233	-195	-131																																					
12	-210.9	-434	-738	-311	-738	-332	-143	332	-738	-332	-751	-738	-362	-452	-362	-311	-274	-233	-195	-131																																					
13	-221.9	-438	-746	-311	-746	-332	-143	332	-746	-332	-751	-746	-362	-452	-362	-311	-274	-233	-195	-131																																					
14	-228.9	-444	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
15	-237.4	-442	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
16	-243.1	-439	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
17	-250.5	-445	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
18	-261.9	-403	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
19	-267.9	-366	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
20	-272.1	-307	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
21	-278.4	-205	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
22	-283.9	89	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
23	-285.7	555	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
24	-287.1	1091	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
25	-282.1	4838	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
26	-249.7	10971	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
27	-247.7	12240	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
28	-215.2	20189	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
29	-208.6	23157	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
30	-199.9	26381	-754	-311	-754	-332	-143	332	-754	-332	-751	-754	-362	-452	-362	-311	-274	-233	-195	-131																																					
AXIAL FORCE (kN)		-101.5														-98.5														15.5														12.7													

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P7-A

TEST PANEL C (PANEL D&E LEGS, PANEL D TOP LEG BUCKLED OUTPLANE)

No.	Load (kN)	Test Serial No. 16														
		17	18	19	20	21	22	23	24							
1	-28.9	-262	-339	-353	-265	-300	-248	-219	-322							
2	-43.7	-273	-445	-407	-274	-403	-332	-292	-434							
3	-71.8	-381	-610	-469	-395	-542	-440	-399	-629							
4	-80.2	-451	-780	-527	-481	-645	-548	-465	-728							
5	-108.1	-543	-932	-627	-559	-804	-644	-564	-864							
6	-128.4	-633	-1101	-707	-647	-917	-707	-644	-964							
7	-148.0	-721	-1278	-802	-751	-1042	-802	-707	-1042							
8	-187.3	-809	-1454	-845	-845	-1205	-845	-707	-1205							
9	-188.8	-898	-1642	-939	-939	-1353	-939	-802	-1353							
10	-189.2	-931	-1798	-996	-996	-1453	-996	-802	-1453							
11	-205.8	-953	-1937	-1025	-1025	-1517	-1025	-802	-1517							
12	-213.0	-975	-1978	-1048	-1048	-1563	-1048	-802	-1563							
13	-228.8	-982	-2027	-1062	-1062	-1608	-1062	-802	-1608							
14	-254.2	-1004	-2099	-1103	-1103	-1668	-1103	-802	-1668							
15	-228.0	-1021	-2080	-1120	-1120	-1688	-1120	-802	-1688							
16	-231.3	-1044	-2111	-1128	-1128	-1708	-1128	-802	-1708							
17	-233.1	-1068	-2071	-1153	-1153	-1745	-1153	-802	-1745							
18	-238.7	-1089	-2099	-1179	-1179	-1779	-1179	-802	-1779							
19	-183.8	-881	-2020	-1488	-1488	-1681	-1488	-802	-1681							
20	-184.6	-885	-2056	-1508	-1508	-1711	-1508	-802	-1711							
21	-180.2	-948	-1878	-1631	-1631	-1777	-1631	-802	-1777							
22	-184.1	-925	-1820	-1727	-1727	-1806	-1727	-802	-1806							
23	-148.2	-688	-1871	-1682	-1682	-1751	-1682	-802	-1751							
24	-144.1	-688	-1817	-1659	-1659	-1725	-1659	-802	-1725							
25	-128.0	-431	-1781	-1811	-1720	-1845	-1720	-802	-1845							
AXIAL FORCE (kN)		-369.6		-387.3		-371.7		-357.2		-371.7		-357.2		-371.7		-357.2
		343.2		183.5		68.4		68.4		68.4		68.4		68.4		68.4

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P8-B

TEST PANEL D (FRONT DIAGONAL TOP HALF BUCKLED INSIDE)

No.	Load (kN)	Test Serial No. 19									
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-34.8	-288	-56	-273	-49	-151	-53	145			
2	-54.0	-447	-76	-426	-80	-233	-80	226			
3	-74.1	-617	-94	-597	-103	-318	-132	308			
4	-94.6	-788	-100	-757	-113	-409	-161	368			
5	-115.3	-968	-92	-937	-112	-505	-236	481			
6	-136.1	-1169	-67	-1127	-99	-613	-300	595			
7	-156.6	-1514	-17	-1323	-76	-732	-347	710			
8	-166.6	-1825	46	-1408	-94	-835	-428	808			
9	-175.4	-2084	80	-1423	-122	-968	-465	967			
10	-181.2	-2189	127	-1434	-151	-1066	-508	935			
11	-183.4	-2262	159	-1417	-182	-1003	-532	973			
12	-186.1	-2374	211	-1407	-208	-1045	-545	1017			
13	-188.2	-2506	261	-1386	-254	-1088	-571	1064			
14	-188.2	-2668	344	-1273	-348	-1167	-607	1131			
15	-192.0	-2760	451	-1108	-504	-1211	-648	1210			
16	-192.7	-2878	603	-776	-769	-1287	-690	1287			
17	-191.9	-3261	932	-351	-1131	-1358	-698	1435			
18	-191.6	-3474	1086	-209	-1242	-1368	-687	1462			
19	-192.0	-3692	1349	43	-1444	-1383	-681	1503			
20	-192.3	-4063	1702	277	-1630	-1391	-626	1534			
21	-193.0	-5233	2201	431	-1756	-1402	-605	1556			
22	-192.9	-7414	3305	612	-1902	-1415	-585	1577			
23	-192.3	-8757	4564	728	-1987	-1419	-587	1588			
24	-192.8	-12069	5954	826	-2076	-1434	-560	1603			
25	-191.9	-15691	8411	970	-2183	-1440	-543	1615			
26	-191.5	-18254	9811	1066	-2260	-1448	-534	1626			
27	-190.1	-20246	11294	1185	-2461	-1457	-528	1639			
28	-196.0	-21199	11946	1220	-2510	-1455	-520	1641			
29	-198.1	-22086	12558	1279	-2565	-1457	-521	1644			
30	-115.0	-20980	11826	1536	-2229	-1387	-775	1411			
31	-113.8	-20656	11804	1517	-2205	-1376	-775	1403			
AXIAL FORCE (kN)				-62.9		21.0		19.6			

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN P8-B

TEST PANEL D (REAR DIAGONAL NOT BUCKLED)

No.	Load (kN)	Test Serial No. 19																			
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain					
1	-34.6	-105	-47	-207	10	13	14	15	16	17	18	19	20	21	22	23					
2	-54.0	-304	-103	-342	29	28	-45	-36	-1												
3	-74.1	-437	-137	-408	59	58	-114	-92	35												
4	-94.6	-565	-167	-477	87	86	-156	-131	66												
5	-115.3	-748	-198	-577	115	114	-214	-179	116												
6	-136.1	-926	-228	-677	143	142	-281	-240	175												
7	-156.6	-1122	-288	-813	171	170	-363	-314	252												
8	-169.6	-1233	-343	-941	200	199	-450	-394	335												
9	-175.4	-1285	-398	-1069	228	227	-504	-442	387												
10	-181.2	-1344	-453	-1197	257	256	-564	-496	447												
11	-183.4	-1367	-467	-1220	285	284	-600	-532	482												
12	-186.1	-1392	-481	-1243	313	312	-641	-570	524												
13	-188.2	-1415	-495	-1266	341	340	-681	-608	564												
14	-189.2	-1442	-509	-1289	369	368	-733	-666	618												
15	-189.5	-1465	-523	-1312	397	396	-770	-708	669												
16	-192.7	-1498	-537	-1335	425	424	-822	-736	712												
17	-191.9	-1489	-529	-1321	453	452	-847	-783	766												
18	-191.6	-1482	-521	-1318	481	480	-853	-771	759												
19	-192.0	-1513	-535	-1341	509	508	-909	-800	795												
20	-192.3	-1544	-549	-1364	537	536	-955	-834	841												
21	-193.0	-1567	-563	-1387	565	564	-1016	-877	901												
22	-192.9	-1605	-577	-1410	593	592	-1078	-930	979												
23	-192.3	-1610	-569	-1403	621	620	-1090	-965	1032												
24	-192.8	-1642	-583	-1426	649	648	-1113	-1000	1100												
25	-191.9	-1615	-575	-1409	677	676	-1168	-1051	1180												
26	-191.5	-1688	-609	-1482	705	704	-1203	-1070	1237												
27	-190.1	-1979	-643	-1755	733	732	-1205	-1068	1262												
28	-198.0	-2081	-677	-1828	761	760	-1194	-1058	1319												
29	-198.1	-2130	-669	-1821	789	788	-1172	-1045	1329												
30	-115.0	-2987	1524	3	1283	1237	-1237	-1048	1329												
31	-113.8	-2834	1487	-5	1277	1223	-1223	-1036	1309												
AXIAL FORCE (kN)		-62					-68.5					-1.9					-1.5				

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN PB-B

TEST PANEL D (PANEL D&E TOP LEGS NOT BUCKLED)

No.	Load (kN)	Strain				Test Serial No. 19
		1	2	3	4	
1	-24.0	-182	-238	-182	-238	-173
2	-46.0	-289	-347	-289	-347	-289
3	-74.1	-448	-506	-448	-506	-364
4	-84.8	-709	-767	-709	-767	-487
5	-118.3	-968	-1026	-968	-1026	-649
6	-128.1	-1024	-1082	-1024	-1082	-649
7	-182.0	-1284	-1340	-1284	-1340	-810
8	-189.6	-1338	-1394	-1338	-1394	-810
9	-178.4	-1381	-1437	-1381	-1437	-810
10	-181.2	-1438	-1494	-1438	-1494	-810
11	-183.4	-1491	-1547	-1491	-1547	-810
12	-186.1	-1548	-1604	-1548	-1604	-810
13	-188.2	-1592	-1657	-1592	-1657	-810
14	-189.2	-1651	-1716	-1651	-1716	-810
15	-192.0	-1708	-1773	-1708	-1773	-810
16	-192.7	-1767	-1832	-1767	-1832	-810
17	-191.8	-1827	-1891	-1827	-1891	-810
18	-191.6	-1884	-1948	-1884	-1948	-810
19	-192.0	-1940	-2006	-1940	-2006	-810
20	-192.3	-1991	-2057	-1991	-2057	-810
21	-193.0	-2042	-2108	-2042	-2108	-810
22	-192.0	-2092	-2158	-2092	-2158	-810
23	-192.3	-2140	-2207	-2140	-2207	-810
24	-192.0	-2188	-2256	-2188	-2256	-810
25	-191.0	-2235	-2304	-2235	-2304	-810
26	-191.0	-2281	-2352	-2281	-2352	-810
27	-190.1	-2326	-2400	-2326	-2400	-810
28	-188.0	-2370	-2448	-2370	-2448	-810
29	-188.1	-2413	-2496	-2413	-2496	-810
30	-183.0	-2455	-2544	-2455	-2544	-810
31	-183.0	-2496	-2592	-2496	-2592	-810
AXIAL FORCE (kN)		-263.6				-262.4

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S2-A

PANEL A (FRONT DIAGONAL BUCKLED WITH BOTTOM HALF OUTSIDE AND TOP INSIDE)

No.	Load (kN)	Test Serial No. 4															
		Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12				
1	10	-138	20	-94	-54	-37	-102	140	-130	-124	-58	-30	-127				
2	20	-281	16	-187	-114	-62	-206	-12	-265	-251	-76	-44	-254				
3	30	-418	22	-278	-176	-90	-322	-18	-392	-368	-84	-80	-366				
4	40	-582	35	-386	-237	-112	-458	36	-565	-565	-58	-110	-487				
5	50	-754	55	-496	-300	-130	-584	92	-765	-770	6	-142	-571				
6	60	-890	68	-574	-352	-148	-688	190	-966	-992	91	-174	-854				
7	70	-1040	71	-670	-406	-190	-790	432	-1284	-1280	242	-210	-730				
8	75	-1076	57	-696	-421	-222	-814	466	-1438	-1417	328	-232	-752				
9	77.5	-1090	46	-710	-431	-240	-814	502	-1496	-1480	366	-242	-758				
10	80	-1107	22	-734	-450	-282	-814	586	-1614	-1508	442	-258	-770				
11	82.5	-1110	10	-740	-458	-306	-808	646	-1690	-1666	493	-274	-772				
12	85	-1097	-42	-752	-476	-370	-782	776	-1846	-1814	602	-292	-777				
13	87.5	-1068	-82	-758	-490	-430	-742	905	-1980	-1950	708	-310	-774				
14	90	-1046	-146	-758	-492	-526	-680	1064	-2164	-2106	820	-326	-782				
15	92.5	-930	-290	-726	-502	-762	-458	1490	-2796	-2552	1152	-358	-735				
16	95	-512	-657	-506	-466	-1316	122	2250	-4324	-3360	1824	-432	-712				
17	83.3	154	-640	-620	291	-3326	2260	3932	-5982	-4342	3120	-632	-226				
AXIAL FORCE (kN)		-14.8		-12.3		-15.1		-17.7		-15.9		-14.5					
MOMENT (N.m)		2.9		0.8		28.9		93.8		86.9		5.6					

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S2-A

PANEL A (REAR DIAGONAL BUCKLED WITH TOP HALF OUTSIDE AND BOTTOM INSIDE, FIRST) Test Serial No. 4

No.	Load kN	17	18	19	20	21	22	23	24	25	26	27	28
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	10	-50	-22	-56	-77	-38	-82	-48	-90	-148	8	-54	-89
2	20	-88	-47	-140	-166	-83	-188	-88	-200	-340	16	-117	-202
3	30	-128	-67	-192	-242	-72	-287	-140	-300	-504	22	-175	-305
4	40	-170	-68	-271	-338	-62	-434	-184	-404	-678	38	-230	-400
5	50	-207	-38	-333	-436	-12	-588	-238	-498	-854	48	-282	-500
6	60	-254	28	-380	-538	94	-774	-310	-588	-1000	44	-350	-587
7	70	-324	208	-406	-676	308	-1054	-484	-558	-1108	-20	-400	-694
8	75	-354	336	-402	-738	470	-1207	-582	-486	-1102	-88	-414	-740
9	77.5	-358	394	-398	-770	554	-1280	-648	-444	-1088	-124	-410	-751
10	80	-378	538	-382	-828	728	-1428	-778	-340	-1038	-208	-408	-784
11	82.5	-386	638	-360	-860	854	-1528	-870	-259	-988	-274	-404	-810
12	85	-402	666	-328	-930	1174	-1756	-1082	-30	-864	-428	-380	-852
13	87.5	-458	1114	-298	-1008	1588	-2082	-1236	250	-690	-588	-337	-888
14	90	-497	1512	-284	-1122	2080	-2478	-1380	578	-472	-784	-285	-920
15	92.5	-527	2040	-314	-1318	2872	-3150	-1492	1000	-170	-1042	-150	-955
16	98	-644	2492	-336	-1488	3892	-3822	-1608	1388	80	-1210	-64	-972
17	53.3	-400	11916	-760	-2486	3850	-3202	-2326	3228	968	-1240	418	-844
	AXIAL FORCE (kN)	23.8		-22.7		0.6		-2.8		-14.3		-13	
	MOMENT (N.m)	-56.4		22.5		-111.6		60.2		25.9		18.5	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S2-A

PANEL A (FRONT AND REAR TENSION DIAGONAL)

Test Serial No. 4

No.	Load (kN)	Strain	13	Strain	14	Strain	15	Strain	16	Strain	19	Strain	20	Strain	29	Strain	30	Strain	31	Strain	32		
1	10	74	86	2	74	16	60	30	30	60	30	30	30	60	30	30	30	60	30	30	30	30	
2	20	158	174	4	168	4	168	4	168	4	168	4	168	4	168	4	168	4	168	4	168	4	168
3	30	236	260	6	244	6	244	6	244	6	244	6	244	6	244	6	244	6	244	6	244	6	244
4	40	332	368	8	344	8	344	8	344	8	344	8	344	8	344	8	344	8	344	8	344	8	344
5	50	438	474	12	418	12	418	12	418	12	418	12	418	12	418	12	418	12	418	12	418	12	418
6	60	528	578	20	504	20	504	20	504	20	504	20	504	20	504	20	504	20	504	20	504	20	504
7	70	646	694	34	600	34	600	34	600	34	600	34	600	34	600	34	600	34	600	34	600	34	600
8	75	680	744	44	640	44	640	44	640	44	640	44	640	44	640	44	640	44	640	44	640	44	640
9	77.5	710	764	48	658	48	658	48	658	48	658	48	658	48	658	48	658	48	658	48	658	48	658
10	80	742	796	56	696	56	696	56	696	56	696	56	696	56	696	56	696	56	696	56	696	56	696
11	82.5	780	818	60	730	60	730	60	730	60	730	60	730	60	730	60	730	60	730	60	730	60	730
12	85	792	858	70	732	70	732	70	732	70	732	70	732	70	732	70	732	70	732	70	732	70	732
13	87.5	816	880	80	752	80	752	80	752	80	752	80	752	80	752	80	752	80	752	80	752	80	752
14	90	848	914	92	782	92	782	92	782	92	782	92	782	92	782	92	782	92	782	92	782	92	782
15	92.5	884	960	118	824	118	824	118	824	118	824	118	824	118	824	118	824	118	824	118	824	118	824
16	96	928	1010	178	868	178	868	178	868	178	868	178	868	178	868	178	868	178	868	178	868	178	868
17	53.3	780	832	10	716	10	716	10	716	10	716	10	716	10	716	10	716	10	716	10	716	10	716
AXIAL FORCE (kN)		15.4	15.0	16.0	16.2	16.0	16.2	16.0	16.2	16.0	16.2	16.0	16.2	16.0	16.2	16.0	16.2	16.0	16.2	16.0	16.2	16.0	16.2
MOMENT (k.m)		-12.6	-16.7	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1	-9.5	-6.1

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S2-B

PANEL A (FRONT DIAGONAL)

Test Serial No. 8

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12
1	10	-202	60	-82	-60	10	-140	12	-147	-175	38	-40	-80
2	20	-410	128	-182	-128	10	-284	28	-304	-350	84	-83	-145
3	30	-616	190	-244	-188	25	-428	60	-470	-528	132	-123	-238
4	40	-800	254	-312	-254	35	-560	105	-644	-705	188	-162	-310
5	50	-990	318	-390	-328	54	-700	168	-838	-898	254	-205	-372
6	60	-1150	380	-420	-386	56	-820	252	-1040	-1086	330	-255	-432
7	70	-1293	388	-475	-474	44	-925	370	-1278	-1300	433	-300	-472
8	80	-1390	378	-494	-566	-18	-980	555	-1578	-1560	578	-360	-500
9	88	-1414	338	-498	-620	-82	-970	688	-1780	-1718	688	-400	-600
10	87.5	-1390	308	-498	-652	-144	-940	626	-1880	-1825	772	-420	-494
11	90	-1360	250	-490	-684	-244	-866	1038	-2075	-1960	885	-445	-488
12	91.25	-1258	154	-460	-700	-370	-745	1220	-2305	-2085	988	-462	-448
13	92.5	-1127	28	-434	-708	-580	-510	1600	-2745	-2203	1220	-480	-380
14	93.75	-810	-155	-348	-670	-854	-174	2030	-3180	-2388	1478	-505	-340
15	95	-450	-445	-268	-595	-1278	444	3080	-4190	-2880	2125	-475	-80
16	96.25	-157	-786	-297	-575	-1733	880	4025	-5236	-3290	2500	-520	-20
17	97.5												
18	54.54	640	-1166	-514	195	-5650	3660	6920	-7070	-4540	4350	-650	480
AXIAL FORCE (kN)		-13.6		-14.2		-13.3		-13.7		-13.0		-11.4	
MOMENT(N.m)		35.2		2.5		17.9		49.8		48.8		2	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S2-B

PANEL A (REAR DIAGONAL)

No.	Load (kN)	Test Serial No. 10											AXIAL FORCE (kN)	MOMENT (N.m)			
		17	18	19	20	21	22	23	24	25	26	27			28		
1	10	-101	-35	-52	-82	-6	-94	-50	-74	-130	-8	-27	-110				
2	20	-232	-60	-105	-180	-22	-208	-109	-165	-271	-17	-46	-226				
3	30	-372	-72	-151	-281	28	-326	-170	-264	-410	-25	-86	-336				
4	40	-516	-74	-189	-384	90	-455	-235	-353	-545	-38	-130	-436				
5	50	-690	-86	-222	-501	34	-615	-314	-428	-674	-56	-234	-536				
6	60	-883	-8	-236	-622	130	-804	-412	-466	-779	-90	-294	-620				
7	70	-1138	102	-224	-767	230	-1068	-575	-428	-841	-172	-370	-686				
8	80	-1517	334	-160	-949	507	-1496	-802	-218	-776	-376	-472	-716				
9	85	-1791	543	-91	-1054	1002	-1990	-1204	40	-628	-578	-526	-702				
10	87.5	-1989	718	-27	-1121	1317	-2377	-1445	271	-481	-744	-558	-682				
11	90	-2184	903	46	-1181	1640	-2869	-1894	833	-292	-932	-682	-844				
12	91.25	-2302	1029	4	-1216	1826	-3121	-1846	726	-150	-1087	-800	-612				
13	92.5	-2410	1150	140	-1248	2055	-3323	-1968	923	-14	-1194	-816	-576				
14	93.75	-2495	1233	167	-1269	2423	-3430	-2100	1088	100	-1295	-826	-550				
15	95	-2662	1397	223	-1325	2896	-3609	-2349	1393	319	-1493	-848	-498				
16	96.25	-2866	1570	252	-1377	2947	-3700	-2586	1665	525	-1652	-862	-444				
17	97.5	-2886	1672	284	-1403	3197	-3849	-3073	1970	721	-1879	-886	-404				
18	54.54	-2361	1593	-162	-456	4012	-4010	-3413	2676	2241	-3076	-994	156				
AXIAL FORCE (kN)		-16.0											-13.7	-14.7	-15.5	-15.5	
MOMENT (N.m)		60.9											24.7	81.4	44.8	12.9	1.2

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S2-B

PANEL A (FRONT AND REAR TENSION DIAGONAL)

No.	Load kN	Test Serial No. 8																				
		Strain	13	Strain	14	Strain	15	Strain	16	Strain	18	Strain	20	Strain	26	Strain	30	Strain	31	Strain	32	
1	10		104	-6		106	-12		56		60		26		68		136		212		28	
2	20		214	-6		222	-6		134		136		108		208		276		282		64	
3	30		324	6		334	6		348		358		147		276		348		358		98	
4	40		428	24		440	16		420		430		186		420		430		430		132	
5	50		538	50		554	42		574		596		226		574		596		596		164	
6	60		634	86		656	74		656		666		270		666		666		666		188	
7	70		732	128		756	114		756		768		326		768		768		768		234	
8	80		825	176		852	164		852		866		366		866		866		866		260	
9	88		874	202		898	192		898		914		366		914		914		914		314	
10	97.5		896	218		922	208		922		946		394		946		946		946		338	
11	90		922	234		946	228		946		960		428		960		960		960		366	
12	91.25		938	248		960	244		960		970		452		970		970		970		388	
13	92.5		952	266		970	264		970		978		478		978		978		978		408	
14	93.75		966	280		978	284		978		992		494		992		992		992		428	
15	95		982	350		1004	364		1004		1004		532		1004		1004		1004		462	
16	96.25		1006	418			442						570								500	
17	97.5																					
18	54.54		850	45		748	150		748		366		622		366		462		462		500	
AXIAL FORCE (kN)			13.6			13.8			13.8		13.8		13.8		13.8		13.4		13.4			
MOMENT (k.m)			-13.5			-14.2			-14.2		-4.7		-4.7		-6.5							

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN S2-C

TEST PANEL A (FRONT DIAGONAL BUCKLED WITH BOTTOM HALF OUTSIDE AND TOP HALF INSIDE)

No.	Load (kN)	Strain	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	Strain	6	7	Test Serial No. 23
1	-23.9		-6	-322	-51	-273		257	44																	7				
2	-30.6		5	-420	-66	-356		326	61																					
3	-37.3		21	-518	-73	-437		395	80																					
4	-43.9		42	-618	-74	-519		463	102																					
5	-54.2		84	-778	-80	-649		563	139																					
6	-60.8		119	-897	-82	-736		679	186																					
7	-67.4		159	-998	-82	-824		800	233																					
8	-70.9		190	-1054	-86	-869		924	289																					
9	-74.2		205	-1054	-86	-869		1041	330																					
10	-77.8		231	-1111	-89	-913		1123	388																					
11	-80.8		258	-1170	-93	-957		1234	449																					
12	-80.7		260	-1234	-96	-999		1297	509																					
13	-84.2		290	-1233	-96	-999		1373	569																					
14	-87.2		329	-1297	-92	-1041		1453	629																					
15	-90.4		377	-1371	-19	-1078		1482	689																					
16	-92.7		422	-1453	-20	-1108		1482	749																					
17	-82.4		408	-1482	-22	-1123		1482	809																					
18	-89.6		342	-1489	-18	-1116		1482	869																					
19	-53.2		729	-1410	14	-1113		1482	929																					
20	-53.6		757	-1766	1085	-1125		1482	989																					
21	-53.6		789	-1822	1222	-2096		1482	1049																					
22	-53.5		758	-1870	1373	-2096		1482	1109																					
23	-48.7		2069	-1850	1808	-2251		1482	1169																					
24	-47.1		-2160	1200	18477	-2444		1482	1229																					
				1305	18247	-22627																								
AXIAL FORCE (kN)			-13.9		-14.5																									16.6

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN S2-C

TEST PANEL A (REAR DIAGONAL FIRST BUCKLED WITH BOTTOM HALF OUTSIDE AND TOP HALF INSIDE)

No.	Strain		Strain		Strain		Strain		Strain		Strain		Strain		Strain							
	Load (kN)	0	9	10	11	12	13	14	15	16	17	18	19	20	21	22						
1	-23.6		-25	-30.4	.77	-240	13	14		15	16											
2	-30.6	-18	-368	-308	-.98	-319	297	64		205	46											
3	-37.3	-4	-506	-506	-1.19	-380	361	81		263	64											
4	-43.9	22	-622	-622	-1.61	-457	424	99		322	81											
5	-54.2	86	-822	-822	-1.86	-550	518	131		362	99											
6	-60.8	151	-1030	-1030	-2.06	-622	563	155		473	131											
7	-67.4	245	-1157	-1157	-2.32	-699	646	178		532	155											
8	-70.9	309	-1285	-1285	-2.57	-778	712	194		590	178											
9	-74.2	386	-1385	-1385	-2.87	-858	774	210		649	194											
10	-77.8	482	-1522	-1522	-3.17	-950	821	226		708	210											
11	-80.8	603	-1663	-1663	-3.55	-1028	885	242		769	226											
12	-80.7	598	-1878	-1878	-3.91	-1125	946	241		809	242											
13	-84.2	760	-1978	-1978	-4.25	-1225	1006	261		850	241											
14	-87.2	1022	-2191	-2191	-4.65	-1328	1064	282		890	261											
15	-89.4	1489	-2789	-2789	-5.12	-1435	1128	282		947	282											
16	-92.7	2353	-4057	-4057	-5.63	-1548	1178	312		1000	312											
17	-92.4	2437	-4187	-4187	-5.88	-1638	1178	312		1000	312											
18	-93.6	3334	-5965	-5965	-6.47	-1847	1178	312		1000	312											
19	-93.2	31309	-40347	-40347	-6.60	-2000	942	364		1000	350											
20	-93.9	31586	-40526	-40526	-6.60	-2000	633	43		1000	350											
21	-93.6	31844	-40847	-40847	-6.60	-2000	622	43		1000	350											
22	-93.5	32081	-40743	-40743	-6.60	-2000	614	64		1000	350											
23	-98.7	32252	-40813	-40813	-6.60	-2000	608	100		1000	350											
24	-47.1	32240	-40755	-40755	-6.60	-2000	795	129		1000	350											
							776	131		1000	350											
AXIAL FORCE (kN)															14.8		14.5		15.1		14.0	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S3-A

PANEL C (FRONT DIAGONAL)

No.	Load (kN)	Test Serial No. 7											
		Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12
1	10	-172	-28	-123	-92	-109	-77	-64	-138	-130	-74	-114	-86
2	20	-365	-40	-248	-186	-248	-145	-110	-298	-288	-128	-230	-180
3	30	-604	-25	-387	-290	-418	-194	-122	-504	-508	-130	-355	-270
4	40	-872	30	-813	-384	-813	-200	-75	-748	-774	-58	-470	-345
5	50	-1207	156	-634	-480	-848	-142	64	-1087	-1142	126	-574	-404
6	55	-1420	258	-898	-550	-988	-78	190	-1276	-1386	286	-828	-430
7	57.5	-1526	315	-724	-550	-1060	-36	264	-1388	-1515	378	-850	-440
8	60	-1643	372	-754	-570	-1128	5	336	-1492	-1536	487	-868	-447
9	62.5	-1770	435	-784	-590	-1200	58	420	-1608	-1770	570	-890	-452
10	65	-1877	498	-814	-608	-1272	110	500	-1718	-1895	669	-708	-460
11	67.5	-2025	580	-854	-630	-1328	180	594	-1843	-2040	784	-728	-470
12	70	-2144	632	-800	-664	-1116	230	644	-1938	-2127	852	-750	-484
13	72.5	-2460	766	-883	-718	-1043	360	745	-2078	-2280	1004	-788	-513
	83.1												
	45.8												
		72.5kN Front Diagonal Failure											
AXIAL FORCE (kN)		-26.4		-27.3		-11.1		-21.3		-19.8		-21.0	
MOMENT (N.m)		87.7		7.6		39.9		80.6		91.8		8.1	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S3-A

PANEL C (REAR DIAGONAL)

Test Serial No. 7

No.	Load kN	Strain	17	Strain	18	Strain	19	Strain	20	Strain	21	Strain	22	Strain	23	Strain	24	Strain	25	Strain	26	Strain	27	Strain	28
1	10	-62	-122	-80	-96	-78	-98	-88	-94	-112	-60	-108	-70												
2	20	-124	-202	-164	-212	-166	-212	-174	-194	-244	-120	-240	-138												
3	30	-180	-412	-250	-336	-250	-344	-276	-294	-376	-184	-382	-210												
4	40	-222	-552	-314	-464	-308	-478	-368	-388	-508	-234	-516	-270												
5	50	-272	-712	-348	-628	-332	-658	-458	-484	-658	-270	-660	-320												
6	55	-300	-800	-334	-748	-310	-794	-510	-538	-754	-276	-748	-354												
7	57.5	-312	-838	-308	-824	-282	-878	-534	-560	-808	-272	-790	-368												
8	60	-328	-872	-272	-912	-244	-970	-558	-586	-864	-284	-834	-378												
9	62.5	-348	-912	-212	-1028	-178	-1090	-584	-610	-928	-248	-878	-390												
10	65	-364	-942	-124	-1160	-88	-1230	-612	-632	-984	-218	-920	-402												
11	67.5	-386	-968	20	-1354	66	-1430	-640	-650	-1090	-176	-970	-418												
12	70	-424	-1000	160	-1530	202	-1616	-670	-680	-1168	-144	-1004	-434												
13	72.5	-472	-1032	410	-1776	486	-1932	-758	-824	-1248	-110	-1034	-450												
14	63.1	-540	-1048	1134	-2360	2314	-4016	-2592	1238	-144	-1088	-332	-360												
15	45.6	-520	-756	2190	-2824	3084	-4376	-2620	1694	-536	-212	-346	-402												
			63.1kN Rear Diagonal Failure																						
	AXIAL FORCE (kN)	-25.5		-19.1		-19.4		-20.7		-19.8		-19.1													
	MOMENT(N.m)	14.6		97.3		139		103.4		27.0		13.5													

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN S3-B

TEST PANEL A (FRONT DIAGONAL BUCKLED TOP HALF OUTSIDE AND BOTTOM INSIDE)

No.	Load (kN)	Test Serial No. 25							
		1	2	3	4	5	6	7	8
1	-24.6	32	-520	4	-500	233	-117	285	-157
2	-33.8	103	-747	66	-721	351	-176	422	-230
3	-43.1	208	-895	161	-863	482	-236	575	-308
4	-52.2	342	-1255	284	-1218	620	-297	732	-383
5	-58.0	445	-1434	376	-1387	710	-331	834	-424
6	-63.4	575	-1630	460	-1540	785	-359	902	-463
7	-68.1	672	-1871	497	-1808	838	-373	981	-479
8	-68.0	773	-1995	511	-1844	870	-383	1014	-482
9	-68.3	897	-2178	492	-1841	893	-389	1038	-486
10	-69.1	989	-2280	456	-1589	863	-388	1038	-500
11	-70.6	1117	-2435	434	-1585	920	-384	1088	-509
12	-71.9	1254	-2603	410	-1587	946	-401	1094	-518
13	-87.3	12851	-20828	-1371	555	908	-273	955	-382
14	-87.2	19608	-25698	-1574	801	905	-238	953	-334
15	-87.4	20181	-26512	-1863	914	912	-219	959	-318
AXIAL FORCE (kN)		-20.4		-18.4		8.1		8.6	

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN S3-B

TEST PANEL A (REAR DIAGONAL BUCKLED TOP HALF INSIDE AND BOTTOM HALF OUTSIDE)

No.	Test Serial No. 25														
	Lead (kip)	Strain	9	Strain	10	Strain	11	Strain	12	Strain	13	Strain	14	Strain	15
1	-24.6	-45	-414	-208	-842	-281	-74	130	113	18	18				
2	-33.8	-2	-842	-282	-378	-500	-201	202	-128	74	113				
3	-43.1	113	-837	-341	-500	-621	-201	303	-293	179	179				
4	-52.2	320	-1318	-388	-621	-684	-201	435	-359	281	281				
5	-58.0	533	-1815	-417	-684	-709	-422	535	-451	419	419				
6	-63.4	800	-1980	-472	-709	-695	-422	639	-451	535	535				
7	-68.1	872	-2188	-522	-709	-695	-422	694	-451	640	640				
8	-68.0	1128	-2327	-563	-695	-695	-422	694	-451	697	697				
9	-68.3	1288	-2717	-609	-695	-695	-422	735	-471	740	740				
10	-68.1	1485	-2878	-720	-695	-695	-422	787	-483	778	778				
11	-70.6	1808	-3282	-804	-695	-695	-422	708	-483	777	777				
12	-71.8	2273	-3933	-1087	-695	-695	-422	801	-483	820	820				
13	-87.3	4238	-7200	-1881	-1369	-1369	-422	831	-483	859	859				
14	-87.2	4953	-7744	-1858	-1858	-1858	-422	831	-483	840	840				
15	-87.4	4818	-7814	-1883	-1883	-1883	-422	824	-483	857	857				
								828	-287	877	877				
AXIAL FORCE (KN)		-20.1		-20.0		4.6		5.7							

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-A

PANEL A(FRONT DIAGONAL TOP HALF BUCKLED INSIDE AND BOTTOM OUTSIDE)

TEST SERIAL NO. 1

No.	Load(kN)	Strain	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18
1	10	-11	0	-20	-81	56	6	32	-28	-2	28	178	-424	178	-86	122	104	132	56	
2	20	-24	5	-39	-174	124	12	72	-58	-6	53	100	-166	132	-102	90	224	-204	-42	
3	30	-36	10	-50	-270	196	18	108	-92	-10	80	-40	-278	180	-110	114	266	-262	-48	
4	40	-48	16	-58	-362	250	22	144	-122	-12	112	-22	-256	226	-178	136	328	-432	-40	
5	50	-60	22	-60	-458	314	30	178	-234	-10	142	62	-368	266	-188	158	360	-630	-12	
6	60	-76	28	-58	-578	384	40	222	-194	-20	158	66	-320	574	-244	164	448	-808	84	
7	70	-90	34	-58	-686	468	50	264	-234	-24	182	14	-114	780	-280	184	530	-1024	150	
8	80	-104	40	-50	-796	544	62	310	-276	-32	208	56	-314	386	-332	200	564	-1146	242	
9	90	-120	46	-40	-916	622	76	358	-318	-36	230	48	-260	434	-334	228	624	-1414	368	
10	100	-138	54	-32	-1034	702	84	402	-364	-44	256	32	-288	508	-368	268	688	-1630	610	
11	110	-160	62	-22	-1144	778	118	458	-488	-50	284	20	-128	488	-438	288	768	-2044	1388	
12	87	-194	116	-140	-1090	830	84	484	-432	-204	22	506	-238	508	-260	234	618	4008		
13	81.25	-324	230	-2505	1684	862	-40	852	-602	-288	112	-210	208	778	-514	368	688	5034		
14	83.6	-128	58	-2262	1534	788	-134	584	-534	-214	110	-160	-122	568	-532	298	604	5044		
	FORCE kN				-18.7												16.9		-20.1	
	MOMENT (N.m)				32.2												-13.2		105.4	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-A

PANEL B(FRONT DIAGONAL BOTTOM HALF BUCKLED OUTSIDE AND TOP INSIDE)

TEST SERIAL NO. 1

No	Load kn	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14	Strain 15	Strain 16	Strain 17	Strain 18	
1	10	6	6	6	14	14	64	6	28	12	22	6	12	22	26	140	36	20		
2	20	-28	-34	-162	-16	70	172	0	48	18	30	18	9	6	42	212	76	-46		
3	30	-26	-56	-245	-32	114	246	0	66	34	40	40	20	6	60	316	106	-82		
4	40	-52	-76	-328	-32	158	324	-6	82	36	48	6	26	6	78	400	148	-110		
5	50	-64	-96	-410	-50	194	396	0	100	50	54	2	18	-6	96	482	180	-146		
6	60	-74	-110	-510	-60	242	484	-12	120	60	40	34	22	-6	110	570	226	-190		
7	70	-90	-128	-614	-70	288	570	-10	134	70	38	34	28	0	126	654	274	-230		
8	80	-100	-146	-724	-84	330	648	-16	150	68	30	56	24	4	146	728	304	-266		
9	90	-116	-156	-838	-98	382	736	-16	172	102	20	67	26	10	158	794	334	-374		
10	100	-134	-178	-944	-124	422	816	-16	188	110	2	78	28	6	168	870	374	-510		
11	110	-130	-180	-1028	-178	484	908	-18	208	128	-12	100	28	10	186	948	398	-768		
12	87	-184	-68	516	264	438	930	-32	188	124	-320	652	68	42	114	858	92	-1980		
13	81.25	-314	80	2120	-3084	222	914	-6	142	152	-100	382	48	30	116	984	62	-2082		
14	83.6	-340	150	1906	-2838	68	772	214	-86	140	-168	344	-20	196	-60	864	-56	-2762		
FORCE kN																				21.6
MOMENT (N.m)																				-15.7

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-A

PANEL C(FRONT DIAGONAL TOP HALF BUCKLED INSIDE AND REAR DIAGONAL TOP HALF BUCKLED OUTSIDE)

SERIAL NO. 1

No.	load	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	10	-42	-32	-64	-48	120	32	12	30	18	-18	-6	-18	22		10	8	-78	-38
2	20	-96	-56	-130	-202	266	62	28	66	30	-36	-20	-52	44		86	282	-382	-72
3	30	-70	-90	-196	-452	400	90	44	106	56	-52	-32	-66	54		138	442	-628	-104
4	40	-208	-120	-244	-602	526	114	62	152	90	-70	-38	-116	64		190	568	-852	-64
5	50	-236	-154	-264	-762	668	148	82	184	112	-80	-48	-50	74		252	762	-1060	-108
6	60	-296	-184	-300	-960	842	166	80	218	106	-96	-50	-144	82		304	890	-1344	-72
7	70	-240	-212	-320	-1170	1000	216	100	262	120	-120	-48	-160	66		366	1036	-1568	-32
8	80	-410	-240	-318	-1364	1170	260	108	294	130	-140	-54	-176	104		434	1174	-1840	46
9	90	-412	-272	-306	-1626	1340	302	120	340	148	-162	-48	-190	154		508	1314	-2128	128
10	100	-491	-302	-270	-1882	1510	344	120	394	164	-200	-46	-212	118		588	1460	-2458	246
11	110	-664	-324	-218	-2136	1698	390	132	438	188	-198	-36	-242	164		674	1664	-2822	440
12	87	-324	-288	-842	-1892	1766	616	84	330	146	-166	164	270	-96		10	738	-1842	-218
13	81.25	-294	-254	366	-3218	1592	594	-284	708	238	40	146	136	-182		8	1014	-3124	704
14	63.6	-57	-6			1410	-370	-554	830	220	110	14	154	-376		-170	756		
FORCE kN				-37.5		33.3										36.4		-35.3	
MOMENT (k.m)				53.4		-37.1										26.4		77.7	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-A

PANEL D (NOT BUCKLED)

TEST SERIAL NO. 1

No	load (kN)	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14	Strain 15	Strain 16	Strain 17	Strain 18
1	10	-90	-120	-35	-70	-30	-82	-70	-80	0	44	100	200	136	-650	-1035	-400	-120	-368
2	20	-48	-38	194	-39	-22	-200	-42	28	55	43	690	320	340	65	-100	-10	800	450
3	30	-80	-20	320	55	30	-242	107	106	58	38	690	312	340	91	-223	-83	837	560
4	40	-148	-33	425	54	-6	-411	79	83	44	33	634	260	368	112	-222	-109	940	873
5	50	-146	-54	512	74	15	-516	130	115	72	21	667	285	412	138	-335	-160	870	770
6	60	-196	-58	590	61	25	-640	107	126	88	12	696	321	398	151	-371	-222	815	828
7	70	-152	-11	747	161	103	-713	178	228	115	86	516	312	436	189	-475	-214	835	1011
8	80	-288	-97	751	118	81	-905	128	174	103	-23	-466	-465	-224	-346	-1300	-912	-65	30
9	90	-250	-20	910	207	159	-841	184	255	145	29	-189	-688	-490	-574	-1365	-599	457	870
10	100	-275	-28	1032	233	178	-1092	233	299	148	12	-282	-422	-356	-487	-1353	-728	440	750
11	118	-382	-99	1183	247	199	-1294	238	318	181	-49	290	-398	-354	246	-1369	-828	832	1348
12	87	-248	187	877	137	237	-728	232	309	173	-193	-688	-540	-120	68	-700	-800	619	1287
13	81.25	-219	308	812	171	200	-793	210	282	170	-48	-1000	-500	260	13	-743	-1415	199	-322
14	63.6	-44	630	711	197	180	-400	47	440	161	41	-900	-1080	-800	-413	-180	-148	-821	1083
FORCE (kN)																			
MOMENT (N.m)																			

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-A

TEST SERIAL NO. 1

PANEL E (NOT BUCKLED)

No	load kn	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18
1	10	-4	-17	48	97	-85	0	8	23	3	-13	3	20	20	43	9	-70	143	39
2	20	-18	-36	90	193	-202	-8	-1	41	0	6	6	24	24	60	13	-163	241	60
3	30	-29	-56	137	284	-318	-14	0	59	-5	25	-20	27	23	70	17	-270	335	82
4	40	-44	-78	182	379	-440	-17	3	75	-6	44	-37	20	21	71	13	-375	409	98
5	50	-52	-97	231	467	-572	-18	2	90	-8	65	-51	26	28	98	35	-472	465	129
6	60	-66	-122	266	531	-687	-18	2	102	-22	69	-50	29	28	108	50	-530	592	160
7	70	-78	-138	309	612	-817	-15	6	116	-41	84	-55	31	25	125	70	-762	677	190
8	80	-92	-158	348	684	-942	-9	6	130	-48	97	-67	35	29	120	112	-828	740	207
9	90	-107	-174	387	762	-1074	-7	10	141	-57	111	-69	41	28	147	176	-1111	839	240
10	100	-119	-197	431	847	-1203	-19	14	155	-65	129	-81	48	36	183	275	-1324	927	270
11	110	-133	-214	471	928	-1258	-33	18	170	-73	150	-88	55	38	182	413	-1500	988	286
12	87	-96	-177	266	565	-686	-92	16	122	-171	-71	107	120	0	129	307	-1610	1037	274
13	81.25	-93	-156	320	643	-674	-53	11	118	-92	76	-16	84	24	131	268	-1239	819	210
14	63.6	-88	-120	307	523	-649	-47	8	59	-102	45	-40	55	50	66	136	-877	660	190
FORCE kN																			
MOMENT (k.m)																			

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-A

PANEL F (NOT BUCKLED)

No	Load kN	TEST SERIAL NO. 1																			
		1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18		
1	10	0	0	42	0	0	0	14	0	0	36	0	16	24	-96	-156	-14	-78	98		
2	20	-36	-12	94	-10	-22	-118	46	-52	0	65	0	35	59	-110	-266	-146	-82	176		
3	30	-48	0	150	20	-16	-204	74	-76	0	86	10	46	84	-131	-367	-177	-88	263		
4	40	-56	0	217	50	-9	-302	104	-102	0	103	6	54	104	-142	-448	-200	-90	344		
5	50	-72	-2	268	84	14	-405	137	-132	6	127	7	63	125	-157	-540	-224	-90	428		
6	60	-83	0	344	108	40	-489	170	-155	0	142	2	96	155	-182	-652	-246	-90	537		
7	70	-94	0	400	143	82	-566	196	-187	0	161	2	113	177	-202	-750	-270	-77	630		
8	80	-105	0	482	170	135	-682	222	-205	-16	178	4	136	202	-222	-843	-295	-72	724		
9	90	-122	0	525	212	188	-800	254	-236	-30	194	7	162	231	-244	-930	-326	-82	823		
10	100	-134	0	568	258	297	-936	290	-262	-37	218	10	186	257	-265	-1004	-370	-52	925		
11	110	-148	26	609	293	428	-1092	317	-290	-44	236	7	212	284	-287	-1038	-443	-39	1028		
12	87	-126	0	366	167	130	-516	192	-170	-200	137	4	392	282	-308	-1034	-432	62	1013		
13	61.25	-108	-5	456	180	144	-622	218	-195	-78	182	12	220	220	-244	-838	-352	-40	827		
14	63.6	-97	0	340	60	90	-498	172	-155	-34	168	10	187	176	-209	-707	-298	-102	660		
FORCE kN																					
MOMENT (N.m)																					

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-B

PANEL A (FRONT DIAGONAL BUCKLED WITH BOTTOM HALF OUTSIDE AND TOP INSIDE) Test Serial No. 3

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12
1	10	-99	-21	-70	-60	-58	-72	32	-160	-181	40	-84	-36
2	20	-190	-43	-136	-114	-110	-142	77	-322	-357	87	-186	-85
3	30	-272	-64	-197	-164	-164	-201	131	-486	-532	142	-277	-90
4	35	-313	-78	-228	-191	-194	-230	184	-578	-628	172	-322	-101
5	40	-350	-91	-260	-216	-224	-258	200	-668	-723	205	-371	-112
6	45	-383	-104	-286	-236	-254	-278	238	-758	-812	237	-417	-120
7	47.5	-402	-114	-303	-250	-272	-292	259	-807	-862	258	-440	-123
8	50	-416	-122	-315	-262	-297	-303	280	-852	-908	270	-465	-127
9	52.5	-431	-132	-328	-274	-303	-312	301	-900	-953	287	-488	-130
10	55	-445	-140	-342	-282	-320	-319	322	-948	-998	306	-510	-134
11	60	-472	-159	-368	-308	-354	-338	370	-1046	-1090	341	-560	-138
12	65	-494	-178	-390	-324	-389	-344	420	-1141	-1182	380	-603	-140
13	70	-513	-202	-415	-344	-430	-362	475	-1244	-1278	419	-648	-142
14	75	-530	-228	-439	-363	-472	-366	533	-1350	-1374	462	-696	-142
15	80	-542	-258	-461	-383	-520	-371	595	-1460	-1471	505	-744	-142
16	85	-551	-289	-482	-400	-569	-370	662	-1577	-1572	552	-793	-140
17	90	-554	-324	-501	-418	-624	-362	734	-1694	-1672	598	-842	-135
18	92.5	-553	-344	-512	-426	-652	-358	772	-1758	-1722	624	-866	-134
19	95	-552	-361	-510	-434	-679	-352	808	-1814	-1768	648	-890	-130
20	97.5	-547	-382	-528	-442	-712	-342	852	-1883	-1821	676	-918	-124
21	100	-542	-406	-536	-450	-750	-330	898	-1958	-1862	708	-944	-120
22	102.5	-536	-430	-544	-459	-780	-320	942	-2022	-1928	734	-968	-114
23	105	-530	-455	-562	-466	-814	-308	996	-2121	-1981	768	-996	-108
24	107.5	-518	-480	-556	-472	-854	-289	1075	-2163	-2040	796	-1024	-102
25	110	-507	-507	-564	-478	-895	-270	1150	-2284	-2097	832	-1052	-92
26	112.5	-490	-538	-570	-484	-942	-244	1240	-2450	-2156	870	-1082	-80
27	115	-468	-574	-573	-490	-998	-210	1347	-2676	-2222	910	-1114	-68
28	117.5	-444	-610	-578	-494	-1054	-173	1484	-3008	-2284	954	-1145	-35
29	120	-414	-648	-578	-498	-1118	-126	1616	-3284	-2350	998	-1178	-38

30	122.6	-390	-699	-578	-498	-1188	-75	1764	-3590	-2418	1050	-1212	-19
31	125	-338	-733	-576	-498	-1266	-10	1960	-4009	-2494	1112	-1249	5
32	127.6	-287	-782	-672	-698	-1387	68	2244	-4660	-2575	1182	-1290	38
33	130	-209	-840	-562	-482	-1480	200	2805	-5818	-2692	1309	-1346	94
34	132.5	-80	-923	-638	-450	-1658	405	4600	-8900	-2848	1492	-1424	183
35	135	108	-1022	-500	-384	-1908	722	8350	-13144	-3170	1786	-1580	310
36	137.5	212	-1098	-480	-350	-2068	913	10378	-15380	-3475	2008	-1720	395
37	140	274	-1132	-478	-330	-2171	1035	11317	-16438	-3608	2120	-1808	435
38	142.5	344	-1180	-473	-302	-2283	1184	12244	-17404	-3728	2220	-1888	468
39	145	1144	-1598	-670	190	-9450	5860	20850	-5193	-5193	3600	-2858	1009
AXIAL FORCE (kN)		-17.1		-17.1		-20.7		-24.9		-21.2		-20.1	
MOMENT(N.m)		14.2		2.2		40.8		130.8		100.9		38.0	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-B

PANEL A (REAR DIAGONAL BUCKLED WITH TOP HALF OUTSIDE AND BOTTOM INSIDE) Test Serial No. 3

No.	Load kN	Strain	17	18	19	20	21	22	23	24	25	26	27	28
1	10	-94	-6	-56	-40	-10	-86	-22	-74	-90	-40	-84		
2	20	-214	-4	-124	-84	-16	-196	-44	-106	-224	-28	-88		
3	30	-336	2	-192	-126	-16	-312	-66	-254	-360	50	-288		
4	35	-396	8	-226	-146	-10	-370	-76	-298	-426	60	-332		
5	40	-460	14	-260	-166	-4	-430	-86	-344	-494	72	-384		
6	45	-526	24	-296	-186	0	-494	-96	-390	-562	84	-432		
7	47.5	-564	34	-315	-196	4	-530	-100	-414	-596	90	-462		
8	50	-596	36	-334	-206	10	-562	-106	-436	-630	96	-486		
9	52.5	-630	46	-352	-216	16	-596	-110	-460	-667	102	-516		
10	55	-666	48	-372	-226	30	-632	-116	-480	-698	107	-542		
11	60	-742	62	-410	-244	48	-704	-126	-526	-766	118	-595		
12	65	-818	82	-447	-262	70	-780	-140	-570	-836	130	-650		
13	70	-892	104	-484	-276	96	-855	-152	-612	-900	140	-700		
14	75	-972	126	-522	-294	126	-936	-166	-652	-966	150	-751		
15	80	-1054	156	-562	-308	162	-1022	-180	-688	-1028	156	-804		
16	85	-1138	180	-600	-310	204	-1116	-200	-722	-1096	164	-854		
17	90	-1226	214	-636	-332	226	-1210	-224	-750	-1148	170	-900		
18	92.5	-1270	234	-658	-336	250	-1258	-235	-764	-1172	170	-928		
19	95	-1316	254	-680	-344	276	-1312	-248	-776	-1200	170	-950		
20	97.5	-1366	274	-700	-348	312	-1364	-264	-785	-1225	168	-974		
21	100	-1426	304	-720	-353	340	-1426	-283	-792	-1251	166	-996		
22	102.5	-1462	321	-739	-356	374	-1477	-299	-797	-1271	162	-1019		
23	105	-1512	344	-758	-358	410	-1534	-320	-800	-1290	158	-1041		
24	107.5	-1566	372	-778	-361	451	-1597	-343	-800	-1309	151	-1062		
25	110	-1616	401	-798	-364	497	-1662	-368	-795	-1325	143	-1082		
26	112.5	-1675	434	-821	-366	548	-1732	-398	-790	-1340	132	-1102		
27	115	-1741	468	-844	-368	602	-1812	-434	-779	-1352	117	-1123		
28	117.5	-1802	508	-864	-368	663	-1888	-472	-762	-1368	100	-1142		

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-C

Test Serial No. 5

PANEL A (FRONT DIAGONAL)

No.	Load (k)	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12
1	10.0	-18	-128	-75	-72	-120	-28	-126	-35	48	-183	-14	-132
2	20.0	-30	-212	-116	-120	-207	-34	-190	-52	78	-300	-22	-210
3	30.0	-36	-316	-172	-172	-314	-42	-286	-70	120	-450	-32	-312
4	40.0	-34	-446	-228	-236	-440	-38	-304	-87	172	-624	-42	-430
5	50.0	-22	-578	-286	-294	-570	-32	-500	-105	224	-788	-48	-538
6	60.0	10	-720	-342	-354	-708	-8	-605	-116	280	-952	-54	-652
7	70.0	42	-864	-394	-408	-860	34	-706	-132	332	-1128	-58	-758
8	75.0	62	-936	-418	-438	-946	56	-760	-138	364	-1208	-62	-812
9	80.0	88	-1022	-442	-468	-1028	86	-810	-146	383	-1288	-62	-868
10	85.0	114	-1102	-468	-492	-1117	118	-860	-154	420	-1370	-65	-922
11	90.0	145	-1188	-490	-520	-1203	156	-902	-170	440	-1444	-70	-988
12	95.5	184	-1280	-513	-546	-1310	202	-944	-184	468	-1524	-72	-1024
13	97.5	212	-1345	-528	-564	-1378	238	-968	-188	476	-1572	-74	-1055
14	100.0	225	-1384	-532	-570	-1410	255	-972	-203	483	-1590	-75	-1070
15	102.5	250	-1428	-548	-586	-1474	290	-990	-220	492	-1628	-80	-1100
16	105.0	275	-1473	-556	-598	-1532	322	-1002	-232	500	-1680	-84	-1120
17	107.5	305	-1528	-566	-612	-1590	363	-1008	-250	502	-1690	-86	-1142
18	110.0	332	-1578	-578	-622	-1646	400	-1018	-266	504	-1718	-80	-1164
19	112.5	362	-1634	-585	-632	-1706	438	-1024	-286	508	-1740	-84	-1182
20	115.0	414	-1720	-598	-650	-1805	506	-1020	-324	508	-1772	-103	-1210
21	117.5	442	-1760	-604	-658	-1852	547	-1018	-346	500	-1784	-105	-1220
22	120.0	482	-1838	-610	-668	-1944	612	-1004	-384	490	-1800	-105	-1240
23	122.5	550	-1918	-620	-680	-2036	698	-982	-437	476	-1812	-124	-1258
24	125.0	614	-2000	-628	-690	-2098	796	-946	-492	454	-1812	-137	-1270
25	127.5	676	-2082	-630	-694	-2180	928	-878	-574	410	-1785	-152	-1272
26	130.0	803	-2383	-622	-698	-2356	1312	-690	-790	308	-1768	-180	-1308
27	133.1	1412	-2840	-636	-714	-4500	2786	28	-1410	-105	-1208	-280	-1074
28	135.0	2226	-3823	-652	-676	-9840	6236	798	-1942	-403	-728	-308	-874
29	137.5	2312	-3993	-652	-656	-10360	6716	1003	-2080	-470	-560	-300	-803

30	140.0	2376	-3994	-642	-626	-10560	6600	1140	-2160	-515	-522	-306	-760
31	142.5	2440	-3920	-635	-605	-10640	6674	1250	-2256	-538	-460	-295	-746
32	145.0	2470	-3930	-632	-594			1326	-2310	-598	-425	-295	-730
AXIAL FORCE (N)		-23.1		-21.2		-19.9		-23.7		-22.4		-23.2	
MOMENT(N.m)		89.4		2.2		163.4		2.9		57.7		31.2	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-C

Test Serial No. 5

PANEL A (REAR DIAGONAL)

No.	Load kN	Strain	17	Strain	18	Strain	19	Strain	20	Strain	21	Strain	22	Strain	23	Strain	24	Strain	25	Strain	26	Strain	27	Strain	28
1	10.0	18	-124	-86	-30	-98	-18	-108	-16	-28	-94	-36	-64												
2	20.0	38	-228	-150	-42	-177	-23	-178	-20	-34	-166	-57	-115												
3	30.0	64	-360	-248	-73	-281	-23	-283	-18	-45	-265	-84	-178												
4	40.0	94	-486	-330	-80	-379	-33	-369	-11	-53	-368	-108	-232												
5	50.0	124	-605	-405	-106	-471	-30	-450	5	-55	-448	-129	-288												
6	60.0	156	-733	-486	-124	-573	-27	-610	27	-50	-552	-150	-342												
7	70.0	192	-855	-563	-140	-672	-11	-732	59	-41	-647	-183	-394												
8	76.0	209	-920	-610	-150	-728	-15	-779	70	-35	-708	-172	-415												
9	80.0	222	-982	-646	-155	-772	-7	-870	88	-28	-754	-178	-445												
10	85.0	245	-1050	-686	-157	-825	0	-931	111	-20	-819	-186	-468												
11	90.0	257	-1107	-723	-167	-878	4	-1000	140	-10	-873	-187	-491												
12	95.5	273	-1172	-765	-172	-931	12	-1071	165	5	-930	-196	-514												
13	97.5	287	-1205	-785	-183	-982	13	-1108	177	14	-972	-201	-528												
14	100.0	290	-1224	-794	-184	-971	17	-1130	190	20	-985	-202	-538												
15	102.5	300	-1255	-819	-185	-1005	23	-1172	202	28	-1116	-203	-550												
16	105.0	305	-1288	-832	-185	-1031	27	-1205	218	35	-1047	-206	-560												
17	107.5	311	-1315	-855	-200	-1055	25	-1255	235	46	-1170	-211	-569												
18	110.0	318	-1341	-877	-200	-1074	30	-1285	250	55	-1104	-214	-581												
19	112.5	325	-1360	-892	-192	-1098	27	-1342	274	61	-1130	-218	-582												
20	115.0	332	-1406	-919	-193	-1134	30	-1389	308	75	-1185	-220	-607												
21	117.5	338	-1424	-929	-217	-1149	35	-1432	322	80	-1217	-220	-614												
22	120.0	343	-1455	-949	-204	-1175	35	-1475	347	85	-1257	-218	-624												
23	122.5	345	-1488	-968	-211	-1202	40	-1540	371	108	-1296	-223	-636												
24	125.0	352	-1512	-970	-228	-1228	40	-1550	401	125	-1330	-220	-647												
25	127.5	354	-1533	-996	-238	-1246	38	-1640	430	139	-1355	-224	-655												
26	130.0	358	-1565	-1024	-254	-1275	33	-1686	469	162	-1410	-222	-666												
27	133.1	366	-1602	-1049	-255	-1310	40	-1767	523	197	-1471	-221	-675												
28	135.0	364	-1661	-1085	-250	-1348	38	-1888	618	266	-1563	-212	-680												

29	137.5	365	-1698	-1112	-266	-1363	22	-2031	705	317	-1657	-206	-665
30	140.0	366	-1717	-1130	-278	-1360	-19	-2190	825	379	-1771	-195	-660
31	142.5	341	-1669	-1133	-311	-1304	-101	-2457	1046	470	-1890	-180	-664
32	145.0	238	-1599	-1090	-355	-1145	-269	-3135	1437	618	-2044	-139	-632
AXIAL FORCE (KN)		-21.9	-23.2	-22.9	-22.8	-22.8	-22.8	-12.7					
MOMENT(N.m)		52	21.1	24.8	110.1	75.5	14.4						

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S4-C

PANEL A (FRONT AND REAR TENSION DIAGONAL)

No.	Load (k)	Test Serial No. 5									
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	10.0	4	14	16	16	29	30	31	32		
2	20.0	6	98	18	76	4	82	2	74		
3	30.0	6	162	30	126	30	146	6	136		
4	40.0	12	240	42	194	14	230	10	214		
5	50.0	20	328	56	266	20	310	16	286		
6	60.0	26	412	74	336	30	384	24	354		
7	70.0	40	502	94	414	42	462	34	426		
8	75.0	46	592	112	488	56	544	48	502		
9	80.0	52	636	122	526	64	582	56	538		
10	85.0	60	680	134	566	72	622	64	574		
11	90.0	60	726	146	606	82	662	74	610		
12	95.5	70	766	156	640	94	700	84	646		
13	97.5	86	814	170	660	104	744	96	684		
14	100.0	84	842	180	704	110	766	104	706		
15	102.5	86	856	182	714	114	776	106	716		
16	105.0	94	880	190	734	122	802	114	736		
17	107.5	100	904	196	754	126	822	122	754		
18	110.0	106	924	202	772	134	840	126	772		
19	112.5	110	946	210	780	142	856	134	780		
20	115.0	114	966	216	806	146	876	140	806		
21	117.5	122	996	226	830	156	904	150	834		
22	120.0	136	1012	232	842	160	918	154	844		
23	122.5	146	1036	244	862	170	940	162	866		
24	125.0	146	1062	254	882	178	964	174	886		
25	127.5	156	1084	264	896	186	984	180	906		
26	130.0	162	1104	274	910	184	1004	186	920		
27	133.1	176	1132	284	912	202	1026	196	944		
28	135.0	200	1172	364	922	212	1054	202	966		
29	135.0	356	1240	506	926	222	1080	210	996		

29	137.5	424	1288	565	956	234	1110	222	1024
30	140.0	472	1324	618	984	250	1140	238	1052
31	142.5	516	1356	656	1014	260	1170	254	1080
32	145.0	544	1380	688	1030	268	1180	272	1100
AXIAL FORCE (kN)		21		19.3		23.7		22.0	
MOMENT (N.m)		-27.4		-17.7		-25.9		-23.7	

PANEL A (NOT BUCKLED)
AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S5-A

Test Serial No.2

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14
1	10	-11	10	-9	-30	9	62	29	-30	20	-19	72	26	8	-100
2	20	-28	18	-206	-54	15	123	56	-52	56	-43	126	56	8	-202
3	30	-43	27	-299	-77	20	176	82	-73	70	-63	240	87	10	-330
4	40	-55	34	-400	-91	26	239	106	-96	92	-83	317	122	22	-436
5	50	-72	43	-516	-96	34	310	136	-120	114	-100	398	167	38	-538
6	60	-89	52	-648	-81	58	402	164	-146	134	-118	484	228	65	-626
7	65	-95	52	-705	-60	60	450	176	-158	138	-122	530	273	82	-660
8	70	-100	52	-745	-32	87	506	181	-163	140	-126	582	338	98	-670
9	71.6	-98	48	-740	-14	112	528	179	-161	136	-124	604	378	105	-656
10	69.3	-86	36	-586	14	168	518	140	-124	138	-126	601	400	99	-615
11	61.4	76	-134	-266	176	288	568	60	-56	172	-150	670	190	296	-1176

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S5-A

PANEL B (NOT BUCKLED)

Test Serial No.2

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14
1	10	15	-18	-19	-74	23	41	4	24	26	-4	28	103	-76	-8
2	20	20	-50	-32	-144	230	80	-4	40	48	-13	57	234	-186	-15
3	30	29	-83	-49	-201	316	120	-4	56	64	-17	88	340	-282	-23
4	40	36	-113	-76	-270	412	165	-6	72	80	-28	106	440	-375	-28
5	50	53	-148	-113	-339	504	213	-8	89	101	-35	140	532	-476	-50
6	60	78	-198	-171	-430	598	265	-7	112	122	-48	176	628	-580	-86
7	65	100	-232	-214	-480	638	286	-9	124	133	-58	179	661	-646	-125
8	70	140	-284	-280	-556	668	292	-11	145	158	-72	174	700	-742	-184
9	71.6	170	-312	-314	-598	668	275	-11	149	162	-72	158	714	-801	-216
10	69.3	222	-362	-367	-708	633	181	-10	126	168	-74	140	712	-814	-238
11	61.4	492	-610	-684	-894	762	-60	-8	114	80	-10	-656	1056	2210	

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S5-A

PANEL C(FRONT COMPRESSION DIAGONAL BUCKLED OUTSIDE TOTALLY AND REAR DIAGONAL

INSIDE

Test Serial No.2

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14
1	10	-27	0	-178	-24	-40	86	3	12	4	14	60	-101	-230	-52
2	20	-76	0	-378	-42	-94	168	10	25	10	32	156	-154	-462	-40
3	30	-108	0	-568	-38	-154	254	15	38	14	56	252	-220	-686	50
4	40	-140	6	-802	0	-228	370	22	52	20	58	368	-308	-954	130
5	50	-176	12	-1110	90	-328	522	32	64	28	77	514	-440	-1278	278
6	60	-228	30	-1542	287	-470	730	42	76	40	88	720	-636	-1700	736
7	65	-264	48	-1786	420	-568	870	50	78	48	92	860	-770	-2018	1156
8	70	-320	82	-2748	884	-732	1080	62	76	58	92	1050	-946	-2608	1548
9	71.6	-346	104	-4690	1780	-800	1160	70	66	60	92	1130	-1032	-3202	1906
10	69.3	-386	150			-760	1055	88	42	58	92	1140	-1036	-3660	
11	61.4	-718	514			-1054	1088	112	6	-718					
12															
	FORCE kN					7.1							1.9		-21.6
	MOMENT (N.m)					-77.1							-85		175.4

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S5-A

PANEL D(FRONT COMPRESSION DIAGONAL BUCKLED OUTSIDE TOTALLY AND REAR DIAGONAL INSIDE

Test Serial No.2

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14
1	10	-150	-19	-90	59	-40	-30	61	-16	-27	-101	390	50	62	118
2	20	-220	-9	-170	153	-50	-44	66	-8	-5	-85	320	-202	172	22
3	30	-270	-16	-250	250	-47	-50	70	19	-22	-70	320	-440	266	-178
4	40	-290	5	-352	360	-50	-3	67	23	-12	-63	440	-664	390	-180
5	50	-318	21	-464	516	-42	83	80	40	16	-36	610	-1232	586	-292
6	60	-415	40	-685	730	-49	245	95	40	37	-49	934	-1585	826	-530
7	65	-454	56	-837	883	-52	387	112	47	49	-20	1332	-2378	996	-508
8	70	-492	84	-1049	1115	-57	658	116	49	66	-37	2278	-3340	1160	-754
9	71.6	-578	108	-1174	1245	-62	960	136	42	70	-85	3172	-5022	1218	-810
10	69.3	-505	146	-1296	1307	-55	1651	114	57	74	-76	4720	-7730	1160	-770
11	61.4	-970	535	-1414	1450	-85		117	23	-17	-20			1152	-1194
	FORCE(kN)			1.4		17.8							-20.9		8.1
	MOMENT (N.m)			95.1		-40.2							212.5		-79.7

PANEL E (NOT BUCKLED)

Test Serial No.2

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14
1	10	14	-29	80	60	-8	-58	-9	24	-6	26	-125	-22	60	143
2	20	12	-61	187	117	-20	-148	-17	40	-20	46	-225	-36	114	263
3	30	36	-92	280	172	-29	-224	-21	49	-29	62	-318	-51	129	366
4	40	38	-126	360	228	-35	-305	-28	67	-39	82	-426	-72	173	474
5	50	49	-161	454	285	-63	-386	-35	81	-47	100	-560	-98	218	578
6	60	74	-211	509	347	-84	-475	-47	104	-60	123	-736	-134	271	682
7	65	100	-245	542	375	-96	-537	-52	117	-68	139	-844	-154	309	721
8	70	130	-292	564	399	-115	-635	-65	132	-80	161	-1069	-170	315	746
9	71.5	134	-345	539	367	-141	-720	-92	109	-106	122	-1137	-218	216	750
10	69.3	132	-448	456	264	-191	-858	-155	56	-148	50	-1186	-276	134	710
11	61.4	380	-680	460	109	520	-2026	-164	61	-48	-50			-170	735

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S5-A
PANEL F (NOT BUCKLED)

Test Serial No.2

No.	Load kN	Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12	Strain 13	Strain 14
1	10	-16	10	-10	70	-92	-25	22	-19	37	-32	-26	-125	106	9
2	20	-20	20	5	148	-210	-50	44	-36	71	-57	-31	-245	202	25
3	30	-40	27	15	217	-306	-62	66	-51	97	-78	-34	-343	292	27
4	40	-60	36	27	289	-411	-73	80	-65	124	-103	-20	-461	364	39
5	50	-71	44	33	370	-522	-79	103	-77	155	-127	0	-580	467	60
6	60	-80	51	35	451	-626	-68	110	-90	180	-154	33	-701	587	94
7	65	-91	52	38	511	-670	-64	117	-92	189	-164	49	-736	658	124
8	70	-96	50	60	561	-700	-44	110	-92	189	-166	69	-756	722	172
9	71.6	-85	45	80	596	-660	-40	110	-87	184	-157	76	-726	760	206
10	69.3	-80	44	128	596	-606	-37	102	-76	164	-140	78	-640	728	227
11	61.4	112	-152	276	701	-217	40	7	7	212	-176	87	-1082	788	-146

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S5-B

PANEL C (FRONT DIAGONAL BUCKLED)

No.	Load (kN)	Test Serial No. 10													
		Strain 1	Strain 2	Strain 3	Strain 4	Strain 5	Strain 6	Strain 7	Strain 8	Strain 9	Strain 10	Strain 11	Strain 12		
1	11.8	-50	-130	-108	-86	-152	-62	-156	-70	-10	-206	-128	-128		
2	20	-83	-272	-181	-168	-273	-100	-267	-93	-11	-347	-197	-155		
3	30	-58	-512	-323	-240	-502	-74	-511	-77	73	-688	-351	-215		
4	40	-37	-588	-380	-326	-606	-63	-623	-71	157	-824	-390	-248		
5	50	0	-837	-454	-389	-782	-61	-792	-61	180	-1053	-486	-304		
6	60	20	-1024	-533	-477	-962	-15	-946	-15	306	-1298	-578	-352		
7	70	84	-1208	-582	-512	-1117	0	-1092	0	390	-1475	-640	-359		
8	75	100	-1296	-595	-566	-1188	0	-1140	0	406	-1581	-650	-374		
9	80	144	-1371	-620	-566	-1294	7	-1238	0	482	-1694	-693	-394		
10	85	206	-1525	-675	-623	-1382	42	-1340	0	570	-1832	-752	-405		
11	90	277	-1675	-690	-675	-1592	128	-1461	51	674	-2096	-822	-468		
12	95	352	-1796	-764	-717	-1694	176	-1521	80	722	-2185	-852	-448		
13	87.5	184	-1667	-700	-729	-1734	188	-1440	0	512	-2112	-787	-447		
14	100	432	-1937	-777	-754	-2225	537	-1430	20	926	-2199	-847	-626		
15	98.6														
16															
AXIAL FORCE (kN)		-29.8													
MOMENT (N.m)		93.1													
		-30.3													
		-33.2													
		-27.9													
		-25.0													
		121.5													
		-1.1													
		-33.0													
		-1.1													

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S3-B

PANEL C (REAR DIAGONAL NOT BUCKLED)

No.	Load kN	Test Serial No. 10														
		Strain 17	Strain 18	Strain 19	Strain 20	Strain 21	Strain 22	Strain 23	Strain 24	Strain 25	Strain 26	Strain 27	Strain 28			
1	11.6	33	-123	-72	-18	-98	5	-97	7	-108	24	-47	-36			
2	20	55	-203	-117	-30	-163	12	-161	12	-180	42	-74	-55			
3	30	164	-490	-272	-65	-383	48	-383	54	-422	123	-164	-118			
4	40	217	-614	-334	-74	-478	71	-481	80	-528	166	-197	-140			
5	50	323	-837	-440	-90	-642	116	-645	132	-707	245	-252	-173			
6	60	433	-1050	-538	-98	-800	167	-804	190	-880	332	-300	-200			
7	70	533	-1234	-820	-104	-936	219	-944	246	-1028	408	-340	-222			
8	75	583	-1325	-660	-106	-1003	245	-1014	276	-1098	448	-360	-230			
9	80	640	-1422	-700	-105	-1076	277	-1089	312	-1180	494	-378	-240			
10	85	715	-1548	-755	-106	-1171	317	-1183	358	-1275	550	-402	-250			
11	90	816	-1712	-825	-102	-1297	375	-1314	424	-1406	632	-430	-263			
12	95	932	-1888	-897	-94	-1430	445	-1453	503	-1542	722	-462	-272			
13	87.5	1322	-2274	-1048	33	-1704	900	-1706	750	-1904	1120	-474	-202			
14	100	1748	-2868	-1290	120	-1884	891	-1948	904	-2178	1542	-557	-210			
15	96.6	2324	-3122	-1683	168	-1950	1009	-2371	1100	-2390	1900	-642	-238			
16																
AXIAL FORCE (kN)		-16.9		-23.2		-19.7		-20.7		-12.4		-17.4				
MOMENT(N.m)		164.4		55.4		109.1		112.1		145.3		15.9				

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN S5-B

PANEL C (FRONT AND REAR TENSION DIAGONAL)

No.	Load (kN)	Test Serial No. 10																							
		Strain	13	Strain	14	Strain	15	Strain	16	Strain	17	Strain	18	Strain	19	Strain	20	Strain	21	Strain	22	Strain	23	Strain	24
1	11.8		-10		64		-10		64		-10		64		-34		30		31		32		33		34
2	20		-18		104		-20		100		-54		70		-68		70		70		70		70		70
3	30		-56		228		-54		218		-118		168		-162		168		168		168		168		168
4	40		-74		284		-66		270		-142		214		-186		214		214		214		214		214
5	50		-104		374		-92		354		-188		294		-268		294		294		294		294		294
6	60		-138		458		-124		430		-224		370		-332		370		370		370		370		370
7	70		-168		534		-146		408		-256		434		-388		434		434		434		434		434
8	75		-182		572		-158		408		-268		468		-414		468		468		468		468		468
9	80		-198		612		-172		566		-286		504		-442		504		504		504		504		504
10	85		-220		666		-188		616		-304		550		-478		550		550		550		550		550
11	90		-246		738		-214		658		-334		612		-526		612		612		612		612		612
12	95		-276		808		-236		742		-374		678		-580		678		678		678		678		678
13	87.5		-278		788		-270		752		-830		910		-820		910		910		910		910		910
14	100		-324		886		-328		854		-1112		1168		-1208		1168		1168		1168		1168		1168
15	98.6		-714		1000		-848		1020		-1306		1380		-1396		1380		1380		1380		1380		1380
16																									
AXIAL FORCE (kN)			11.1				10.4				1.1				0										
MOMENT (k.m)			-47.6				-46.5				-89.6				-94.9										

AXIAL FORCES AND BENDING MOMENTS IN MEMBERS FOR SPECIMEN SS-B

PANEL C (THREE LEGS)

No.	Load (kN)	Test Serial No. 10																							
		Strain	33	Strain	34	Strain	35	Strain	36	Strain	37	Strain	38	Strain	39	Strain	40	Strain	41	Strain	42	Strain	43	Strain	44
1	11.8		-61		32		0		-31		13		0		0		24		24		14		0		20
2	20		-88		52		10		-46		24		0		0		32		32		14		0		28
3	30		-200		110		20		-94		42		0		0		60		60		8		0		52
4	40		-242		134		10		-112		56		0		10		66		72		14		-10		66
5	50		-322		180		16		-144		70		-8		12		80		84		12		0		76
6	60		-390		220		28		-170		80		-12		12		92		104		10		0		94
7	70		-450		256		26		-198		92		-12		16		102		118		10		4		104
8	75		-478		270		26		-206		102		-14		18		108		120		0		0		108
9	80		-512		284		28		-222		100		-20		20		114		136		4		0		122
10	85		-552		318		32		-236		110		-20		22		120		142		8		0		130
11	90		-606		350		36		-256		120		-20		26		126		150		0		0		134
12	95		-666		390		50		-278		124		-20		24		136		168		-6		10		154
13	87.5		-758		496		88		-306		82		42		-12		162		176		-18		-6		158
14	100		-806		620		128		-364		72		60		-14		174		264		-30		-8		162
15	98.6		-1280		964		248		-466		80		50		-30		190		114		50		48		110
16																									
AXIAL FORCE(NN)			-5.7		-4.6		2.6		3.2		4.6		3.1		4.6		3.1		4.6		3.1		3.1		3.1
MOMENT(N.m)			60.0		18.6		-0.5		-7.4		-11.6		-6.7		-11.6		-6.7		-11.6		-6.7		-6.7		-6.7

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN 35-C

TEST PANEL C (FRONT DIAGONAL BUCKLED FIRST)

No.	Load(kN)	Test Serial No. 12														
		1	2	3	4	5	6	7	8	9	10	11				
1	14.8	-1.26	-56	-122	-49	-26	54	-21	47							
2	28.3	-253	-107	-268	-88	-51	107	-43	98							
3	43.9	-437	-160	-471	-119	-92	189	-82	178							
4	58.5	-583	-186	-644	-123	-133	257	-124	244							
5	70.7	-785	-188	-875	-96	-189	352	-189	336							
6	84.8	-1338	-37	-1502	149	-386	647	-382	630							
7	91.8	-1408	-8	-1588	197	-421	681	-408	673							
8	94.7	-1462	9	-1685	257	-446	738	-438	673							
9	98.9	-1476	12	-1723	283	-458	757	-448	740							
10	100.5	-1506	27	-1762	306	-467	780	-457	783							
11	102.2	-1521	24	-1824	353	-481	808	-474	783							
12	104.0	-1524	12	-1885	410	-494	837	-489	824							
13	108.0	-1828	2	-1948	487	-504	881	-492	848							
14	108.0	-1487	618	-2025	618	-519	904	-525	886							
15	111.1	-1337	928	-2185	928	-528	950	-549	954							
16	114.3	-1134	-403	-2318	1283	-534	973	-559	968							
17	118.9	-964	-910	-3895	2305	-489	997	-557	1031							
18	118.8	101	-1451	-6636	4818	-422	989	-523	1042							
19	120.4	788	-1868	-14938	12376	-332	991	-482	1059							
20	122.6	843	-2001	-15446	12830	-324	995	-458	1084							
21	123.8	864	-2286	-18273	13888	-318	1008	-481	1078							
22	107.4	747	-2031	-15823	13190	-378	996	-464	1025							
23	109.4	783	-2082	-15897	13250	-379	1009	-465	1038							
24	110.8	770	-2084	-15958	13297	-384	1022	-468	1049							
25	113.2	785	-2111	-16033	13358	-388	1037	-472	1062							
26	114.5	795	-2136	-16118	13423	-396	1052	-478	1076							
27	115.4	818	-2177	-16274	13550	-408	1078	-488	1100							
28	105.0	711	-2084	-16674	13915	-451	1153	-505	1151							
29	106.7	803	-1855	-24787	22289	-863	1120	-1166	1267							
AXIAL FORCE (kN) FOR 188.8kN			30.2	29.0	29.0	7.1	6.9									
BENDING MOMENT(N.m)			60.1	91	53.7	13.7	12.4									
AXIAL FORCE (kN) FOR 123.8kN			-28.4	-8.7	-51.9											
BENDING MOMENT(N.m)			125.4	251.7												

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN S5-C

TEST PANEL C (REAR DIAGONAL BUCKLED SECOND)

No.	Load(kN)	Test Serial No. 12									
		strain	strain	strain	strain	strain	strain	strain	strain	strain	strain
1	14.8	-131	-21	19	11	12	13	14	15	16	18
2	28.3	-278	-39	29	-123	-24	-29	48	-24	50	
3	43.9	-487	-44	44	-268	-43	61	104	-55	110	
4	58.5	-662	-28	28	-478	-46	-114	180	-107	202	
5	70.7	-888	16	16	-658	-24	-162	285	-156	283	
6	84.8	-1428	209	209	-882	31	-231	368	-221	382	
7	91.8	-1495	238	238	-1481	277	-398	640	-383	681	
8	94.7	-1564	266	266	-1558	318	-417	677	-401	719	
9	98.9	-1586	272	272	-1638	362	-437	713	-421	758	
10	100.5	-1618	293	293	-1688	378	-444	728	-428	773	
11	102.2	-1652	308	308	-1704	399	-452	744	-437	792	
12	104.0	-1685	323	323	-1749	428	-463	785	-448	816	
13	108.8	-1788	333	333	-1793	451	-473	787	-457	839	
14	108.0	-1752	350	350	-2081	481	-482	888	-468	888	
15	111.1	-1788	388	388	-2148	542	-497	837	-482	884	
16	114.3	-1822	375	375	-2300	602	-512	875	-498	938	
17	118.3	-1836	385	385	-2395	643	-519	897	-508	960	
18	118.6	-1831	343	343	-2502	734	-528	928	-517	1001	
19	120.4	-1741	286	286	-2655	856	-522	981	-519	1031	
20	122.6	-1719	201	201	-3537	1181	-520	1002	-531	1101	
21	123.9	-1682	186	186	-3780	1230	-523	1011	-538	1115	
22	107.4	-1828	498	498	-4118	1388	-543	1038	-568	1181	
23	109.4	-1839	504	504	-4248	1650	-1050	1125	-1105	1301	
24	110.8	-1778	457	457	-4319	1702	-1081	1144	-1119	1328	
25	113.2	-1880	381	381	-4788	1875	-1074	1155	-1148	1355	
26	114.5	-1518	250	250	-5079	2023	-1082	1165	-1170	1384	
27	115.4	-1212	9	9	-5895	2281	-1089	1171	-1197	1416	
28	105.0	-548	-401	-401	-6666	2879	-1097	1176	-1238	1462	
29	108.7	-158	-583	-583	-7874	3386	-1222	1177	-1431	1541	
AXIAL FORCE (kN) FOR 188.8kN		-27.3	-29.9	6.4	7.8	1000	-1140	-1448	-1488	1484	
BENDING MOMENT (kN.m)		80.3	87.9	-80.8	-82.1	8.4	8.4	7.8	-82.1	-82.1	
AXIAL FORCE (kN) for 172.8kN		30.2	-40.4	9.8	11.5						
BENDING MOMENT (kN.m)		72.1	148.3	-82.1	-87.8						

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN 85-C

TEST PANEL C (LEGS)

Sta.	Load (kip)	Strain	17	18	19	20	21	22	23	24	25	26	27	28
1	14.8	Strain												
2	28.3	Strain	-56	-29	2	-28	17	0	0	17	18	2	2	20
3	114	Strain	62	62	3	55	36	1	1	36	36	4	3	17
4	182	Strain	106	106	6	86	59	2	2	59	59	5	4	36
5	254	Strain	141	141	9	128	84	-4	-4	74	74	5	4	54
6	325	Strain	185	185	14	165	99	-5	-5	93	93	7	6	77
7	397	Strain	227	227	22	247	148	-10	-10	128	128	10	7	96
8	472	Strain	269	269	22	256	155	-11	-11	130	134	12	12	132
9	547	Strain	311	311	22	264	161	-13	-13	135	140	12	18	134
10	624	Strain	353	353	22	268	163	-13	-13	135	145	12	21	137
11	700	Strain	394	394	22	272	167	-13	-13	136	148	11	23	138
12	775	Strain	436	436	22	277	171	-14	-14	137	149	11	24	138
13	851	Strain	478	478	21	283	174	-16	-16	137	153	11	26	138
14	927	Strain	520	520	20	287	176	-16	-16	138	156	10	28	140
15	1000	Strain	561	561	20	294	180	-17	-17	141	162	9	34	130
16	1074	Strain	602	602	17	302	189	-19	-19	142	166	9	41	132
17	1145	Strain	643	643	15	307	193	-20	-20	144	169	9	47	132
18	1218	Strain	683	683	13	315	199	-21	-21	144	167	13	65	117
19	1290	Strain	723	723	13	321	203	-23	-23	143	158	17	89	90
20	1363	Strain	763	763	13	328	210	-27	-27	141	149	28	128	51
21	1436	Strain	803	803	20	334	211	-28	-28	141	149	28	128	51
22	1509	Strain	843	843	22	340	211	-34	-34	141	149	28	128	51
23	1582	Strain	883	883	22	346	211	-34	-34	141	149	28	128	51
24	1655	Strain	923	923	20	352	210	-47	-47	141	149	28	128	51
25	1728	Strain	963	963	17	358	210	-50	-50	141	149	28	128	51
26	1801	Strain	1003	1003	11	364	209	-51	-51	141	149	28	128	51
27	1874	Strain	1043	1043	11	369	209	-56	-56	141	149	28	128	51
28	1947	Strain	1083	1083	11	375	209	-56	-56	141	149	28	128	51
29	2020	Strain	1123	1123	11	380	209	-56	-56	141	149	28	128	51
30	2093	Strain	1163	1163	11	386	209	-56	-56	141	149	28	128	51
31	2166	Strain	1203	1203	11	392	209	-56	-56	141	149	28	128	51
32	2239	Strain	1243	1243	11	398	209	-56	-56	141	149	28	128	51
33	2312	Strain	1283	1283	11	404	209	-56	-56	141	149	28	128	51
34	2385	Strain	1323	1323	11	410	209	-56	-56	141	149	28	128	51
35	2458	Strain	1363	1363	11	416	209	-56	-56	141	149	28	128	51
36	2531	Strain	1403	1403	11	422	209	-56	-56	141	149	28	128	51
37	2604	Strain	1443	1443	11	428	209	-56	-56	141	149	28	128	51
38	2677	Strain	1483	1483	11	434	209	-56	-56	141	149	28	128	51
39	2750	Strain	1523	1523	11	440	209	-56	-56	141	149	28	128	51
40	2823	Strain	1563	1563	11	446	209	-56	-56	141	149	28	128	51
41	2896	Strain	1603	1603	11	452	209	-56	-56	141	149	28	128	51
42	2969	Strain	1643	1643	11	458	209	-56	-56	141	149	28	128	51
43	3042	Strain	1683	1683	11	464	209	-56	-56	141	149	28	128	51
44	3115	Strain	1723	1723	11	470	209	-56	-56	141	149	28	128	51
45	3188	Strain	1763	1763	11	476	209	-56	-56	141	149	28	128	51
46	3261	Strain	1803	1803	11	482	209	-56	-56	141	149	28	128	51
47	3334	Strain	1843	1843	11	488	209	-56	-56	141	149	28	128	51
48	3407	Strain	1883	1883	11	494	209	-56	-56	141	149	28	128	51
49	3480	Strain	1923	1923	11	500	209	-56	-56	141	149	28	128	51
50	3553	Strain	1963	1963	11	506	209	-56	-56	141	149	28	128	51
51	3626	Strain	2003	2003	11	512	209	-56	-56	141	149	28	128	51
52	3699	Strain	2043	2043	11	518	209	-56	-56	141	149	28	128	51
53	3772	Strain	2083	2083	11	524	209	-56	-56	141	149	28	128	51
54	3845	Strain	2123	2123	11	530	209	-56	-56	141	149	28	128	51
55	3918	Strain	2163	2163	11	536	209	-56	-56	141	149	28	128	51
56	3991	Strain	2203	2203	11	542	209	-56	-56	141	149	28	128	51
57	4064	Strain	2243	2243	11	548	209	-56	-56	141	149	28	128	51
58	4137	Strain	2283	2283	11	554	209	-56	-56	141	149	28	128	51
59	4210	Strain	2323	2323	11	560	209	-56	-56	141	149	28	128	51
60	4283	Strain	2363	2363	11	566	209	-56	-56	141	149	28	128	51
61	4356	Strain	2403	2403	11	572	209	-56	-56	141	149	28	128	51
62	4429	Strain	2443	2443	11	578	209	-56	-56	141	149	28	128	51
63	4502	Strain	2483	2483	11	584	209	-56	-56	141	149	28	128	51
64	4575	Strain	2523	2523	11	590	209	-56	-56	141	149	28	128	51
65	4648	Strain	2563	2563	11	596	209	-56	-56	141	149	28	128	51
66	4721	Strain	2603	2603	11	602	209	-56	-56	141	149	28	128	51
67	4794	Strain	2643	2643	11	608	209	-56	-56	141	149	28	128	51
68	4867	Strain	2683	2683	11	614	209	-56	-56	141	149	28	128	51
69	4940	Strain	2723	2723	11	620	209	-56	-56	141	149	28	128	51
70	5013	Strain	2763	2763	11	626	209	-56	-56	141	149	28	128	51
71	5086	Strain	2803	2803	11	632	209	-56	-56	141	149	28	128	51
72	5159	Strain	2843	2843	11	638	209	-56	-56	141	149	28	128	51
73	5232	Strain	2883	2883	11	644	209	-56	-56	141	149	28	128	51
74	5305	Strain	2923	2923	11	650	209	-56	-56	141	149	28	128	51
75	5378	Strain	2963	2963	11	656	209	-56	-56	141	149	28	128	51
76	5451	Strain	3003	3003	11	662	209	-56	-56	141	149	28	128	51
77	5524	Strain	3043	3043	11	668	209	-56	-56	141	149	28	128	51
78	5597	Strain	3083	3083	11	674	209	-56	-56	141	149	28	128	51
79	5670	Strain	3123	3123	11	680	209	-56	-56	141	149	28	128	51
80	5743	Strain	3163	3163	11	686	209	-56	-56	141	149	28	128	51
81	5816	Strain	3203	3203	11	692	209	-56	-56	141	149	28	128	51
82	5889	Strain	3243	3243	11	698	209	-56	-56	141	149	28	128	51
83	5962	Strain	3283	3283	11	704	209	-56	-56	141	149	28	128	51
84	6035	Strain	3323	3323	11	710	209	-56	-56	141	149	28	128	51
85	6108	Strain	3363	3363	11	716	209	-56	-56	141	149	28	128	51
86	6181	Strain	3403	3403	11	722	209	-56	-56	141	149	28	128	51
87	6254	Strain	3443	3443	11	728	209	-56	-56	141	149	28	128	51
88	6327	Strain	3483	3483	11	734	209	-56	-56	141	149	28	128	51
89	6400	Strain	3523	3523	11	740	209	-56	-56	141	149	28	128	51
90	6473	Strain	3563	3563	11	746	209	-56	-56	141	149	28	128	51
91	6546	Strain	3603	3603	11	752	209	-56	-56	141	149	28	128	51
92	6619	Strain	3643	3643	11	758	209	-56	-56	141	149	28	128	51
93	6692	Strain	3683	3683	11	764	209	-56	-56	141	149	28	128	51
94	6765	Strain	3723	3723	11	770	209	-56	-56	141	149	28	128	51
95	6838	Strain	3763	3763	11	776	209	-56	-56	141	149	28	128	51
96	6911	Strain	3803	3803	11	782	209	-56	-56	141	149	28	128	51
97	6984	Strain	3843	3843	11	788	209	-56	-56	141	149	28	128	51
98	7057	Strain	3883	3883	11	794	209	-56	-56	141	149	28	128	51
99	7130	Strain	3923	3923	11	800	209	-56	-56	141	149	28	128	51
100	7203	Strain	3963	3963	11	806	209	-56	-56	141	149	28	128	51
101	7276	Strain	4003	4003	11	812	209	-56	-56	141	149	28	128	51
102	7349	Strain	4043	4043										

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN 96-A

TEST PANEL A (FRONT DIAGONAL)

No.	Load(kN)	Test Serial No. 11									
		strain	strain	strain	strain	strain	strain	strain	strain		
1	10.0	-.86	-.17	-.77	-.24	63	6	6	57		
2	20.2	-1.63	-.28	-1.48	-.45	117	2	12	109		
3	30.0	-2.48	-.40	-2.23	-.67	175	6	19	164		
4	40.0	-3.47	-.53	-3.09	-.89	239	11	27	227		
5	50.0	-4.42	-.60	-3.92	-1.10	302	16	36	287		
6	60.0	-5.38	-.63	-4.76	-1.27	362	21	43	345		
7	70.5	-6.41	-.62	-5.61	-1.45	425	27	51	407		
8	80.2	-7.34	-.57	-6.37	-1.57	480	33	59	461		
9	91.0	-8.48	-.46	-7.26	-1.71	545	39	66	524		
10	100.9	-9.48	-.28	-7.99	-1.84	609	44	73	578		
11	110.5	-10.56	-.8	-8.75	-1.94	656	49	81	634		
12	118.9	-11.30	12	-9.24	-.201	666	54	87	672		
13	121.9	-12.08	32	-9.73	-.208	734	58	92	710		
14	127.6	-1.297	59	-10.21	-.215	774	62	98	748		
15	132.6	-1.368	84	-10.62	-.221	809	65	104	784		
16	138.9	-1.427	110	-10.97	-.227	840	69	108	814		
17	144.1	-1.521	152	-11.43	-.236	862	75	114	857		
18	149.5	-1.649	214	-11.82	-.253	935	81	122	908		
19	152.0	-1.719	249	-12.20	-.281	962	85	126	938		
20	157.6	-1.832	332	-12.20	-.281	981	68	116	963		
21	160.2	-1.867	368	-12.25	-.305	1009	66	114	983		
22	144.5	-1.794	384	-11.67	-.285	961	36	87	831		
23	140.2	-1.842	411	-11.78	-.295	968	42	93	858		
24	141.6	-1.911	453	-11.89	-.313	1019	49	103	992		
25	142.0	-2.113	693	-11.70	-.381	1097	55	114	1068		
26	142.7	-2.186	781	-11.47	-.419	1124	58	121	1096		
27	144.3	-2.289	844	-11.19	-.456	1144	63	127	1116		
28	108.0	-2.233	680	-10.74	-.480	1142	70	137	1113		
29	106.0	-9.909	8131	1824	-.2554	1236	182	38	1134		
AXIAL FORCE (kN)	-27.3	-31.2								23.9	24.6
BENDING MOMENT(kN.m)	120.6	28.1								-42.5	-38.9

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN 36-A

TEST PANEL A (REAR DIAGONAL)

No.	Load(kN)	Test Serial No. 11													
		strain	strain	strain	strain	strain	strain	strain	strain	strain	strain	strain	strain		
1	10.0	.77	-.12	-.80	.11	13	14	15	16	17	18	19	20		
2	20.2	-.163	-.37	-.174	-.23	0	64			6					
3	30.0	-.234	-.57	-.256	-.26	5	136			11					
4	40.0	-.315	-.47	-.347	-.32	9	186			17					
5	50.0	-.386	-.56	-.442	-.32	24	281			25					
6	60.0	-.474	-.62	-.535	-.29	32	326			36					
7	70.5	-.558	-.68	-.636	-.21	41	366			44					
8	80.2	-.632	-.70	-.729	-.9	50	453			54					
9	91.0	-.720	-.72	-.840	9	60	512			62					
10	100.9	-.782	-.71	-.939	32	60	560			76					
11	110.5	-.869	-.69	-.1043	58	69	639			86					
12	115.9	-.921	-.66	-.1116	80	90	696			97					
13	121.9	-.969	-.82	-.1187	102	93	739			106					
14	127.6	-.1022	-.58	-.1262	127	100	778			114					
15	132.8	-.1066	-.54	-.1330	152	107	853			123					
16	138.9	-.1104	-.50	-.1362	176	114	883			131					
17	144.1	-.1159	-.43	-.1476	211	124	925			139					
18	149.5	-.1224	-.32	-.1581	287	135	977			149					
19	152.0	-.1262	-.26	-.1645	296	142	977			163					
20	157.6	-.1348	-.3	-.2042	420	146	1007			170					
21	160.2	-.1372	3	-.2155	462	153	1071			176					
22	144.5	-.1136	32	-.1750	303	188	1093			186					
23	140.2	-.1307	117	-.1828	391	12	966			223					
24	141.6	-.1489	208	-.2067	477	118	958			44					
25	142.0	-.1788	441	-.2473	669	118	972			84					
26	142.7	-.2001	555	-.2867	776	118	1060			970					
27	144.3	-.2184	634	-.2888	833	12	1096			308					
28	108.0	-.2202	715	-.3126	904	13	1134			339					
29		-.8457	7298	249	1959	13	1167			366					
							1296								
AXIAL FORCE (kN)	-29.2													14.5	13.8
BENDING MOMENT(N.m)	169.5													-60.3	-53.9

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN 30-A

No.	Load (kip)	Test Serial No. 11																			
		17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36
1	10.0	5	3	3	-2	5	3	3	18	-17											
2	20.2	15	6	5	-3	10	3	10	41	-26											
3	30.0	24	9	7	-7	17	9	17	62	-54											
4	40.0	32	13	9	-9	22	14	22	82	-73											
5	50.0	41	15	12	-12	28	18	28	104	-92											
6	60.0	48	18	15	-15	34	23	34	125	-111											
7	70.5	56	23	18	-18	40	28	40	147	-131											
8	80.2	64	28	21	-21	47	33	47	167	-148											
9	91.0	74	31	22	-22	54	38	54	189	-168											
10	102.8	84	38	24	-24	59	43	59	209	-185											
11	110.5	97	43	27	-27	65	48	65	230	-204											
12	115.8	108	43	28	-28	69	48	69	243	-216											
13	121.8	105	44	31	-31	74	48	74	254	-228											
14	127.6	112	48	34	-34	77	54	77	270	-241											
15	132.6	118	51	35	-35	81	54	81	283	-252											
16	138.8	122	53	38	-38	85	54	85	294	-264											
17	144.1	129	56	39	-39	89	60	89	308	-276											
18	149.5	136	62	40	-40	94	60	94	324	-291											
19	152.0	144	64	42	-42	98	60	98	338	-302											
20	157.8	153	81	47	-47	115	60	115	359	-325											
21	160.2	160	85	49	-49	118	64	118	377	-339											
22	164.5	172	98	56	-56	131	64	131	417	-380											
23	168.2	179	102	62	-62	140	64	140	439	-402											
24	171.6	181	105	64	-64	144	64	144	454	-416											
25	174.0	184	108	66	-66	148	64	148	470	-430											
26	176.7	187	110	68	-68	152	64	152	486	-444											
27	183.3	197	119	74	-74	166	64	166	521	-480											
28	187.6	204	125	81	-81	178	64	178	559	-504											
29	194.0	211	131	87	-87	189	64	189	600	-528											
30	198.0	218	137	93	-93	200	64	200	644	-552											
AXIAL FORCE (KN)	-33.9	-20.6	16.9	17.6	13.9	22.1															
BENDING MOMENT (K.L)	417.3	21.9	-112.5	-189.2	-422.6	-263.7															

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN 36-B

TEST PANEL A (FRONT DIAGONAL BUCKLED)

No.	Load (kN)	Test Serial No. 13									
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-23.7	19	2	3	4	5	6	7	8		
2	-40.8	23	10	38	-178	144	18	171			
3	-58.9	40	28	434	-321	253	-1	300			
4	-77.4	64	50	567	-476	368	63	434			
5	-96.1	91	83	688	-608	484	93	567			
6	-105.5	108	108	759	-809	598	125	688			
7	-114.5	123	131	823	-990	654	143	759			
8	-124.1	140	159	869	-1089	710	159	823			
9	-133.4	158	184	891	-1188	768	179	869			
10	-142.4	178	234	891	-1283	826	198	891			
11	-151.7	202	277	840	-1408	862	219	952			
12	-154.4	220	317	840	-1489	940	239	1078			
13	-153.8	237	358	840	-1551	1000	268	1148			
14	-154.9	245	375	840	-1579	1015	243	1165			
15	-122.1	179	199	840	-974	850	192	987			
16	-121.8	181	187	840	-867	853	320	991			
17	-122.2	184	198	840	-971	859	323	997			
18	-122.0	186	195	840	-868	863	328	1000			
19	-121.2	180	183	840	-960	869	333	1005			
20	-121.3	198	192	840	-987	872	331	1007			
21	-120.4	211	196	840	-897	871	302	1003			
22	-115.1	329	287	840	-1232	860	83	972			
23	-114.9	349	279	840	-1263	885	60	976			
24	-114.4	387	300	840	-1317	890	25	987			
25	-113.8	416	314	840	-1356	889	5	984			
26	-112.9	518	369	840	-1495	934	85	1031			
27	-111.4	689	448	840	-1665	1011	186	1097			
28	-111.7	791	476	840	-1732	1050	218	1132			
29	-104.9	1297	448	840	-1808	1187	283	1228			
30	-103.5	4008	1451	840	354	1199	132	1234			
AXIAL FORCE (kN)		-25.4									
BENDING MOMENT(N.m)		144.9									
		88.5									
		17.3									
		17.9									
		-58.6									
		-91.1									

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN S6-B

TEST PANEL A (REAR DIAGONAL BUCKLED FIRST)

No.	Load (kN)	Test Serial No. 13									
		9	10	11	12	13	14	15	16		
1	-23.7	-49	-184	-84	-148	148	133	17	18	16	
2	-40.6	-77	-331	-103	-286	242	222	31	242	29	
3	-58.6	-96	-487	-140	-428	347	320	45	320	47	
4	-77.4	-101	-661	-174	-567	459	422	61	422	61	
5	-96.1	-98	-853	-206	-710	575	528	78	528	113	
6	-105.5	-73	-955	-220	-779	631	582	86	582	125	
7	-114.5	-59	-1062	-237	-844	689	634	97	634	139	
8	-124.1	-23	-1178	-253	-913	749	691	107	691	151	
9	-133.4	14	-1301	-272	-975	807	744	118	744	165	
10	-142.4	61	-1430	-297	-1029	864	797	128	797	178	
11	-151.7	120	-1564	-321	-1078	922	850	138	850	194	
12	-154.4	150	-1600	-338	-1066	930	853	134	853	194	
13	-153.6	164	-1582	-354	-1019	912	829	118	829	185	
14	-184.9	186	-1813	-389	-1029	829	833	114	833	183	
15	-122.1	155	-1314	-281	-858	738	639	5	639	80	
16	-121.8	160	-1314	-259	-852	733	634	-10	634	77	
17	-122.2	165	-1319	-259	-852	734	634	-13	634	75	
18	-122.0	166	-1316	-257	-847	731	630	-19	630	71	
19	-121.2	176	-1311	-253	-839	724	620	-29	620	64	
20	-121.3	181	-1314	-253	-837	724	620	-33	620	61	
21	-120.4	184	-1312	-255	-831	728	620	-35	620	61	
22	-115.1	174	-1295	-267	-810	740	638	-18	638	73	
23	-114.9	177	-1297	-269	-807	743	640	-19	640	73	
24	-114.4	186	-1303	-269	-803	748	645	-22	645	73	
25	-113.8	189	-1300	-268	-798	750	645	-24	645	71	
26	-112.9	206	-1308	-268	-785	761	654	-30	654	69	
27	-111.4	227	-1299	-264	-757	773	660	-42	660	65	
28	-111.7	241	-1305	-264	-749	782	667	-45	667	64	
29	-106.0	227	-1148	-224	-645	730	603	-90	603	33	
30	-103.5	255	-1223	-293	-621	776	634	-125	634	15	
AXIAL FORCE (kN)		-28.3		-27.3		20.5			20.1		
BENDING MOMENT(N.m)		78.7		26.9		-31.7			-25.6		

AXIAL FORCES AND BENDING MOMENT IN TEST MEMBERS FOR SPECIMEN 96-B

TEST PANEL A (LEGS)

No.	Test Serial No. 13														
	Load (kN)	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-23.7	-15	5	-10	-1	4	2	3	3	4	3	3	3	3	3
2	-40.6	-28	10	-16	-2	7	2	5	5	7	5	5	5	5	5
3	-58.6	-42	16	-24	-3	10	6	7	7	10	7	7	7	7	7
4	-77.4	-57	22	-32	-3	13	9	10	10	13	9	9	9	9	9
5	-96.1	-73	29	-40	-4	18	11	11	11	18	11	11	11	11	11
6	-105.5	-90	35	-44	-4	19	13	13	13	19	13	13	13	13	13
7	-114.5	-98	35	-48	-6	20	14	14	14	20	14	14	14	14	14
8	-124.1	-96	38	-53	-6	15	15	15	15	21	15	15	15	15	15
9	-132.4	-105	41	-57	-6	16	16	16	16	23	16	16	16	16	16
10	-142.4	-113	45	-61	-6	18	18	18	18	24	18	18	18	18	18
11	-151.7	-121	48	-66	-7	18	18	18	18	26	18	18	18	18	18
12	-154.4	-126	51	-69	-6	19	19	19	19	26	19	19	19	19	19
13	-153.6	-129	55	-68	-6	18	18	18	18	25	18	18	18	18	18
14	-154.8	-134	61	-68	-4	20	20	20	20	25	20	20	20	20	20
15	-122.1	-114	51	-58	-4	36	36	36	36	21	36	36	36	36	36
16	-121.8	-114	52	-58	-4	37	37	37	37	21	37	37	37	37	37
17	-122.2	-115	55	-58	-5	37	37	37	37	21	37	37	37	37	37
18	-122.0	-115	52	-58	-5	37	37	37	37	21	37	37	37	37	37
19	-121.2	-112	49	-57	-6	38	38	38	38	20	38	38	38	38	38
20	-121.3	-111	47	-56	-6	41	41	41	41	20	41	41	41	41	41
21	-120.4	-101	38	-52	-10	40	40	40	40	20	40	40	40	40	40
22	-115.1	-42	28	-26	-27	33	33	33	33	20	33	33	33	33	33
23	-114.8	-35	22	-25	-29	33	33	33	33	20	33	33	33	33	33
24	-114.4	-21	24	-18	-34	31	31	31	31	20	31	31	31	31	31
25	-113.8	-10	48	-14	-38	30	30	30	30	21	30	30	30	30	30
26	-112.9	2	66	2	-53	29	29	29	29	21	29	29	29	29	29
27	-111.4	92	150	27	-77	24	24	24	24	20	24	24	24	24	24
28	-111.7	117	175	37	-88	22	22	22	22	21	22	22	22	22	22
29	-108.9	188	217	66	-104	14	14	14	14	18	14	14	14	14	14
30	-103.5	132	183	48	-103	10	10	10	10	18	10	10	10	10	10
AXIAL FORCE (kN)	-22.2			-18.6		8				7.3				-69.7	20.7
BENDING MOMENT (kN.m)	125.1			532		32.4				1154.3				786.3	1893.8

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN S6-C

TEST PANEL A (FRONT, PANEL A NOT BUCKLED BECAUSE OF PANEL C,D BUCKLED AFTER THE FIRST HORIZONTAL BUCKLED)

No.	Load (kN)	Strain										Strain	Strain	Strain	Strain				
		1	2	3	4	5	6	7	8	9	10								
1	-54.8	-255	-80	-267	-71	39	202	47	202	202		7	7				202		
2	-55.6	-418	-114	-434	-103	66	323	71	323	323		71	71				323		
3	-71.9	-551	-134	-568	-124	88	418	100	418	418		100	100				418		
4	-68.8	-688	-147	-709	-139	112	513	125	513	513		125	125				513		
5	-108.4	-851	-167	-857	-150	140	622	152	622	622		152	152				622		
6	-118.2	-983	-141	-956	-155	159	681	172	682	682		172	172				682		
7	-138.4	-1144	-116	-1112	-159	191	787	202	787	787		202	202				787		
8	-154.1	-1338	-73	-1262	-156	225	896	232	896	896		232	232				896		
9	-171.9	-1548	-4	-1411	-154	264	989	262	989	989		262	262				989		
10	-182.9	-1693	57	-1497	-152	285	1051	278	1051	1051		278	278				1051		
11	-186.6	-1749	90	-1519	-155	288	1070	280	1070	1070		280	280				1070		
12	-187.1	-1787	128	-1518	-156	273	1072	267	1072	1116		267	267				1116		
13	-185.4	-1800	149	-1510	-159	255	1064	252	1064	1107		252	252				1107		
14	-184.6	-1832	177	-1508	-181	241	1081	239	1081	1102		239	239				1102		
15	-184.9	-1729	172	-1449	-131	186	986	185	986	1027		185	185				1027		
16	-185.0	-1783	200	-1465	-131	163	998	183	998	1039		183	183				1039		
17	-183.6	-1771	197	-1456	-131	178	991	179	991	1028		179	179				1028		
18	-167.9	-1661	280	-1483	-139	165	1028	186	1028	1065		186	186				1065		
19	-169.3	-1828	292	-1502	-149	185	1040	189	1040	1078		189	189				1078		
20	-171.7	-1968	335	-1509	-163	189	1059	194	1059	1097		194	194				1097		
21	-175.0	-2121	452	-1499	-204	193	1082	200	1082	1128		200	200				1128		
22	-168.0	-2134	519	-1437	-155	115	1040	128	1040	1070		128	128				1070		
23	-126.4	-1236	184	-932	-92	198	860	214	860	875		214	214				875		
24	-141.0	-1187	177	-897	-84	234	878	250	878	895		250	250				895		
25	-143.4	-1051	166	-786	-61	310	911	324	911	927		324	324				927		
26	-148.4	-823	171	-693	-33	402	985	418	985	990		418	418				990		
27	-150.2	-879	178	-655	-20	434	986	450	986	1000		450	450				1000		
AXIAL FORCE (KN)																			

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN 318RB

(TOP LEG BUCKLED FIRST, NO DIAGONAL BUCKLED)

No.	Load kN	Test Serial No. 14											
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
		1	2	3	4	5	6	7	8	9	10	11	12
1	-8.2	-36	-26	32	37	40	29	-34	-33	-37	-35	32	38
2	-15.5	-68	-49	64	75	73	53	-69	-69	-69	-61	65	77
3	-23.7	-104	-73	100	119	109	80	-112	-108	-103	-94	104	124
4	-27.9	-124	-88	117	138	130	86	-131	-124	-123	-112	122	143
5	-32.3	-147	-102	135	156	152	112	-151	-140	-144	-131	139	163
6	-36.7	-169	-116	151	175	177	130	-171	-157	-169	-149	158	184
7	-41.2	-182	-131	169	196	200	147	-190	-175	-190	-170	176	204
8	-45.6	-216	-147	186	215	224	163	-210	-192	-215	-190	194	225
9	-50.1	-241	-161	204	234	248	181	-231	-209	-237	-209	214	246
10	-54.7	-267	-176	222	254	272	201	-251	-226	-261	-229	233	267
11	-54.6	-264	-175	221	255	271	201	-250	-224	-260	-229	230	266
12	-62.4	-306	-196	253	289	313	232	-285	-254	-300	-264	264	302
13	-68.9	-332	-213	270	308	335	249	-308	-273	-323	-284	284	324
14	-71.4	-356	-226	287	329	359	268	-328	-289	-347	-303	302	345
15	-75.8	-381	-238	304	348	383	286	-347	-304	-370	-323	320	363
16	-80.4	-407	-251	323	369	406	306	-371	-321	-394	-343	339	386
17	-85.0	-435	-264	342	386	430	323	-393	-339	-418	-363	356	408
18	-89.6	-461	-275	359	407	453	342	-414	-354	-442	-383	378	428
19	-94.1	-488	-286	377	428	477	360	-437	-370	-465	-403	396	446
20	-98.5	-513	-296	395	446	499	378	-457	-385	-486	-422	415	467
21	-103.0	-542	-305	414	465	523	398	-481	-401	-510	-441	434	487
22	-107.5	-570	-315	433	484	544	414	-503	-416	-533	-461	453	508
23	-111.8	-595	-321	449	502	566	434	-524	-428	-555	-480	471	527
24	-116.3	-625	-329	469	521	590	449	-547	-443	-579	-500	491	547
25	-120.7	-655	-336	486	540	610	469	-571	-457	-598	-520	510	567
26	-124.9	-684	-342	503	557	631	485	-594	-468	-618	-540	528	593
27	-129.9	-710	-346	517	574	650	502	-615	-478	-633	-560	546	603
28	-133.0	-740	-350	532	591	669	519	-638	-489	-650	-582	564	621
29	-136.9	-769	-351	548	608	689	532	-661	-498	-665	-602	583	638
30	-139.8	-782	-351	559	618	701	543	-680	-503	-685	-624	598	651
31	-142.2	-814	-349	567	627	710	550	-697	-502	-690	-647	613	661
32	-143.6	-834	-347	578	633	718	563	-714	-509	-691	-668	628	680
33	-89.8	-507	-265	332	384	418	320	-474	-248	-417	-361	428	495
34	-88.3	-499	-260	325	376	411	312	-466	-244	-414	-351	456	418
AXIAL FORCES(kN)		-46.1											
		47.2											
		49.6											
		-47.4											
		-51.5											
		50.5											

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN 318RB

No.	Load (kN)	Test Serial No. 14																
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain
1	-8.2	37	33	-31	-37	-38	29	33	36	22	28	28	43	24				
2	-15.5	69	61	-69	-74	-55	101	69	72	40	53	86	86	43				
3	-23.7	103	92	-109	-118	-83	155	109	108	59	83	137	86	86				
4	-27.9	124	112	-126	-137	-100	186	128	128	71	96	161	161	161				
5	-32.3	148	131	-147	-157	-116	227	145	152	86	107	185	185	185				
6	-36.7	169	151	-165	-176	-134	266	163	176	100	119	210	210	210				
7	-41.2	190	171	-182	-196	-151	303	182	196	118	129	238	238	238				
8	-45.6	213	181	-201	-216	-171	339	200	222	131	142	281	281	281				
9	-50.1	235	213	-219	-237	-188	371	219	246	147	151	288	288	288				
10	-54.7	259	234	-237	-257	-205	398	238	270	163	163	316	316	316				
11	-54.6	258	233	-237	-257	-206	408	237	270	163	163	316	316	316				
12	-62.4	297	269	-269	-294	-235	481	269	311	180	179	363	363	363				
13	-68.9	321	292	-285	-315	-252	522	288	359	206	189	392	392	392				
14	-71.4	343	313	-302	-338	-268	554	307	359	224	197	421	421	421				
15	-75.6	365	332	-321	-358	-284	591	324	381	239	206	448	448	448				
16	-80.4	389	355	-338	-378	-300	630	345	406	257	213	478	478	478				
17	-85.0	411	376	-355	-402	-316	663	363	430	274	220	510	510	510				
18	-89.6	434	398	-372	-425	-332	693	381	456	281	226	542	542	542				
19	-94.1	456	418	-388	-449	-346	727	399	481	306	230	574	574	574				
20	-98.5	479	439	-402	-470	-359	764	418	504	325	235	606	606	606				
21	-103.0	502	460	-418	-496	-372	807	437	528	343	239	640	640	640				
22	-107.5	524	481	-432	-520	-385	847	455	555	361	241	674	674	674				
23	-111.8	545	501	-444	-544	-407	890	473	578	378	241	708	708	708				
24	-116.3	567	522	-455	-573	-425	930	491	603	396	241	744	744	744				
25	-120.7	590	543	-465	-602	-447	969	509	629	413	241	783	783	783				
26	-124.9	611	563	-489	-634	-468	1046	525	653	429	237	820	820	820				
27	-129.9	630	583	-499	-668	-490	1102	542	677	445	232	857	857	857				
28	-133.0	650	601	-498	-705	-494	1132	558	703	459	225	895	895	895				
29	-136.9	669	618	-481	-745	-439	1171	574	727	474	218	933	933	933				
30	-139.8	685	633	-430	-802	-426	1214	584	751	485	203	965	965	965				
31	-142.2	699	644	-380	-873	-407	1239	583	773	480	184	981	981	981				
32	-143.6	709	643	-318	-888	-372	1269	600	793	482	180	1006	1006	1006				
33	-86.6	453	386	-1149	-1149	215	1058	594	594	230	158	581	581	581				
34	-88.3	446	381	-1145	-1145	214	1068	585	585	226	155	571	571	571				
AXIAL FORCE(kN)		53.1		-49.7		-51.5	73			50.2		-46.7						

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN 318RB

No.	Load (kN)	Test Serial No. 14												
		Strain 25	Strain 26	Strain 27	Strain 28	Strain 29	Strain 30	Strain 31	Strain 32	Strain 33	Strain 34	Strain 35	Strain 36	
1	-8.2	-43	-36	-29	-38	-39	-37	-29	-28	-4	-37	-35	-35	
2	-15.5	-84	-74	-51	-67	-77	-66	-54	-6	-74	-6	-65	-65	
3	-23.7	-127	-111	-73	-99	-118	-100	-87	-115	-13	-115	-87	-87	
4	-27.9	-151	-131	-89	-120	-136	-120	-87	-132	-15	-132	-116	-116	
5	-32.3	-168	-149	-104	-142	-156	-143	-116	-143	-17	-151	-138	-138	
6	-36.7	-189	-169	-120	-167	-175	-164	-135	-171	-20	-171	-161	-161	
7	-41.2	-210	-187	-137	-191	-194	-186	-153	-22	-186	-22	-182	-182	
8	-45.6	-229	-208	-151	-211	-214	-214	-172	-24	-207	-26	-206	-206	
9	-50.1	-249	-228	-167	-236	-233	-233	-190	-26	-233	-26	-227	-227	
10	-54.7	-268	-245	-181	-261	-253	-253	-209	-29	-244	-29	-250	-250	
11	-59.6	-270	-248	-182	-262	-257	-257	-210	-29	-243	-29	-250	-250	
12	-62.4	-307	-284	-208	-310	-296	-296	-241	-32	-276	-32	-269	-269	
13	-68.9	-324	-300	-221	-332	-313	-313	-261	-35	-295	-35	-313	-313	
14	-71.4	-345	-324	-236	-358	-333	-333	-279	-37	-336	-37	-336	-336	
15	-75.6	-362	-340	-247	-381	-351	-351	-297	-39	-329	-39	-358	-358	
16	-80.4	-382	-362	-261	-410	-370	-370	-317	-41	-350	-41	-382	-382	
17	-85.0	-405	-385	-275	-441	-387	-387	-336	-44	-368	-44	-405	-405	
18	-89.6	-421	-405	-285	-469	-410	-410	-355	-46	-386	-46	-429	-429	
19	-94.1	-440	-425	-295	-494	-428	-428	-375	-48	-403	-48	-452	-452	
20	-98.5	-457	-444	-304	-522	-445	-445	-394	-51	-421	-51	-475	-475	
21	-103.0	-476	-467	-314	-554	-464	-464	-412	-53	-438	-53	-496	-496	
22	-107.5	-493	-488	-321	-583	-481	-481	-433	-56	-456	-56	-522	-522	
23	-111.8	-511	-512	-331	-616	-500	-500	-452	-58	-473	-58	-543	-543	
24	-116.3	-528	-534	-337	-649	-518	-518	-472	-60	-489	-60	-567	-567	
25	-120.7	-543	-557	-339	-681	-535	-535	-493	-62	-508	-62	-580	-580	
26	-124.9	-556	-579	-340	-716	-550	-550	-516	-65	-524	-65	-612	-612	
27	-129.9	-573	-605	-342	-751	-564	-564	-537	-67	-537	-67	-634	-634	
28	-133.0	-587	-630	-339	-791	-579	-579	-561	-69	-554	-69	-655	-655	
29	-136.9	-598	-653	-334	-829	-594	-594	-584	-71	-569	-71	-677	-677	
30	-139.8	-608	-678	-322	-868	-603	-603	-610	-73	-578	-73	-692	-692	
31	-142.2	-613	-698	-308	-900	-608	-608	-631	-75	-588	-75	-703	-703	
32	-143.8	-613	-709	-288	-925	-607	-607	-645	-76	-595	-76	-712	-712	
33	-89.8	-389	-465	-150	-629	-411	-411	-420	-48	-397	-48	-454	-454	
34	-89.3	-392	-456	-147	-619	-394	-394	-414	-47	-390	-47	-447	-447	
AXIAL FORCE (kN)		-51.6		-47.4		-50.1		-51.0		-26.2				

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN 33SRC

No.	Load kN	Test Serial No. 16													
		Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain	Strain		
1	-48.0	-97	-149	-86	-108	-120	-134	-105	-80	-132	-115	-121			
2	-58.4	-121	-138	-104	-134	-149	-161	-127	-100	-164	-142	-151			
3	-68.2	-147	-181	-123	-163	-182	-193	-151	-123	-190	-174	-184			
4	-87.3	-190	-277	-152	-211	-237	-245	-190	-156	-282	-225	-239			
5	-110.8	-243	-351	-208	-268	-304	-308	-237	-197	-339	-286	-309			
6	-124.0	-272	-392	-208	-302	-342	-344	-265	-219	-384	-324	-348			
7	-138.5	-300	-434	-227	-333	-378	-378	-290	-239	-427	-357	-388			
8	-162.2	-355	-519	-264	-393	-450	-450	-343	-277	-515	-425	-482			
9	-177.8	-387	-688	-284	-430	-500	-494	-378	-300	-571	-466	-506			
10	-182.9	-418	-824	-305	-465	-544	-538	-406	-320	-625	-508	-555			
11	-235.4	-477	-966	-340	-527	-626	-617	-461	-353	-727	-577	-640			
12	-235.4	-507	-1128	-357	-590	-720	-660	-482	-368	-827	-616	-687			
13	-251.8	-540	-1259	-375	-655	-843	-782	-525	-396	-945	-657	-738			
14	-265.5	-567	-1320	-390	-624	-922	-868	-553	-398	-1028	-682	-782			
15	-282.1	-599	-1438	-407	-657	-1016	-943	-584	-410	-1138	-733	-834			
16	-287.0	-628	-1517	-421	-687	-1128	-1042	-614	-432	-1271	-770	-881			
17	-323.1	-676	-1597	-441	-737	-1259	-1089	-664	-436	-1405	-835	-970			
18	-338.5	-703	-1685	-458	-782	-1373	-1128	-720	-437	-1568	-973	-1022			
19	-353.4	-731	-1778	-464	-815	-1438	-1179	-745	-434	-1719	-909	-1078			
20	-367.0	-753	-1857	-468	-834	-1517	-1234	-766	-431	-1847	-944	-1125			
21	-378.7	-771	-1937	-468	-855	-1600	-1286	-780	-422	-1978	-973	-1168			
22	-382.3	-793	-2017	-468	-876	-1685	-1332	-815	-407	-2119	-1008	-1220			
23	-407.8	-814	-2155	-468	-883	-1778	-1378	-836	-381	-2260	-1042	-1281			
24	-421.2	-832	-2253	-461	-897	-1865	-1411	-852	-301	-2402	-1077	-1344			
25	-433.3	-843	-2353	-434	-897	-1957	-1450	-857	-256	-2543	-1121	-1405			
26	-438.3	-847	-2447	-404	-885	-2050	-1488	-857	-207	-2684	-1143	-1465			
27	-443.7	-850	-2547	-381	-888	-2147	-1527	-866	-116	-2825	-1165	-1528			
28	-448.8	-851	-2647	-361	-879	-2247	-1566	-873	20	-2966	-1189	-1591			
29	-454.7	-854	-2747	-343	-869	-2347	-1605	-881	254	-3107	-1254	-1748			
30	-460.3	-857	-2847	-48	-859	-2447	-1644	-881	284	-3248	-1371	-1870			
31	2.6	112	-188	-48	60	-435	-58780	31908	40688	-22604	-10430	-13307			
AXIAL FORCES (kN)		buckled inside												buckled outside	
		-144.3												-150.5	
		-154												-212.8	
		-147.3													
		-148.6													

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN 33SRC

Test Serial No. 16

No.	Load (kN)	Strain	13	14	15	16	17	18	19	20	21	22	23	24
1	-48.0	-121	-113	-122	-112	-122	-82	-127	-105	-114	-115	-122	-128	-108
2	-56.4	-147	-136	-147	-135	-147	-112	-155	-128	-139	-143	-151	-159	-135
3	-68.2	-174	-163	-174	-159	-174	-135	-186	-154	-167	-175	-183	-191	-161
4	-87.3	-219	-206	-222	-202	-219	-168	-240	-195	-214	-222	-237	-249	-208
5	-110.8	-273	-260	-278	-252	-273	-210	-308	-247	-271	-280	-304	-319	-264
6	-124.0	-303	-291	-312	-279	-303	-231	-346	-274	-303	-316	-345	-358	-298
7	-138.5	-332	-321	-342	-307	-332	-253	-383	-302	-334	-343	-379	-393	-322
8	-162.2	-388	-382	-406	-361	-388	-282	-460	-356	-398	-404	-456	-472	-388
9	-177.8	-423	-421	-445	-394	-423	-317	-509	-390	-437	-445	-507	-519	-420
10	-182.9	-457	-457	-485	-426	-457	-339	-556	-422	-475	-478	-550	-561	-453
11	-220.3	-519	-528	-566	-484	-519	-383	-646	-482	-547	-536	-633	-639	-511
12	-235.4	-552	-565	-596	-516	-552	-404	-697	-515	-588	-588	-680	-685	-543
13	-251.8	-588	-606	-640	-549	-588	-426	-752	-551	-631	-600	-726	-732	-577
14	-265.5	-618	-641	-678	-577	-618	-445	-801	-580	-669	-630	-774	-771	-606
15	-282.1	-655	-685	-721	-611	-655	-468	-861	-617	-714	-662	-823	-819	-639
16	-297.0	-698	-725	-763	-642	-698	-487	-915	-649	-756	-691	-870	-863	-683
17	-323.1	-743	-785	-834	-683	-743	-517	-1014	-704	-829	-738	-953	-939	-718
18	-338.5	-775	-836	-877	-723	-775	-532	-1073	-736	-873	-766	-1004	-983	-745
19	-353.4	-806	-877	-920	-751	-806	-546	-1133	-767	-916	-790	-1056	-1029	-772
20	-367.0	-832	-916	-959	-777	-832	-557	-1188	-795	-955	-812	-1102	-1069	-788
21	-378.7	-856	-948	-993	-796	-856	-566	-1238	-817	-991	-831	-1142	-1101	-818
22	-382.3	-882	-986	-1033	-822	-882	-575	-1297	-846	-1030	-853	-1191	-1145	-839
23	-407.8	-912	-1030	-1079	-849	-912	-583	-1366	-876	-1077	-877	-1246	-1183	-865
24	-421.2	-940	-1072	-1124	-872	-940	-589	-1430	-904	-1120	-897	-1297	-1238	-885
25	-433.3	-962	-1114	-1166	-890	-962	-589	-1483	-931	-1159	-910	-1342	-1277	-883
26	-438.3	-970	-1132	-1185	-897	-970	-588	-1522	-943	-1175	-917	-1364	-1297	-901
27	-443.7	-979	-1150	-1204	-904	-979	-587	-1552	-954	-1182	-922	-1384	-1317	-904
28	-448.8	-986	-1172	-1228	-912	-986	-590	-1597	-967	-1210	-922	-1407	-1336	-904
29	-454.7	-995	-1186	-1248	-920	-995	-574	-1629	-983	-1231	-924	-1434	-1356	-904
30	-460.3	-1003	-1228	-1271	-928	-1003	-563	-1672	-999	-1232	-927	-1466	-1381	-904
31	2.6	-18	-380	-378	-48	-106	106	-61	-41	14	68	-74	10	14
AXIAL FORCES (kN)		-142.6		-141.1		-143.4		-144.4		-153.5			-146.6	

AXIAL FORCES IN TEST MEMBERS FOR SPECIMEN 339RC

No.	Load kN	Test Serial No. 16														
		Strain	25	26	27	28	29	30	31	32	33	34	35	36		
1	-48.0	.13	-106	-129	-116	-119	-78	-139	-103							
2	-58.4	.15	-122	-160	-143	-145	-95	-173	-127							
3	-68.2	.16	-159	-194	-171	-174	-116	-210	-153							
4	-87.3	.23	-203	-250	-222	-223	-146	-272	-197							
5	-110.8	.26	-258	-320	-282	-283	-185	-348	-249							
6	-124.0	.33	-289	-357	-316	-312	-209	-381	-277							
7	-136.5	.36	-320	-394	-348	-341	-230	-433	-304							
8	-162.2	.42	-382	-489	-415	-407	-277	-519	-359							
9	-177.8	.46	-420	-515	-456	-447	-304	-571	-392							
10	-192.9	.49	-457	-557	-493	-481	-331	-622	-422							
11	-220.3	.56	-527	-637	-562	-551	-381	-716	-475							
12	-235.4	.60	-566	-681	-598	-591	-408	-767	-502							
13	-251.8	.64	-608	-727	-638	-634	-440	-824	-530							
14	-265.5	.67	-646	-766	-670	-671	-468	-873	-553							
15	-282.1	.71	-680	-813	-708	-712	-497	-932	-579							
16	-297.0	.75	-731	-855	-743	-752	-525	-986	-602							
17	-323.1	.81	-802	-930	-802	-820	-574	-1084	-639							
18	-336.5	.84	-845	-975	-838	-861	-602	-1142	-658							
19	-353.4	.88	-888	-1018	-872	-904	-630	-1203	-677							
20	-367.0	.91	-925	-1059	-901	-940	-653	-1259	-682							
21	-378.7	.93	-959	-1084	-927	-971	-675	-1309	-703							
22	-392.3	.96	-996	-1141	-954	-1010	-700	-1379	-711							
23	-407.6	1.00	-1044	-1200	-982	-1056	-725	-1472	-709							
24	-421.2	1.03	-1084	-1257	-1000	-1099	-749	-1564	-699							
25	-433.3	1.06	-1124	-1316	-1010	-1138	-771	-1674	-673							
26	-438.3	1.07	-1140	-1342	-1012	-1154	-779	-1722	-660							
27	-443.7	1.08	-1157	-1365	-1017	-1171	-788	-1768	-649							
28	-448.8	1.09	-1176	-1395	-1018	-1190	-797	-1830	-630							
29	-454.7	1.11	-1198	-1421	-1022	-1209	-806	-1883	-615							
30	-460.3	1.12	-1218	-1446	-1026	-1233	-814	-1939	-600							
31	2.6	2	-20	.54	5	6	11	.175	57							
AXIAL FORCES (kN)			-85.3		-159.5		-131.3		-162.4							

VITA AUCTORIS

Yean Sun was born in Weifang, China on March 3rd, 1966.

He graduated with a Bachelors Degree of Civil Engineering from Harbin Architectural University, China in July 1988. Later on, he had been working as a senior professional engineer in the field of civil and environmental engineering during the period 1988-1998.

He registered as a student pursuing the degree of Master of Applied Science in Civil Engineering at the University of Windsor, Windsor, Ontario, Canada in September 1998.